



VOLUME II SATURN I COUNTDOWN

JOHN F. KENNEDY SPACE CENTER, NASA

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SA-7

MANUAL

SATURN I COUNTDOWN MANUAL

(VOL. II)

SA-7

TEST NUMBER

7-LSVI-300

LAUNCH COUNTDOWN

SEPTEMBER 5, 1964

SATURN I COUNTDOWN MANUAL

(VOL. I & VOL. II)

Prepared by:

Approved by:

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APPROVED BY:

OPERATIONS

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LAUNCH OPERATIONS TEST	125
1. TITLE: LAUNCH COUNTDOWN	7-LSVI-300 3. SA-7
4. OBJECTIVES: TO PROVIDE A SEQUENCE OF OPERATION REQUIRED TO LAUNCH VEHICLE TO THE POINT OF FIRING COMMAND (INITIATI FIRING SEQUENCE); TO PROVIDE EMERGENCY PROCEDURES FOR S IN THE EVENT OF STANDARD MALFUNCTIONS; TO PROVIDE THE ALISTING OF INTERRELATED ITEMS MENT BEFORE AND AFTER LIFTOFF; AND TO PROVIDE THE COMPLITHE IMMEDIATE POST-LAUNCH SECURING OPERATIONS.	PREPARE THE ON OF TERMINAL AFING OF VEHICLE WR RANGE AND FOR ACCOMPLISH-
5. TEST DESCRIPTION: 1. THE COUNTDOWN INCLUDES OR REFERENCES ALL ACTIVITIES TO ACCOMPLISH THE LAUNCH OF THE SATURN I VEHICLE. THIS DIVIDED INTO TWO PARTS (SPLIT COUNT) NORMALLY TO BE PER CONSECUTIVE DAYS AFTER THE COMPLETION OF FIRST STAGE RP	OPERATION IS FORMED ON TWO

AND ASSOCIATED OPERATIONS.

2. EMERGENCY PROCEDURES FOR STANDARD SAFING OPERATIONS ARE INCLUDED

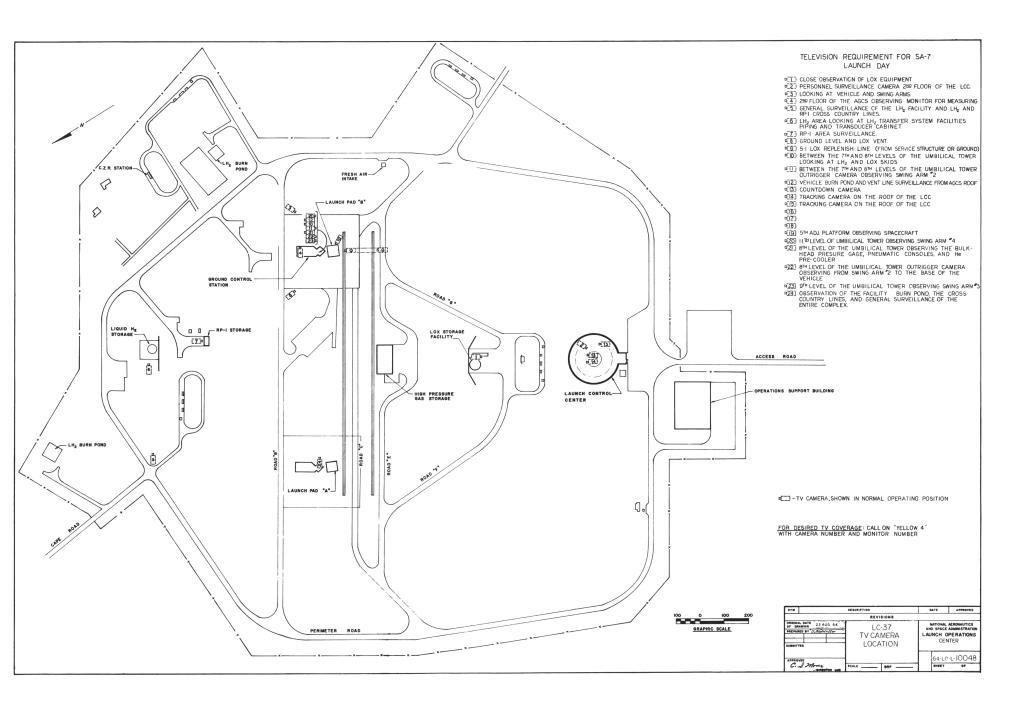
AS A PART OF THE COUNTDOWN MANUAL.

3. Post-Liftoff operations terminate with payload injection into orbit.

6. STATUS:

1. ALL PRELAUNCH PREPARATIONS AND CHECKOUT SATISFACTORILY COMPLETED.

7. PREPARED BY:	9. NASA SYSTEMS ENGINEER	11. DATE
		September 4, 1964
8. CONTRACTOR GROUP ENGINEER	10. NASA SYSTEMS SUPERVISOR	12. CONTRACTOR REFERENCE NR.



ENGINE CUTOFF AND EMERGENCY OPERATIONS

1. NETWORK PANEL CUTOFF

CUTOFF FROM THE NETWORK PANEL SHOULD BE INITIATED ONLY
IN A SITUATION WHERE A TURBINE SPINNER SQUIB INADVERTENTLY
IGNITES PRIOR TO THE IGNITION COMMAND. THE CUTOFF BUTTON
SHOULD BE DEPRESSED UNTIL PREVALVES 1-8 SWITCHES ARE
TURNED OFF. (FOR DETAILED PROCEDURE SEE "EMERGENCY
PROCEDURE" INCORPORATED IN THE LAUNCH COUNTDOWN.)

2. FIRING PANEL CUTOFF

Once firing command has been initiated at T-2'33" and the automatic sequence begins, it is necessary to give cutoff from the firing panel to stop the sequence and interrupt the associated firing circuitry. This cutoff will
not fire the conax valve squibs nor close the prevalves
unless ignition command has been initiated. (For
Detailed procedure see "Emergency Procedure" incorporated
in the Launch Countdown.)

3. <u>Automatic Malfunction Cutoffs</u>

	Ac	TIVE	
Type of Cutoff	From	То	Description
Premature Commit	Power ON	Time for Commit	This circuit will detect premature voltage on the +1D14 bus.
Sequence Failure	Time for Ignition	Ignition Command	This cutoff will come at "time for ignition" if any one or more of the prerequisites in the "ready for ignition" chain is not met.
S-I Voltage Failure	Ignition Command	Сомміт	This circuit detects loss of the S-I inverter bus.
S-IV FAILURE	Ignition Command	Сомміт	THIS CUTOFF WILL BE INITIATED IF ANY ONE OR MORE OF THE PREREQUISITES FOR "S-IV READY TO LAUNCH" ARE NOT MET BY THE TIME S-I IGNITION COMMAND OCCURS.
IU FAILURE	IGNITION Command	Сомміт	This cutoff will be initiated if any of the prerequisites in the "IU Ready" chain are not met by the time S-I ignition occurs.

	Ac	TIVE	
Type of Cutoff	From	То	Description
Fire Detection	IGNITION COMMAND	Сомміт	This circuit will initiate cutoff if thermocouples mounted on the heat shield detect a predetermined rate of heat rise.
Rough Combustion	Ignition Command	Сомміт	THIS CIRCUIT WILL DETECT ROUGH COM- BUSTION OF ANY ENGINE, CUTOFF THIS ENGINE IMME- DIATELY AND INI- TIATE SEQUENCE CUT- OFF TO THE REMAIN- ING ENGINES.
THRUST FAILURE	Time for Commit	Сомміт	This circuit will initiate cutoff if any one or more of the S-I engines is not running by "Time for Commit"
Launch Failure	Ignition Command	Liftoff	THIS CIRCUIT WILL INITIATE CUTOFF 8 SECONDS AFTER IGNITION IF THE VEHICLE HAS NOT LIFTED OFF BY THIS TIME. (NOMINAL TIME IS APPROXI- MATELY 3.4")

4. PRIMARY (60 CYCLE AC) POWER FAILURES

A MOMENTARY LOSS OF AC POWER (CAPE CRITICAL OR INDUSTRIAL) WILL

CAUSE ALL GENERATORS TO DROP OUT; HOWEVER, BACKUP BATTERIES

(160 AMPERE-HOUR) ACROSS THE GENERATOR OUTPUT LINES PROVIDE

28 VDC POWER FOR MAINTAINING THE VEHICLE IN A SAFE CONDITION.

DURING ANY EXTENDED LOSS OF AC POWER, ANY UNNECESSARY LOADS

WILL BE REMOVED FROM THE BUSES.

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TEST NO.

VEHICLE

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COMPLEX 37 COMMUNICATIONS

INTERCOM: (IC)

TO BE UTILIZED FOR "T" COUNT ANNOUNCEMENTS, OPERATIONAL AND EMERGENCY ANNOUNCEMENTS, AND FOR PAD NORCOM REQUEST ANNOUNCEMENTS.

PAGING: (PA)

TO BE UTILIZED FOR ADMINISTRATIVE PHONE CALL ANNOUNCEMENTS, "T" COUNT ANNOUNCEMENTS, AND PAGING SPECIFIC PERSONNEL TO A HEADSET CHANNEL.

NORCOM: (OIS)

THE TEST AND CHECKOUT OPERATIONAL COMMUNICATIONS
UTILIZED AS ASSIGNED OR INDICATED IN THE
PROCEDURE FOR THE TEST OPERATIONS. TEST
SUPERVISORY PERSONNEL SHOULD ALWAYS BE AVAILABLE
IN THE FOLLOWING CIRCUITS.
S-I CHIEF OPERATIONS ENGINEER (CHRYSLER) SI
S-IV CHIEF OPERATIONS ENGINEER (DOUGLAS) SIV
LAUNCH VEHICLE TEST CONDUCTOR (NASA-KSC) S
SPACECRAFT TEST CONDUCTOR (NASA-MSC) SC
VEHICLE TEST SUPERVISOR S, SRO

RANGE PHONE: (SRO)

A SPECIAL COMMUNICATIONS NETWORK IS ESTABLISHED BETWEEN THE TEST SUPERVISOR AND THE EASTERN TEST RANGE SUPERVISOR OF RANGE OPERATIONS. THIS CIRCUIT WILL BE UTILIZED FOR ALL TEST RELATED COMMUNICATIONS BETWEEN THE LAUNCH COMPLEX AND CENTRAL CONTROL. IT WILL BE BUSSED TO THE OIS AS REQUIRED BY THE TEST PROCEDURE OR AT THE DISCRETION OF THE TEST SUPERVISOR.

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NORCOM CHANNEL ASSIGNMENTS

	CHANNEL		PRIMARY FUNCTION
	RED 1	(RD1)	S-I VEHICLE MECHANICAL
	RED 2	(RD2)	S-IV VEHICLE MECHANICAL
ŧ	RED 3	(RD3)	ALL LAUNCH VEHICLE GSE (MECHANI
	RED 4	(RD4)	PROPELLANTS
	YELLOW 1	(YW1)	GYRO AND LAYING
	YELLOW 2	(YW2)	LAUNCH VEHICLE CONTROL
	YELLOW 3	(YW3)	GUIDANCE AND RCA 110 COMPUTER
	YELLOW 4	(YW4)	PAD SAFETY, TV AND COMMUNICATIO
	YELLOW 5	(YW5)	(NOT ASSIGNED)
	BROWN 1	(BR1)	S-I MEASURING AND BH MONITOR CI
	BROWN 2	(BR2)	S-IV MEASURING
	BROWN 3	(BR3)	FACILITY AND GSE MEASURING
	BROWN 4	(BR4)	SERVICE STRUCTURE MEASURING STA
•	BLUE 1	(BL1)	NASA GROUND ELECTRICAL NETWORK
	BLUE 2	(BF5)	DAC GROUND ELECTRICAL NETWORK
•	BLUE 3	(BL3)	S-I, IU VEHICLE NETWORK
	BLUE 4	(BL4)	DAC S-IV VEHICLE NETWORK
	GRAY 1	(GY1)	TELEMETER S-I AND S-IV
	GRAY 2	(GY2)	ALL OTHER RF VEHICLE SYSTEMS
	GREEN 1	(GN1)	SC TEST PROJECT ENGINEER
	GREEN 2	(GN2)	SC LAUNCH ESCAPE SYSTEM/ORDNANC
	GREEN 3	(GN3)	MSC QUALITY CONTROL
	GREEN 4	(GN4)	NAA QUALITY CONTROL
	GREEN 5	(GN5)	(NOT ASSIGNED)
V.C. E.O.D	3-81 C (6/64)		

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WHITE 3	(WH3)	SC FACILITIES
WHITE 4	(WH4)	SC ENVIRONMENTAL CONTROL
WHITE 5	(WH5)	SC ELECTRICAL POWER SYSTEM
PURPLE 1	(PR1)	SC INSTRUMENTATION
PURPLE 2	(PR2)	SC RF SYSTEMS
PURPLE 3	(PR3)	SC GROUND SUPPORT EQUIPMENT
*S-I (WHITE 1	(SI)	S-I, IU TEST CONDUCTOR
*S-IV (WHITE	2) (SIV)	S-IV TEST CONDUCTOR
SC	(SC)	SPACECRAFT TEST CONDUCTOR
*S	(S)	SPACE VEHICLE TEST SUPERVISOR

* CHANNELS LOCATED ON "ALL" END UNITS

KSC FORM 28-81C (6/64)

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OPERATING STATIONS CODE IDENTIFICATION

BLOCKHOUSE CONSOLES

C1CM	S-I CAMERA PANEL
C1CT	S-I COMPONENT TEST PANEL
CIDP	S-I DESTRUCT PANEL
C1FP	S-I FIRING PREP PANEL
C1FR	S-I FIRING PANEL
C1LS	LAUNCH SEQUENCER PANEL
C1MP	S-I MEASURING PANEL
CINP	S-I NETWORKS PANEL
C1OP	S-I ORDNANCE MONITOR PANEL
C1TC	S-I TEST CONDUCTOR
CIVP	S-I VEHICLE PRESSURE PANEL
C4DP	S-IV DESTRUCT PANEL
C40P	S-IV ORDNANCE MONITOR PANEL
CATC	S-IV TEST CONDUCTOR
CATE	S-IV ASTRIONICS TEST ENGINEER
CDCG	DC GENERATOR PANELS
CDIR	DIRECTOR KSC
CETE	S-IV ELECTRICAL EQUIPMENT TEST ENGINEER
CGCM	GROUND CAMERA PANEL
CINV	INVERTER PANELS
CIPC	S-IV INSTRUMENTATION POWER CONTROL AND MONITOR PANEL
CISP	INSTRUMENTATION STATUS PANEL
CLTC	LAUNCH VEHICLE LEST CONDOCTOR
CLVN	VEHICLE NETWORKS PANEL
CNTS	NASA TEST SUPERVISOR
CPER	PERISCOPE
CPPC	S-IV PROPULSION SYSTEM PREPARATION AND CONTROL PANEL
CPSC	S-IV PNEUMATIC SYSTEM CONTROL PANEL
CPSO	PAD SAFETY OFFICER
CPSP	POWER SWITCHING PANEL
CPTE	S-IV PROPULSION TEST ENGINEER
CSPC	S-IV STAGE POWER CONTROL AND MONITOR PANEL
CSRP	SEQUENCE RECORDER READOUT PANEL
CSSP	S-IV STAGE SYSTEM STATUS PANEL
CSTC	SPACECRAFT TEST CONDUCTOR
CUMP	IU MEASURING PANEL
	IU NITROGEN CONTROL PANEL IU NETWORKS PANEL
CVMP	S-IV VACUUM MONITOR PANEL
CVTE	S-IV VACOUM MONITOR PANEL S-IV VEHICLE DESIGN TEST ENGINEER
CVIE	amin abuinds neathin lear cindings.

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BLOCKHOUSE RACKS

COMMSEMROCAMOCMO 4 A A B C C C C C C C C C C C C C C C C C	HYDRAULIC CONTROL AND MONITOR PANEL AUTOMATIC THEODOLITE CONTROL BLOCKHOUSE MEASURING COMBUSTION STABILITY MONITOR ENVIRONMENTAL CONTROL SYSTEM S-IV FLIGHT CONTROL ENGINEER FIRE DETECTION MONITOR S-IV FLIGHT SEQUENCE RECORDER GUIDANCE COMPUTER CONTROL GROUND PRESSURE PANEL GROUND FACILITY MEASURING LIFTOFF CLOCK LH2 CONTROL LH2 MONITOR LOX COMPUTER
RRRRRRRRRRRRRRRRRRRRRRRRRRRRRRRRRRRRRR	LOX CONTROL S-IV PROPELLANT LOADING CONTROL PROPELLANT RECORDER READOUTS PROPELLANT SEQUENCE RECORDERS S-IV PROPELLANT UTILIZATION PANEL RP-1 COMPUTER RP-1 CONTROL ST-124 CONTROL PANEL S-IV STRIP CHART RECORDERS S-I AND IU SEQUENCE RECORDER TRAFFIC LIGHT CONTROL PANEL UMBILICAL ARM CONTROL VEHICLE FLIGHT CONTROL S-I AND IU VISICORDER WATER CONTROL PANEL S-IV VISICORDER

SERVICE STRUCTURE, LAUNCH PEDESTAL AND VEHICLE

```
S124
         ST-124 IU
         FLIGHT CONTROL S-I ENGINE
S1FC
SINA
         NETWORKS UNIT 1-9
SINF
         NETWORKS UNIT 10-12
S1TM
         S-I TM
         S-I VEHICLE MECHANICAL PLATFORM NO. 1
S1VM
         S-IV VEHICLE ASTRIONICS ENGINEER
S4AE
S4CC
         S-IV CREW CHIEF
S4EE
         S-IV VEHICLE ELECTRICAL EQUIPMENT ENGINEER
S4FC
         FLIGHT CONTROL S-IV STAGE
         S-IV MECHANICAL EQUIPMENT ENGINEER
S4ME
         S-IV ORDNANCE ENGINEER
S40E
S4PE
         S-IV PROPULSION ENGINEER
S4SE
         S-IV STRUCTURES ENGINEER
         S-IV VEHICLE DESIGN ENGINEER
S4VE
         ENGINE SERVICE PLATFORM
SESP
SGNP
         GROUND NETWORKS ON PEDESTAL
SGMP
         GSE MECHANICAL ON PEDESTAL
SHZN
         HORIZON SENSOR
SMRP
         S-IV MATERIAL RESEARCH AND PROCESS ENGINEER
SMTE
         S-IV MECHANICAL TEST ENGINEER
         OVERALL TEST BATTERY ROOM
SORR
         S-IV OPERATIONS ENGINEER
SOPS
SQFC
         FLIGHT CONTROL Q-BALL
SSM1
         SHORT CABLE MAST NO. 1
         SHORT CABLE MAST NO. 2
SSM2
SSMS
         STRUCTURE MEASURING STATION
SSSC
         SERVICE STRUCTURE CONTROL
SUFC
         FLIGHT CONTROL IU
SUGU
         GUIDANCE IU
SUTM
         IU TM
SUVM
         IU VEHICLE MECHANICAL PLATFORM NO. 3
SUVN
         NETWORKS IU
SVME
         VEHICLE MEASURING
SVMP
         VEHICLE MECHANICAL ON PEDESTAL
         VEHICLE NETWORKS OAT CONSOLE
SVOC
SVOR
         VEHICLE NETWORKS RECORDERS
SVRF
         VEHICLE RF
```

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AGCS

A4AE	S-IV AGCS ELECTRICAL
AEBR	EMERGENCY BATTERY ROOM
AGAA	ST-124 ALIGNMENT AMPLIFIER
ANAG	AGCS NETWORKS
APCD	PNEUMATIC CONTROL DISTRIBUTOR
AECS	ECS ON AGCS ROOF

BLOCKHOUSE FIRST FLOOR

B1TM B4TM	S-1 BLOCKHOUSE TELEMETER STATION S-IV BLOCKHOUSE TELEMETER STATION
BCDC	COUNTDOWN CLOCK
BCOM BDOP	COMMUNICATIONS AND TV CONTROL ODOP GROUND STATION
BGRN	LCC NETWORKS
BRCA	RCA SEQUENCER
BRFS BSOC	BLOCKHOUSE RF STATION SATURN OPERATIONAL COMPUTER

UMBILICAL TOWER

U4PE	S-IV PROPULSION ENGINEER
UECS	ECS ON UMBILICAL TOWER
UGRN	UMBILICAL GROUND NETWORKS
UGSE	UMBILICAL GSE
UPNC	S-IV PNEUMATIC CONSOLES A AND B
UGBL	Q-BALL GSE
USA1	UMBILICAL SWING ARM NO, 1 CONTROL PANEL
USA2	UMBILICAL SWING ARM NO. 2 CONTROL PANEL
USA3	UMBILICAL SWING ARM NO. 3 CONTROL PANEL
USA4	UMBILICAL SWING ARM NO. 4 CONTROL PANEL

PAD AND OTHER COMPLEX AREAS

PALN	VEHICLE AL	IGNMENT STATIONS
PATB	AUTOMATIC '	THEODOLITE BLDG.
PHPF	HIGH PRESSI	JRE FACILITY
PLHF	LH2 FACILI	ΓY
PLOF	LOX FACILI'	ΓY
PPSO	PSO ON PAD	
PRPF	RP-1 FACIL	[TY
PVP1	VALVE PIT I	NO. 1

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STATIONS EXTERNAL TO COMPLEX 37

ORNG OTMD

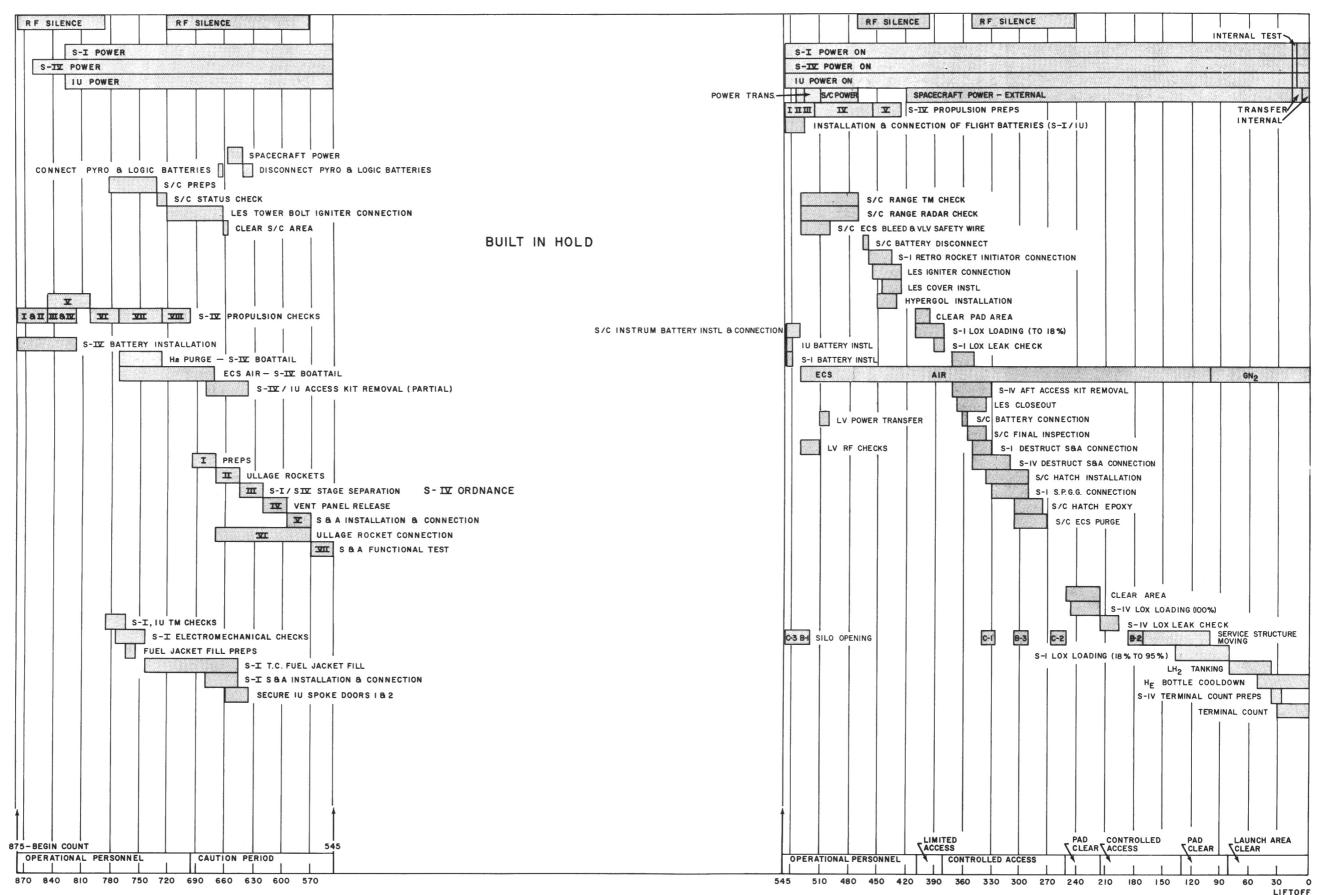
DATE: SEPTEMBER 5, 1964

REVISION

OCCF CONVERTER COMPRESSOR FACILITY RANGE OPERATIONS THROUGH THE SRO

TELEMETER HANGAR D STATION

PRELAUNCH PREPARATIONS — L-I DAY PART ™



PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
					PART IV	
	v.				LAUNCH PRECOUNT	
·875¹						
	SRO	1	CNTS	ORNG	VERIFY COMPLEX IS ON CRITICAL POWER.	
		2		S1VM SUVM	OPEN VEHICLE.	
	BL1	3		CINV	GROUND INVERTER 2 AND 60 KW GENERATOR	
		4		SVME	"ON". REMOVE ALL PNEUMATIC CALIBRATION LINES AND SAFETY WIRE CALIBRATION VALVES.	
		5		APCD	PREPARE THE PNEUMATIC DISTRIBUTION SYSTEM FOR PRELAUNCH OPERATIONS PER PROCEDURE LV0=E=7015.	
	PA	6		CNTS	RF SILENCE "ON".	
	RD3	7		RWCP	PREPARE WATER SYSTEM PER PROCEDURE LVO-E-7014, PART I.	
	RD4	8		PLHF	SET UP LH2 TRANSFER SYSTEM PER PROCEDURE LVO-L-1034.	
	SIV	9	C4TC	CSPC	APPLY POWER TO S-IV PER PROCEDURE DAC-N-7001.	
	SIV	10	C4TC		VERIFY UMBILICAL TOWER WATER SYSTEM "ON".	
	BL4	11	C4TC	CATE	BEGIN S-IV BATTERY INSTALLATION PER PROCEDURE DAC-N-7005.	
	RD2	12	C4TC	CPTE	BEGIN S-IV PROPULSION PRELIMINARY PREPS PER PROCEDURE DAC-P-7005, PART I.	
	RD1	13	S4PE		REMOVE CHILLDOWN DUCT DESSICATORS AND STUB FIN MAST ADAPTERS.	
	RD2	14	C4TC	CPTE	PERFORM PROPELLANT LINES VACUUM CHECK PER DAC=P=7005, PART II.	
	RD3	15		RECS	PERFORM ECS PREPARATIONS AND OPERATIONS PER PROCEDURE LVO=E-7018.	

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TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-875°						
	GY1	16	B1TM	SUTM S1TM B1TM	PREPARE S1 & IU TELEMETER SYSTEMS PER PROCEDURE LVO-R-3001.	
	GY2	17	BRFS	SVRF	PREPARE S1 & IU RF SYSTEMS PER PROCEDURE LVO-R-3002.	
	WH2	18	C4TC	S4AE	PERFORM RF & TM SYSTEMS PRE-TEST PREPARATIONS PER PROCEDURE LVO-R-3003.	
-865°						
	SIV	1	C4TC	SMTE	VERIFY QUICK DISCONNECTS AND UMBILICAL CARRIERS PROPERLY INSTALLED, (SWING ARM NO. 2 AND GH2 VENT COUPLING).	
					NOTE	
					STEEL SHEAR PINS, SAFETY BLOCKS AND SAFETY WIRE INSTALLED, LANYARDS NOT CONNECTED, AND EJECT ACTUATOR PNEUMATIC LINES NOT CONNECTED.	
	S	2	CLTC	C4TC	VERIFY POWER ON S-IV.	
	SIV	3	C4TC		ALL STATUS SELECT SWITCHES TO "ON SCHEDULE",	
-845 '						
	RD2	1	C4TC		VERIFY COMPLETION OF S-IV PROPULSION PRELIMINARY PREPS PER DAC-P-7005, PART I.	
	RD2	2	C4TC		VACUUM PUMP HELIUM HEATER AND ENGINE IGNITERS PER PROCEDURE DAC=P=7005, PART III.	
	KD2	3	C4TC	CPTE	PERFORM LH2 AND LOX SYSTEMS PURGES AND SAMPLING PER PROCEDURE DAC=P=7005, PART V.	
	RD2	4	C4TC		PERFORM BULKHEAD SAMPLING PER PROCEDURE DAC-P-7005, PART IV.	

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TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-830 9						
~825°	,	1		S1NF SVOC	VERIFY DESTRUCT MONITOR SYSTEM READY.	
	D1.4			C4 ND	ADDLY DOUGD TO C T	
	BL1	1		C1NP	APPLY POWER TO S-I.	
	BL1	2		CUNP	APPLY POWER TO IU.	
		3		SVMP	CHECK MAIN FUEL VALVES FOR LEAKAGE AND RECORD IF LEAKAGE IS NOTED.	
		4		SVMP	INSTALL FUEL DRAIN SCREWS/WASHERS.	
		5		SVMP	CHECK THAT THE CONTROL 750 REGULATOR IS STILL ADJUSTED TO 750 PSIG.	
	BL1	6	,	CPSP	SERVICE STRUCTURE OAT POWER "ON".	
	BL1	7		SVOC	PULSE SENSOR POWER "ON".	
	BL1	8		SVOC	TEST DESTRUCT PULSE SENSORS AND RESET.	
	BL1	9	C4TC	S1NA	BEGIN NO VOLTAGE AND CONTINUITY CHECKS ON TURBINE SPINNER INITIATOR CIRCUITS PER PART I OF PROCEDURE 1-LSII-709.	
-820°						
	RD1	1.	C1CT		VERIFY SOI LOX TANK PRESSURE MEASURE- MENTS INDICATE TANK PRESSURE BEFORE VENTING.	
~815°						
	BL1	1			STANDBY TO TURN IGNITION CIRCUIT BREAKER AND CONTACTOR "ON".	
	RD1	2	C1TC		PREPARE TO RUN ELECTRO-MECHANICAL TEST PER PROCEDURE LVO-MV-P-1006.	
		3		SHZN	HORIZON SENSOR CHECKS,	
	RD2	4	C4TC		VERIFY COMPLETION OF HELIUM HEATER AND ENGINE IGNITER VACUUM PUMPING PER DAC-P=7005, PART III.	

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TIME	COMM.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-815'						
	RD2	5	C4TC	CPTE	VERIFY COMPLETION OF PROPELLANT LINES VACUUM CHECK PER DAC-P-7005, PART II	
	BL4	6	C4TC	CATE	VERIFY COMPLETION OF S-IV BATTERY INSTALLATION PER DAC-N-7005.	
	RD2	7	C4TC	CPTE	VERIFY COMPLETION OF BULKHEAD SAMPLING PER DAC-P=7005, PART IV.	
-8000						
	RD2	1	C4TC	CPTE	VERIFY COMPLETION OF LH2 AND LOX SYSTEM PURGES AND SAMPLING PER DAC-P-7005. PART V.	
	RD2	2	C4TC	CPTE	PERFORM PROPELLANT VALVES AND IGNITERS FUNCTIONAL CHECK PER PROCEDURE DAC-P-7005, PART VI.	
-790°						
	SRO	1	CNTS	ORNG	DO NOT INTERROGATE C-BAND BEACON WITH 1.16 RADAR UNTIL REQUESTED.	
-785°						
	BL1	1	C1TC	S1NA	VERIFY NO VOLTAGE AND CONTINUITY CHECKS COMPLETE ON S.P.G.G. INITIATOR CIRCUITS PRIOR TO RF CHECKS.	
		2	C1TC	C1NP	RESET IGNITION SEQUENCER	
		3	CATE	S4CC	START FORWARD INTERSTAGE AUXILIARY BLOWER.	
	SRO	4	CNTS		RF CLEARANCE FOR ALL SYSTEMS AND VERIFY READY FOR READOUTS.	
	SRO	5	CNTS		DO NOT INTERROGATE C-BAND BEACON WITH 1.16 RADAR UNTIL REQUESTED.	
		6			RF AND TM CHECKS.	
	PA	6-1		CNTS	RF SILENCE "OFF".	

SATURN/APOLLO

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TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-785°						
	S	6=2	CLTC	C1MP	TELEMETERS F1, F2, F3, P2, S1 AND S2 "ON",	
	S	6-3	CLTC	CUMP	TELEMETERS F5, F6, S3 AND P1 "ON".	
	S	6-4	CLTC	C1MP,	AUXILIARY EQUIPMENT "ON".	
	S	7	CLTC	C1MP CUMP	MEAS. ON DDAS "ON".	,
	S	7-1	CLTC	C1MP CUMP	SWEEP FREQUENCY CALIBRATOR "ON".	
	S	7-2	CLTC	BDOP	ODOP GROUND TRANSMITTER NO. 2 "ON".	
	S	7-3	CLTC	CUMP	ALTIMETER "ON".	
	S	7-4	CLTC	CUMP	GUIDANCE COMMAND RECEIVER "ON".	
	S	7-5	CLTC	CUMP	ODOP "ON",	
	S	7-6	CLTC	CUMP	MINITRACK "ON".	
	S	7-7	CLTC	CUMP	AZUSA "QN",	
	S	7-8	CLTC	CUMP	C.BAND BEACON (RADAR) TO "ON".	
	S	7=9	CLTC	CUMP	MISTRAM "ON",	
	SRO	7 ~ 1.0	CNTS	ORNG	CHECK MISTRAM AND AZUSA AND REPORT READOUT.	
	S	7-11	CLTC	CIPC	TELEMETER D1. D2. AND D3 "ON".	
	S	7-12	CLTC	C1MP CUMP CIPC	VERIFY TAPE RECORDER READY INDICATIONS.	
	S	7=13			TELEMETER CALIBRATIONS TO "PREFLIGHT".	
	S	7-14	CLTC		CALIBRATION SWITCH TO MANUAL (VERIFY VCO CALIBRATION LIGHT OUT),	
	S	7-15		C1MP CUMP	PRE-FLIGHT CALIBRATION TO 50%.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-785°						
	S	7-16	CLTC	B4TM	TELEMETER RECORDINGS "ON".	
	S	7-17	CLTC	B4TM	PERFORM TELEMETER CHANNEL READOUTS.	
					NOTE	
					VERIFY CHANNEL 21=24 OF TELEMETER D2 HIGH LEVEL PDM READING 100%.	
-782°						
		1			RF AND TM CHECKS,	
	S	1-1	CLTC	OTMD	TELEMETER RECORDINGS "ON",	
	S	1-2	CLTC	ORNG	COMMAND CARRIER "ON".	
	S	1-3	CLTC	RBHM	SANBORN RECORDER TO 2 MM/SEC.	
	S	1-4	CLTC	CSSP	VERIFY MSFC TALKBACK ENABLE.	
	S	1-5	CLTC	CLVN	DESTRUCT ENABLE "ON",	
	S	1-6	CLTC	C4DP	VERIFY SAFETY BUS INDICATION "ON".	
	S	1-7	CLTC	C1DP	Sel CDR'S NO. 1 AND NO. 2 "ON".	
	S	1-8	CLTC	C10P	MODULE POWER SUPPLY "ON",	
	S	1=9	CLTC	C10P	FUNCTION SELECTOR TO FIRING UNIT VOLTAGE CHECK.	
	S	1-10	CLTV	C40P	PANEL SELECT TO "AUXILIARY RECORDER".	
	S	1-11	CLTC	R4VC	VISICORDER MONT.	
	S	1-12	CLTC	RFSR	SEQUENCE RECORDERS TO 10 MM/SEC.	
	S	1-13	CLTC	C4DP	S=IV CDR #1 AND #2 "ON".	
	S	1-14			VERIFY EBW°S #1 AND #2 "OFF" AND "NOT READY",	
	S	1-14		C4DP		

PAGE TEST NO. VEHICLE

REVISION					VEHICLE	SA=
TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-780°						
		1			RF AND TM CHECKS.	
	SRO	1~1	CNTS	ORNG	INTERROGATE IU AND SPACECRAFT C-BAND BEACONS WITH 1.16 RADAR AND 0.18 RADAR RESPECTIVELY.	
	S	1-2	CLTC	RSEQ	S=I RECORDERS TO MINUTE SPEED AND TIME PULSE "ON".	
	S	1-3	CLTC	C1CM	TELEVISION TO "FILAMENT".	
	S	1-4	CLTC	C4DP	S=IV CDR #1 AND #2 TO "INTERNAL",	
	S	1-5	CLTC	RBHM	SANBORN RECORDER TO 25 MM/SEC.	
	S	1-6	CLTC	RSEQ	RECORDERS FAST.	
	S	1-7	CLTC	ORNG	DESTRUCT COMMAND (MOMENTARILY THROUGH CUTOFF).	
	S	1-8	CLTC	RSEQ	RECORDERS SLOW.	
	S	1-9	CLTC	RBHM	SANBORN RECORDER TO 5 MM/SEC.	
	S	1-10	CLTC	C4DP	VERIFY S-IV EBW NO, 1 AND NO, 2 GO TO INTERNAL AND CHARGE.	
	S	1-11	CLTC	C1DP	VERIFY S-I EBW'S NO. 1 AND NO. 2 LATCH INTERNAL AND CHARGE ON GROUND POWER.	
	S	1-12	CLTC	CNTS	VERIFY CUTOFF COMMAND FOR 2.0 SECONDS.	9
	S	1-13	CLTC	CNTS	VERIFY DESTRUCT COMMAND 2.0 SECONDS AFTER CUTOFF RECEIVED.	
	S	1-14	CLTC	svoc	VERIFY SRI AND S-IV DESTRUCT DOES NOT OCCUR AT THE VEHICLE.	
	S	1-15	CLTC	ORNG	RELEASE DESTRUCT COMMAND.	
	S	1-16	CLTC	C4DP	S-IV CDR SWITCHES NO. 1 AND NO. 2 "OFF" (VERIFY CDR'S REMAIN ON).	
	S	1-17	CLTC	ORNG	"SAFE" COMMAND AND RELEASE UPON REQUEST	
	S	1-18	CLTC	C4DP	VERIFY "SAFE" COMMAND.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-780 9						
	S	1-19	CLTC	C4DP B4TM	VERIFY Solv CDR #1 AND #2 AND EBW #1 AND #2 REMAIN "ON".	
	S	1-20	CLTC	C1DP C4DP	EBW #1 AND #2 TO "EXTERNAL" (VERIFY "OFF" AND "NOT READY").	
	S	1-21	CLTC	C4DP	S-IV CDR S #1 AND #2 TO "EXTERNAL" AND VERIFY "OFF"	
	S	1=22	CLTC	C1DP	S=I CDR NO. 1 AND NO. 2 "OFF".	
	S	1=23	CLTC	CLVN	DESTRUCT ENABLE "OFF".	
	S	1-24	CLTC	RFSR	SEQUENCE RECORDERS TO 2 MM/SEC.	
	S	1~25	CLTC	R4VC	VISICORDER **OFF**.	
		1-26	CLTC	врор	ODOP GROUND TRANSMITTER #2 **OFF** AND #1 **ON**	
	S	1-27	CLTC	RSEQ	RECORDERS TO HOUR SPEED, TIME PULSE "OFF",	
	S	1-28	CLTC	RBHM	SANBORN RECORDER TO 2 MM/SEC.	
	SRO	1-29	CNTS	ORNG	COMMAND CARRIER "OFF",	
	SRO	1 ≂30	CNTS		REPORT COMMAND CARRIER DEVIATION, ANTENNA, ANTENNA ORIENTATION, AND POWER OUTPUT.	
	RD3	2	CLTC		VERIFY COMPLETION OF PNEUMATIC PREPS FOR PRELAUNCH PROCEDURE LVO-E-9015.	
	SC	3	CNTS	CSTC	PERFORM SPACECRAFT PREPARATIONS.	
-776°						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	CIPC	MANUAL CALIBRATION TO 0 VOLTS.	
	S	1-2	CLTC	B1TM	TELEMETER RECORDINGS "ON",	
	S	1-3	CLTC	C1MP CUMP	TELEMETER CALIBRATION TO "AC COMMAND".	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-775 ! 15 "						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS "ON" AND "OFF",	
775 1						
		1			RF AND TM CHECKS.	
	S	1=1	CLTC	C1MP CUMP	TELEMETER CALIBRATION TO PREFLIGHT.	
	S	1-2	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMAND.	
	S	1-3	CLTC	C1MP CUMP	PREFLIGHT CALIBRATION TO 0%.	
	S	1-4	CLTC	CIPC	MANUAL CALIBRATION TO +2.5 VOLTS.	
	S	1~5	CLTC	C1CM	TELEVISION TO "B#",	
	RD1	2		C1CT C1FP S1VM	VENT LOX TANK STANDBY PRESSURE.	
	RD1	3		C1TC SVMP SGMP	PERFORM ELECTRO-MECHANICAL TEST PER PROCEDURE LVO-MV-P-1006.	
7741						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	CIPC	MANUAL CALIBRATION TO *5 VOLTS.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
*773° 30"						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1MP CUMP	PREFLIGHT CALIBRATIONS TO 100%.	
	S	1-2	CLTC	CIPC	MANUAL CALIBRATION TO 0 VOLTS.	
-772 º 30 "						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1MP CUMP	RECORDER TRANSFERS "ON".	
	S	1-2	CLTC	B1TM OTMD	8 KC OSCILLATORS "ON",	
	S	1-3	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS.	
	S	1-4	CLTC	CIPC	MANUAL CALIBRATION TO LIGHTS OFF STEP.	
	S	1=5	CLTC	C1MP CUMP CIPC	TAPE RECORDERS TO RECORD.	
	S	1 6	CLTC		PREFLIGHT CALIBRATIONS TO 0%, 25%, 50%. 75%, AND 100% IN 2 SECOND INCREMENTS.	
9	s	1-7	CLTC	CIPC	STEP THROUGH MANUAL CALIBRATION IN 2 SECOND INCREMENTS.	
	S	1=8	CLTC	C1MP CUMP CIPC	TAPE RECORDERS #OFF#.	
	S	1-9		C1MP CUMP	TELEMETER CALIBRATION TO INFLIGHT.	
	s	1-10	CLTC	CIPC	CALIBRATION TO AUTOMATIC.	
	s	1-11	CLTC	CIPC	GIVE CAL START.	
	S	1≈12			TELEMETER CALIBRATION COMMAND "ON" FOR 5 SECONDS.	

SATURN/APOLLO

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSI STA.	E DESCRIPTION	REMARKS
-772° 30"						
	S	1-13	CLTC	CIPC	BRIDGE CALIBRATE HIGH FOR 10 SECONDS.	
	S	1-14	CLTC	CIPC	BRIDGE CALIBRATE LOW FOR 10 SECONDS.	
	S	1-15	CLTC	CUMP	P1 TELEMETER CALIBRATION.	
	S	1-16	CLTC	RBHM	S=I AND IU HIGH CALIBRATE COMMAND (15 SEC.).	
	S	1-17	CLTC	RBHM	S=I AND IU LOW CALIBRATE COMMAND (15 SEC.).	
	S	1=18	CLTC	RBHM	S=I AND IU RUN COMMAND.	
	S	1-19	CLTC	B1TM OTMD	8 KC OSCILLATOR "OFF".	
≂770°						
	RD2	1	C4TC	CPTE	VERIFY COMPLETION OF PROPELLANT VALVES AND IGNITERS FUNCTIONAL CHECK PER DAC-P-7005, PART VI.	
	RD2	2	C4TC	CPTE	PERFORM SaIV ENGINE AND HELIUM HEATER PURGE PER PROCEDURE DAC-P-7005, PART VII,	
					NOTE	
					THERE WILL BE NO ACTIVITY IN THE S-IV AFT INTERSTAGE UNTIL THE COMPLETION OF PART VII OF PROCEDURE DAC-P-7005.	
	RD3	3	CPTE		PROVIDE ECS AIR TO S-IV AFT SECTION DURING HELIUM PURGE OPERATIONS AS REQUESTED.	
7691						
		1			RF AND TM CHECKS	
	S	1-1		C1MP CUMP	RECORDER TRANSFERS "OFF".	
	S	1-2	CLTC	CIPC	GIVE CAL START.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
~769°						
	S	1-3	CLTC	C1MP	TELEMETER CALIBRATION COMMANDS "ON" FOR 5 SECONDS.	
	S	2	CLTC	CUMP	P1 CALIBRATION COMMAND "ON" AND "OFF".	
	S	2-1	CLTC	C1MP CUMP CIPC	TAPE RECORDERS TO "PLAYBACK".	
	S	2-2	CLTC	CUMP	C-BAND BEACON (RADAR) "OFF" WHEN READOUT COMPLETE.	
	S	2-3	CLTC	CUMP	AZUSA "OFF" (WHEN READOUT IS COMPLETE).	
-766° 30"						
		1			RF AND TM CHECKS.	
	S	1 == 1	CLTC	C1MP CUMP CIPC	CONFIRM TAPE RECORDER READY INDICATIONS (BEFORE PROCEEDING).	
	S	1-2	CLTC	C1MP CUMP CIPC	TAPE RECORDERS MOFFM.	
	S	1=3	CLTC	CIPC	VERIFY VCO CALIBRATION "OFF" AND "AUTO"	
	S	1-4	CLTC		TELEMETERS F1, F2, F3, P2, S1 AND S2 MOFFM,	
	S	1-5	CLTC	CUMP	TELEMETERS F5, F6, S3 AND P1 "OFF".	
	S	1-6	CLTC	C1MP CUMP	SWEEP FREQUENCY CALIBRATOR "OFF".	
	S	1=7	CLTC	C1MP CUMP	AUXILIARY EQUIPMENTS "OFF".	
	S	1-8	CLTC	CIPC	TELEMETERS D1, D2 AND D3 "OFF",	
	S	1⇔9	CLTC	B1TM B4TM OTMD	TELEMETER RECORDINGS "OFF".	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-766° 30"						
	S	1-10	CLTC	CUMP	ALTIMETER "OFF".	
	S	1-11	CLTC	C1CM	TELEVISION MOFFM.	
		2			RF AND TM CHECKS.	
	S	2-1	CLTC	CUMP	ODOP "OFF"	
	S	2-2	CLTC	CUMP	MINITRACK "OFF".	
	S	2-3	CLTC	CUMP	MISTRAM "OFF"	
	S	2 4	CLTC	CUMP	GUIDANCE COMMAND RECEIVER "OFF".	
	S	2=5	CLTC	BDOP	ODOP GROUND TRANSMITTER NO. 1 "OFF".	
	S	2=6	CLTC	B4TM	NORMALIZE ALL SWITCHES AND SECURE STATION.	
	RD1	3	C1TC	SVMP	BEGIN FUEL JACKET FILL PREPARATIONS.	
≈755°						
	BR1	1	C1TC	RFDM	VERIFY FIRE DETECTION PROBE COVERS ARE REMOVED.	
	BR1	2	C1TC	RÇSM	VERIFY CSM ACCELEROMETERS INSTALLED.	
	RD1	3	C1TC	SVMP	VERIFY COMPLETION OF FUEL JACKET PREPS.	
- 745 ⁰						
	RD1	1	CITC		VERIFY COMPLETION OF THE ELECTRO- MECHANICAL TEST PER LVO-MV-P-1006.	
		2			VERIFY THAT GEARCASE PRESSURE IS INDICATING BETWEEN 2 AND 10 PSIG AT EACH ENGINE POSITION.	
-740°						
		1			REMOVE PROTECTIVE COVERS FROM HEAT SHIELD PANEL MEASUREMENTS.	
		2			REMOVE TURBINE TACHOMETER PROTECTIVE COVER, 8 PLACES.	

PAGE TEST NO. VEHICLE

153 7=LSVI=300 SA=7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
≈730°						
	SC	1	CNTS	CSTC	VERIFY PREPARATIONS COMPLETE.	
			CIVIS			
	SÇ	2		CSTC	STATUS CHECK.	
₽725 ⁰						
	RD2	1	C4TC	CPTE	VERIFY COMPLETION OF S=IV ENGINE AND HELIUM HEATER PURGES PER DAC=P=7005, PART VII.	
	RD2	2	C4TC	CPTE	PERFORM S=IV PROPULSION SECURING PER PROCEDURE DAC=P=7005, PART VIII.	
		3	S4PE	SVMP	INSTALL H2 DUCT DESSICATORS	
-720°						
	PA	1		CNTS	RF SILENCE ON AND CONTROLLED SWITCHING.	
	SC	2		CSTC	VERIFY SPACE VEHICLE RF SILENCE AND COMPLEX 37 RF SILENCE AND ORDNANCE CREW ON STATION.	
×	SC	3		CSTC	BEGIN LES TOWER ORDNANCE CONNECTION PER C-10001.	
⊸695°						
	SIA	1	C4TC	CPTE	VERIFY COMPLETION OF S-IV PROPULSION SECURING PER DAC-P-7005, PART VIII.	
	RD1	2	C1TC	S1VM	BEGIN INSTALLATION OF FLIGHT CLOSURES ON THE FORWARD END OF FUEL TANKS 3 AND 4.	
		3		RECS	DISCONTINUE ECS AIR TO THE SEIV AFT SECTION AFTER OK FROM CATC.	
	RD3	4	CLTC	RUAÇ	PERFORM UMBILICAL SWING ARM PREPARA- TIONS FOR LAUNCH PER PROCEDURE LVO-E-7016.	
	RD3	5	C1TC	SGMP	VERIFY LIFTOFF AND SWING ARM CONTROL SWITCHES ARE CALIBRATED FOR LAUNCH AND PURGE HOSES ARE CONNECTED.	
					•	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-695°						
	RD3	6	C1TC	SGMP	PREPARE HOLDDOWN ARMS FOR LAUNCH PER PROCEDURE LVO-E-7017.	
	BL1	7	CNTS	ANAG	TURN BREAKER 66A5A10-1 MOFF" (Q-BALL POWER BREAKER)	
	SC	8		CNTS	VERIFY TO SPACECRAFT TO THE Q-BALL POWER BREAKER IS "OFF".	
	RD2	9	C4TC	CVTE	VERIFY COMPLETION OF S-IV ORDNANCE INSTALLATION PRELIMINARY CHECKS PER DAC-0-7011, PART I.	
	RD2	10	C4TC	CVTE	PERFORM ULLAGE ROCKET JETTISON DETONATOR INSTALLATION AND CONNECTION PER PROCEDURE DAC=0=7011, PART II.	
	RD2	11	C4TC	CVTE	PERFORM ULLAGE ROCKET INITIATORS INSTALLATION AND CONNECTION PER DAC-0-7011, PART VI.	
- 685°						
	RD1	1		S1VM	VERIFY STIV ENGINE CHAMBER SIDE AND EXIT COVERS INSTALLED AND REMOVE PROTECTIVE COVER FROM LOX/SOX PLENUM.	
-680 ₹						
	RD1	1	C1TC	S1VM	INSTALL SEI DESTRUCT SEA UNIT.	
	BL3	2	C1TC	SINF	PERFORM S&A UNIT CHECK PER PROCEDURE	
	GY2	3	BRFS	SVRF	SECURE S1 AND IU RF SYSTEMS FOR B.I.H. I-LSII=710. PART I.	
	SC	4		CSTC	VERIFY COMPLETION OF LES TOWER ORDNANCE CONNECTION BEGIN LES IGNITER INSTALLATION.	
-660 9						
	BR1	1	CLTC	SSMS	VERIFY MEASURING READY TO SECURE IU.	
	BL1	2	CLTC	SUVN	VERIFY NETWORKS READY TO SECURE IU.	
	BL1	2	CLTC	SUVN	VERIFY NETWORKS READY TO SECURE IU.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-660 '						
	RD1	3	CLTC	SUVM	INSTALL IU 70 INCH TUBE DOOR.	
	RD1	4	C1TC	S1VM	INSTALL DOORS ON INSTRUMENT CONTAINERS SECURE DOOR 1 FOR FLIGHT. SECURE DOOR 2 FOR BUILT IN HOLD.	
-650						
	RD1	1	CLTC	SUVM	BEGIN INSTALLATION OF IU DOORS AT TUBES 1 AND 2 FOR LAUNCH, TUBE 3 AND 4 FOR B,I,H,, AND STOW WORK STANDS,	
	RD1	2	C1TC	S1VM	VERIFY ALL BLAST PLATES ON S=I FORWARD ARE SECURED FOR LAUNCH.	
-645°						
	GY1	1	B1TM	S1TM SUTM	SECURE S1 AND IU TELEMETER SYSTEMS FOR B,I,H,	
	BR1	2	C1TC	SVME	VERIFY PRESSURE MEASUREMENTS FOR AIR BEARINGS OPEN AND SAFETY WIRED.	
	RD1	3	C1TC	S1VM	VERIFY FUEL TANK 3 AND 4 FORWARD CLOSURES ARE INSTALLED FOR FLIGHT.	
		4		S1VM	SET UP GN2 PURGE LINES AND HAND VALVES FOR RETROROCKET INITIATOR CONNECTION.	
	RD3	5	C1TC	SGMP	PERFORM SHORT CABLE MAST PREPARATIONS FOR LAUNCH PER PROCEDURE LVO-E-7013.	
	RD1	6	C1TC	S1VM	VERIFY COMPLETION OF S-I DESTRUCT S&A INSTALLATION.	
	RD2	7	C4TC	CVTE	VERIFY COMPLETION OF ULLAGE ROCKET JETTISON DETONATOR INSTALLATION AND CONNECTION PER DAC=0-7011, PART II.	
	RD2	8	C4TC	CVTE	PERFORM STAGE SEPERATION DETONATOR INSTALLATION AND CONNECTION PER PROCEDURE DAC=0=7011, PART III.	
		9		C1CT C1FP SVMP	APPLY LOX TANK STAND-BY PRESSURE.	

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TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-645						
	SC	10		CSTC	VERIFY COMPLETION OF LES IGNITER POST INSTALLATION CHECKS.	
	RD1	11	C1TC	CVMP	VERIFY COMPLETION OF THRUST CHAMBER FUEL JACKET FILL PER LVO-MV-P-1002.	
-6351						
	BR1	1		SVME	VERIFY ALL COVERS INSTALLED ON UMA	
	BR1	2		SVME	VERIFY CALIBRATE PLUGS ARE REMOVED FROM LIQUID LEVEL RACKS.	
	RD2	3	C4TC	CVTE	PERFORM S-IV ORDNANCE INSTALLATION PRELIMINARY CHECKS PER PROCEDURE DAC-0-7011, PART I.	
		4		RGCC	FLIGHT COMPUTER SYSTEM "OFF" AFTER VERIFICATION PROCEDURE 5-LIUI-749A IS COMPLETE.	
		5		CNTS	RELEASE SPACECRAFT PERSONNEL FROM STATIONS.	
-620 0						
	RD3	1	C1TC	SGMP	VERIFY THAT THE FUEL MAST AND LOX MAST ARE IN A STATE OF OPERATIONAL READINESS FOR LAUNCH AS FOLLOWS:	
					A. VERIFY THAT WEATHER PROTECTIVE TAPES AND COVERS HAVE BEEN REMOVED FROM THE MASTS.	
					B. AFTER VERIFYING WITH C1CT THAT THE RELEASE VALVE SWITCHES ARE IN THE "OFF" POSITION, CONNECT THE RETRACT CYLINDERS TO THE MASTS AND VERIFY THAT THE CONNECTING PINS ARE SECURED WITH BALLLOC PINS.	
					C. VERIFY THAT THE AUTOMATIC RETRACT LINES ARE CONNECTED AND THE FITTINGS ARE TIGHT.	
					D. VISUALLY INSPECT THE LOX AND FUEL MAST ASSEMBLIES FOR PROPER	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-620 °						
					MECHANICAL INSTALLATION AND NON-INTERFERENCE WITH ASSOCIATED HARDWARE.	
					E. VERIFY FUEL MAST PURGE LINE CONNECTED .	
				,	F. VERIFY VACUUM LINE SCREEN INSTALLED AT LAUNCHER LEG.	
	RD2	2	C4TC	CVTE	VERIFY COMPLETION OF STAGE SEPARATION DETONATOR INSTALLATION AND CONNECTION PER DAC-0-7011, PART III.	
	RD2	3	C4TC	CVTE	PERFORM VENT PANELS RELEASE DETONATORS INSTALLATION AND CONNECTION PER DAC-O-7011, PART IV.	
~595¹						
	BR1	1	C1TC	SVME	VERIFY ALL INSTALLED RADIATION SHIELD PNEUMATIC TUBING AND TEMPERATURE MEASUREMENTS ARE CONNECTED.	
		2		SVME	REMOVE COVERS FROM CALORIMETER FACES.	
	RD2	3	C4TC	CVTE	VERIFY COMPLETION OF VENT PANELS RELEASE DETONATORS INSTALLATION AND CONNECTION PER DAC-0-7011, PART IV.	
	RD2	4	C4TC	CVTE	PERFORM DESTRUCT S&A INSTALLATION AND CONNECTION PER DAC-0-7011, PART V.	
-570 9						
	RD1	1	C1TC		PERFORM THRUST CHAMBER FUEL JACKET FILL PER PROCEDURE LVO-MV-P-1002.	
=545 0						
	RD2	1	C4TC		VERIFY COMPLETION OF S-IV DESTRUCT S&A FUNCTIONAL TEST PER DAC-0-7011, PART VII, INCLUDING ORDNANCE SECURING OPERATION,	
		2	CNTS		VERIFY THAT S-IV FLIGHT S&A DEVICE IS ELECTRICALLY DISCONNECTED AND THAT DUMMY DEVICE IS CONNECTED.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
545 9						
		3			VERIFY THAT S-IV DUMMY S&A DEVICE IS CONNECTED, AND THAT FLIGHT S&A IS DISCONNECTED FROM S-IV ELECTRICAL	
		4		C1NP	REMOVE POWER FROM S-I.	
		5		CUNP	REMOVE POWER FROM IU.	
		6		RBHM	SANBORN RECORDER "OFF".	
		7		S1VM SUVM	SECURE VEHICLE FOR STANDBY.	
		1	CNTS		"BUILT IN HOLD",	
	S	1 = 1	CNTS	CSTC	PROCEED TO ARM AND SAFE THE LES.	
		1-2	C4TC		PRESSURIZE REMAINING S=IV FORWARD ELECTRONIC BOXES.	
	SRO	1=3	CNTS		LOAD COMPLEX CAMERAS AND CHECKOUT TO BE COMPLETED PRIOR TO RESUMPTION OF COUNT,	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
					PART V	
					LAUNCH COUNTDOWN	
					PRIOR TO PICKUP	
		2		S1VM	OPEN S=I FOR BATTERY INSTALLATION.	
		3		SUVM	OPEN IU FOR BATTERY INSTALLATION.	
		4		S1VM	INSTALL S.I FLIGHT BATTERIES,	
		5		S1NF	CONNECT S-I FLIGHT BATTERIES.	
		6		SUVM	INSTALL IU FLIGHT BATTERIES.	
		7		SUVN	CONNECT IU FLIGHT BATTERIES.	
		8		RVFC	PERFORM FLIGHT CONTROL PREPARATIONS PER PROCEDURE 5-LLVI-319.	
		9			MAKE ALL PREPARATIONS NECESSARY FOR S=I AND IU POWER APPLICATION PRIOR TO T=545 MIN PICKUP TIME.	
		10		RGC¢	BEGIN FLIGHT COMPUTER SYSTEM PREPARATIONS PER PROCEDURE 5-LIUI-749.	
		11		SUVM	REMOVE REMAINDER OF S-IV/IU ACCESS KIT.	
	GR1	12	B1TM	S1TM SUTM	PREPARE S1 & IU TELEMETER SYSTEMS PER LV0=R=3001.	
	GR2	13	BRFS	SVRF	PREPARE S1 & IU RF SYSTEMS PER LV0-R-3002.	
545°						
	SRO	1	CNTS	ORNG	VERIFY COMPLEX IS ON CRITICAL POWER.	
	SIV	2	C4TC		VERIFY UMBILICAL TOWER WATER "ON".	
	YW3	3	CLTC	BSOC	VERIFY COMPLETION OF 5-LLVI-722.	
	YW3	4	CLTC	BSOC	START DISCRETE MONITORING.	
	SIV	5	C4TC		APPLY S-IV STAGE POWER PER PROCEDURE DAC-N-7001.	

PAGE TEST NO. VEHICLE

						SA ∞
TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
545 ⁰						
	01.0			5.50		
	BL2	6		R4FC	SET UP SHIV HYDRAULICS FOR OPERATION.	
	RD2	7	CATC	CPTE	PERFORM S.IV PROPULSION PRELIMINARY PREPARATIONS PER DAC-P-7009, PART I.	
	BL1	8	CSRP	S1NF	VERIFY SOI SAFETY SWITCHES ARE INSTALLED.	
	BL1	9	CSRP	SUVN	VERIFY IU SAFETY SWITCHES ARE INSTALLED	
	BL1	10	CSRP	SVOC S1NF	VERIFY DESTRUCT MONITORING SYSTEM READY	
	BL1	11	CSRP	CSCG	APPLY S=I GROUND POWER.	
	BL1	12	CSRP	CSCG	APPLY IU GROUND POWER.	
		13		C1NP	APPLY S-I STAGE POWER.	
		14		CUNP	APPLY IU STAGE POWER.	
		15		C1MP	GROUND MEASURING VOLTAGE "ON".	
		16		C1MP	VEHICLE MEASURING VOLTAGE "ON".	
	BL1	17	CSRP	C1MP	MEASURING RACKS "ON",	
	BL1	18	CSRP	CUMP	VEHICLE MEASURING VOLTAGE "ON".	
	BL1	19	CSRP	CUMP	MEASURING RACKS "ON".	
	BL1	20	CSRP	CUMP	GROUND MEASURING VOLTAGE "ON".	
	BL1	21	CSRP	CINV	S-I VEHICLE INVERTER #1 "ON",	
	BL1	22	CSRP	CINV	IU VEHICLE INVERTER #1 "ON".	
	BL1	23	CSRP	CINV	IU VEHICLE INVERTER NO. 2 "ON".	
	BL1	24	CSRP	CPSP	SERVICE STRUCTURE OAT POWER "ON".	
	BL1	25	CSRP	RBHM	SANBORN RECORDER TO 2 MM/SEC.	
	BL1	26	CSRP	SVOÇ	PULSE SENSOR POWER "ON",	
	BL1	27	CSRP		PULSE SENSOR TEST VOLTAGE "ON" AND	
	BL1	28	CSRP		VERIFY S=I AND S=IV DESTRUCT INDICATIONS "ON".	

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VEHICLE SA-7 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
- 545 ¹						
	BL1	29	CSRP	SVOC	RESET PULSE SENSORS.	
	BL1	30	CSRP	SVOC	VERIFY Sal AND S-IV DESTRUCT INDICATIONS "OFF".	
	BL1	31	CSRP	CINV	GROUND INVERTER NO. 1 "ON",	
	BL1	32	CSRP	CINV	GROUND INVERTER NO. 2 "ON",	
	BL1	33	CSRP	C10P	MODULE POWER SUPPLY "ON",	
	RD2	3 4	C4TC	CPTE	PERFORM PROPELLANT LINES VACUUM CHECKS PER DAC=P=7009, PART II.	
	RD2	35	C4TC	CPTE	PERFORM COMMON BULKHEAD MONITOR PREPARATIONS PER DAC-P-7009, PART III,	
	RD1	36	S4PE	SVMP	REMOVE CHILLDOWN DUCT DESSICATORS AND STUB FIN MAST ADAPTERS.	
		37		SVMP	PRECHARGE HYDRAULIC ACCUMULATORS TO 1600 PSIG AND LEAK CHECK SHRADER FITTING: POSITION 1 POSITION 2 POSITION 3 POSITION 4	
	YW4	3 8	CLTC	SSSC	OPEN SILO C=3 AND B=1.	
	SC	39	CSTC		INSTALL AND CONNECT SPACECRAFT INSTRUMENTATION BATTERIES PER PROCEDURE C=0007.	
∞ 540¹						
	YW3	1		RGCC	FLIGHT COMPUTER POWER "ON".	
	SC	5	CSTC		APPLY SPACECRAFT GSE POWER PER PROCEDURE C-0007.	
		3			BEGIN REMOVING S-I ENVIRONMENTAL PROTECTION.	
		4		SQFC	REMOVE PROTECTIVE COVER FROM Q-BALL.	
	RD3	5			PREPARE THE PNEUMATIC DISTRIBUTION SYSTEM FOR PRELAUNCH OPERATIONS PER PROCEDURE LVO=E=7015.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-540°						
	BL1	6		ANAG	TURN BREAKER 66A5A10-1 "ON" Q=BALL POWER BREAKER).	
	BL1	7		SUVN	VERIFY IU FLIGHT BATTERIES ARE INSTALLED AND ELECTRICALLY CONNECTED.	
	BL1	8		CUNP	VERIFY +8D10 AND +8D20 BATTERY INDICATIONS,	
	BL1	9		S1NF	VERIFY SEI FLIGHT BATTERIES ARE INSTALLED AND ELECTRICALLY CONNECTED.	
	BL1	10		C1NP	VERIFY *1D10 AND *1D20 BATTERY INDICATIONS.	
=535°	BL1	11	RS4C	ANAG	ST-124 HEATER SWITCH TO NORMAL POSITION	
	RD2	1		SMTE	VERIFY QUICK DISCONNECTS AND UMBILICAL CARRIERS PROPERLY INSTALLED (SWING ARM 2 AND GH2 VENT COUPLING.	
					NOTE	
					STEEL SHEAR PINS, SAFETY BLOCKS AND SAFETY WIRE INSTALLED, LANYARDS NOT CONNECTED, AND EJECT ACTUATOR PNEUMATIC LINES NOT CONNECTED.	
		2		C4TC	VERIFY Selv POWER "ON".	
	SIV	3		C4TC	ALL S-IV STATUS SELECT SWITCHES TO "ON SCHEDULE".	
	BR1	4			VERIFY CONNECTION OF SPGG TEMPERATURE MEASURING AND PROPER INDICATION.	
	SC	5		CSTC	VERIFY COMPLETION OF SPACECRAFT POWER APPLICATION.	
	SRO	6	CNTS		VERIFY FREQUENCY CLEARANCE FOR ALL SPACE VEHICLE RF SYSTEMS.	
	SRO	7	CNTS		VERIFY PATRICK 0-18 RADAR READY TO SUPPORT SPACECRAFT BEACONS AND THAT RADAR IS AWAY FROM THE PAD.	
	4 23-81 B (5/64					

PAGE TEST NO. VEHICLE

163 7=LSVI=300 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
- 535 ⁰						
	SRO	8	CNTS	ORNG	VERIFY 1,16 RADAR READY TO SUPPORT LAUNCH VEHICLE (IU) BEACON AND THAT RADAR IS AWAY FROM THE PAD AND WILL NOT INTERROGATE UNTIL REQUESTED.	
	SRO	9	CNTS	ORNG	STANDBY FOR READOUT OF SPACECRAFT TM.	
-5301						
	S	1	CLTC	RVFC	CONTROL COMPUTER "ON".	
	S	2	CLTC	RVFC	CHECK CONTROL COMPUTER NULLS.	
	S	3	CLTC	RVFC	CONTROL VOLTAGE "ON".	
	S	4	CLTC	RVFC	Q=BALL "ON" .	
	S	5	CLTC	RVFC	CONTROL ACCELEROMETER CHECK.	
	S	6	CLTC	RVFC	IU RATE GYRO CHECK,	
	S	7	CLTC	RVFC	S=I RATE GYRO CHECK.	
	s	8	CLTC	RVFC	ACCELEROMETER SWITCH CHECKS.	
	RD3	9		RECS	OPERATE THE ENVIRONMENTAL CONTROL SYSTEM PER PROCEDURE LVO-E-7018.	
		10		RECS	START ECS AIR TO THE IU.	
		11		SVMP	BEGIN REMOVAL OF STRIPPABLE COATING FROM FIN BASE (8 PLACES).	
	RD1	12		C1FP	VERIFY THAT POWER IS APPLIED TO S-I, THEN OPERATE PRE-VALVES 1 THRU 8 SWITCHES TO "OPEN".	
		13		SVMP	REMOVE CAPS FROM PRE-VALVE CONTROL VALVE VENT PORTS (8 PLACES).	
		14		S1VM RUAC	DISCONNECT HEISE PANEL CONNECTED TO FUEL VENT CONTROL LINE. RESTORE CONTROL LINE TO NORMAL AND VERIFY THAT FUEL VENTS 1 AND 2 "CLOSE".	
		15			RF AND TM CHECKS	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
530'						
	PA	15-1		CNTS	RF SILENCE "OFF".	
	S	15-2	CLTC	CUMP	IU C-BAND BEACON (RADAR) "ON".	
	SC	15-3	CSTC	CICM	SPACECRAFT TELEMETER A (LINK 1) "ON".	
	SC	15-4	CSTC	SICM	SPACECRAFT TELEMETER B (LINK 6) "ON".	
	SC	15-5	CSTC	SICM	SPACECRAFT TELEMETER C (LINK 14) "ON",	
	SC	15-6	CSTC	SICM	SPACECRAFT RADAR BEACON NO. 1 "ON".	
	SRO	15-7	CNTS	ORNG	PROCEED WITH TM READOUT AND REPORT WHEN AVAILABLE.	
	S	15-8	CLTC	C1MP	TELEMETERS F1, F2, F3, P2, S1, AND S2,	
	S	15-9	CLTC	CUMP	TELEMETERS F5, F6, S3, P1 AND MINITRACK "ON".	
	S	15-10	CLTC	C1MP	AUXILIARY EQUIPMENTS "ON".	
	S	15-11	CLTC	C1MP CUMP	MEAS. ON DDAS "QN".	
	S	15-12	CLTC	C1MP CUMP	SWEEP FREQUENCY CALIBRATOR "ON".	
	S	15-13	CLTC	C1MP CUMP C1PC	VERIFY TAPE RECORDER READY INDICATIONS.	
	S	15-14	CLTC	C1MP CUMP	TELEMETER CALIBRATIONS TO "PREFLIGHT".	
	S	15-15	CLTC	C1MP CUMP	PREFLIGHT CALIBRATIONS TO 50%.	
		15-16		CIPC	TELEMETERS D1, D2, AND D3 "ON".	
	S	15-17	CLTC		CALIBRATION SWITCH TO "MANUAL" (VERIFY VCO CALIBRATION LIGHTS "OFF"),	

PAGE VEHICLE

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TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-530 9						
	S	15-18	CLTC	B4TM	PERFORM TELEMETER CHANNEL READOUTS.	
					NOTE	
					VERIFY CHANNELS 21-24 OF TELEMETER D2, HIGH LEVEL PDM READING 100%.	
	S	15-19	CLTC	CUMP	RADAR ALTIMETER "ON".	
~525 °	S	16	CLTC	CUMP	GUIDANCE COMMAND RECEIVER "ON".	
723						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	ORNG	BEGIN INTERROGATION OF SPACECRAFT BEACON WITH 0.18 RADAR AND PROCEED WITH READOUT, REPORT WHEN COMPLETE.	
	S	1-2	CLTC	ORNG	INTERROGATE IU C-BAND BEACON WITH 1.16 RADAR AND REPORT READOUT WHEN COMPLETE.	
	S	1-3	CLTC	B1TM OTMD B4TM	TELEMETER RECORDINGS "ON".	
	S	1-4	CLTC	C1MP CUMP	TELEMETER CALIBRATIONS TO "AC COMMAND",	
	S	1~5	CLTC	C1MP CUMP	TM CALIBRATION COMMANDS "ON" AND "OFF". (DELAY 10 SEC BEFORE NEXT ITEM).	
	S	1=6	CLTC	C1MP CUMP	TELEMETER CALIBRATIONS TO "PREFLIGHT".	
	S	1=7	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS.	
	S	1=8	CLTC	CIPC	STEP THRU MANUAL CALIBRATION IN 2 SEC INCREMENTS.	
	S	1-9	CLTC	CIPC	BRIDGE CAL "HIGH" FOR 10 SECONDS.	
		1-10	CLTC	CIPC	BRIDGE CAL "LOW" FOR 10 SECONDS.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
~525°						
	S	2	CLTC	RPUP	PERFORM P.U. SLEW CHECKS.	
	s	3	CLTC	RBHM	SANBORN RECORDER TO 5 MM/SEC.	
	S	4	CLTC	RS4C	BRING ST=124 UP TO OPERATING CONDTIONS PER PROCEDURE LVO=G-7040.	
	S	5	CLTC	RATC	BRING AUTO THEODOLITE UP TO OPERATING CONDITIONS PER PROCEDURE LVO-G-7044.	
	BR1	6		SVME	VERIFY ALL SOUND INTENSITY MICROPHONE COVERS ARE REMOVED.	
	BRU	7		SVME	VERIFY ALL AMBIENT PRESSURE MEASUREMENTS ARE "OPEN" TO ATMOSPHERE.	S
	BL1	8		SGNP	CHECK ESE LIMIT SWITCHES PER PROCEDURE 1=LLVI=716.	
	RD3	9		RWCP	PREPARE THE PAD WATER SYSTEM FOR LAUNCH PER PROCEDURE LVO-E-7014, PART II,	
		10		ORNG	BRING WATER PRESSURE TO 175 PSI ON 36-INCH LINE WITH 1 PUMP.	
	RD1	11		S1VM C1FP	VENT LOX TANK STANDBY PRESSURE,	
		12		S1VM	AFTER LOX TANKS ARE VENTED, INSTALL CAP ON LOX TANK TEST SENSING LINE AT LOX TANK 4.	
		13		SQFC UQBL	START INSTALLATION OF Q-BALL RETRACTABLE	
		14		PLHF	SET UP LH2 SYSTEM PER LVO-L-1032.	
	SC	15		CSTC	VERIFY COMPLETION OF FLIGHT BATTERY INSTALLATION AND CONNECTION.	
	SC	16		CSTC	BEGIN SPACECRAFT POWER TRANSFER TEST PER PROCEDURE C=0007.	
		17		SVME	BEGIN MANUAL PREFLIGHT PHOTOGRAPHIC INSTRUMENTATION SYSTEM PURGE.	
	YW3	18	CLTC	RGCC	REQUEST APS=102 (SHORT COMMAND SYSTEM TEST);	

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167 TEST NO. 7 - LSV I - 300
VEHICLE SA - 7 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
≖525°						
~522°	SRO	19	CNTS	ORNG	BEGIN GUIDANCE COMMAND SHORT TEST). TRANSMISSION.	
726						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1MP CUMP	PREFLIGHT CALIBRATIONS TO 0%.	
	S	1-2	CLTC	C1CM	TV TO "FILAMENT".	
	S	1-3	CLTC	CUMP	AZUSA "ON",	9
	S	1-4	CLTC	CUMP	MISTRAM "ON".	
	S	1-5	CLTC	CUMP	ODOP "ON",	
	S	1-6	CLTC	врор	ODOP GROUND TRANSMITTER NO. 1 "ON".	
∍520° 30″						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1MP CUMP	PREFLIGHT CALIBRATION TO 100%.	
-520°						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1CM	TV TO "B+",	
	S	1-2	CLTC	C1MP	VERIFY EBW MEAS. "OFF".	
	S	1-3	CLTC	CLVN	DESTRUCT ENABLE "ON".	
	S	1-4	CLTC	C4DP	VERIFY SAFETY BUS INDICATION "ON".	
	S	1-5	CLTC	R4VC	VISICORDER "ON".	
	S	1 = 6	CLTC	C10P	PERFORM MODULE CHECK.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-520°						
	S	1-7	CLTC	C1DP	S=I CDR #1 AND #2 "ON".	
	S	1-8	CLTC	C10P	CHECK S⇔I EBW #1 AND #2 VOLTAGES.	
	S	1-9	CLTC	C1DP	S=I CDR #1 AND #2 TO "INTERNAL" .	
	SRO	1-10	CNTS	ORNG	INTERROGATE MISTRAM AND REPORT READ OUT	<u> </u>
	SRO	1-11	CNTS	ORNG	INTERROGATE AZUSA AND REPORT READ OUT.	
	RD1	2	RVFC	SVMP	VERIFY HYDRAULIC ACCUMULATOR PRECHARGE COMPLETE.	
	BL1	3	RVFC	ANAG	CLOSE S-I HYDRAULIC PUMP CIRCUIT BREAKERS.	
	RD1	4		C1FP	AFTER VERIFICATION THAT PRE-VALVE CONTROL VALVES HAVE BEEN RESTORED TO "NORMAL", CYCLE EACH PRE-VALVE FROM "OPEN" TO "CLOSE" TO "OPEN".	
	RD1	5		C1FP	AFTER VERIFICATION THAT FUEL VENT CONTROL LINE IS RETURNED TO "NORMAL", OPERATE FUEL VENT SWITCH TO "OPEN",	
		6		S1VM	LEAK CHECK FUEL VENT CONTROL LINE.	
		7		S1VM	INSTALL CAP ON FUEL TANK TEST SENSING LINE AT LOX TANK 4 AFTER FUEL VENTS ARE "OPEN".	
-516° 30"						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	B1TM OTMD	8 KC OSCILLATORS "ON".	
	S	1-2	CLTC		CALIBRATION SWITCH TO "AUTOMATIC" AND GIVE CAL. START.	
	S	1-3	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS.	
	S	1-4	CLTC	C1MP CUMP CIPC	TAPE RECORDERS TO "RECORD".	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-516° 30"						
	S	1-5	CLTC	C1MP CUMP	PREFLIGHT CALIBRATIONS TO 0%, 25%, 50%, 75%, AND 100% IN 2 SEC INCREMENTS.	
	S	1-6	CLTC	C1MP CUMP	RECORDER TRANSFERS "ON".	
	S	1~7	CLTC	C1MP CUMP	TELEMETER CALIBRATIONS TO "INFLIGHT".	
	S	1-8	CLTC	CUMP	P1 TELEMETER CALIBRATION COMMAND.	
	S	1-9	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS "ON" FOR 5 SECONDS.	
	S	1-10	CLTC	CIPC	GIVE TELEMETER CALIBRATION COMMAND.	
	S	1-11	CLTC	RBHM	S=I AND IU HIGH CALIBRATE COMMAND (15 SECONDS).	
	S	1-12	CLTC	RBHM	S-I AND IU LOW CALIBRATE COMMAND (15 SECONDS).	
	S	1-13	CLTC	RBHM	S=I AND IU RUN COMMAND.	
	S	1-14	CLTC	C1MP CUMP	RECORDER TRANSFERS "OFF".	
-515°						
		1	CLTC	RGCC	INITIATE APS 103 (CHI-ALIGN),	
		2			RF AND TM CHECKS.	
	S	2~1	CLTC		TELEMETER CALIBRATION COMMANDS "ON" FOR 5 SEC.	
	S	2-2	CLTC	CUMP	P1 TELEMETER CALIBRATION COMMAND.	
	S	2=3	CLTC		TAPE RECORDERS "OFF" . (DELAY 10 SEC BEFORE NEXT STEP)	
	S	2=4	CLTC	C1MP CUMP		

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TEST NO. 7-LSVI-300
VEHICLE SA-7 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
~515°						
				CIPC	TAPE RECORDERS TO "PLAYBACK".	
	S	2=5	CLTC	C1MP CUMP CIPC	CONFIRM TAPE RECORDERS READY INDICATIONS (BEFORE PROCEEDING).	
	S	2-6	CLTC	B1TM OTMD	8 KC OSCILLATORS "OFF".	
	S	2-7	CLTC	B1TM OTMD B4TM	TELEMETER RECORDINGS "OFF".	
	RD2	3	C4TC	CPTE	VERIFY COMPLETION OF S-IV PROPULSION PRELIMINARY PREPS PER DAC-P-7009, PART I.	
	RD2	4	C4TC	CPTE	VERIFY COMPLETION OF BULKHEAD VACUUM MONITOR PREPS PER DAC-P-7009, PART III,	
	RD2	5	C4TC	CPTE	VERIFY COMPLETION OF PROPELLANT LINES VACUUM CHECK PER DAC-P-1002, PART II.	
	RD2	6	C4TC	CPTE	PERFORM LO2 AND LH2 SYSTEM PURGES AND COMPONENTS CHECK PER DAC-P=7009, PART IN	Y
		7		S1VM	VERIFY LOX VENT AND RELIEF VALVES "OPEN" AND THEN INSTALL LOX VENT LINES.	
	SRO	8	CNTS	ORNG	COMMAND CARRIER "ON",	
	S	9	CLTC	CUNP	HORIZON SENSOR POWER "ON".	
	S	10	CLTC	CUNP	HORIZON SENSOR "ON".	
	S	11	CLTC	RFSR	SEQUENCE RECORDERS TO 10 MM/SEC.	
	S	12	CLTC	RSEQ	SEQUENCE AND E+1 RECORDERS TO MINUTE SPEED AND TIME PULSE ON.	
,	S	13	CLTC	RSEQ	VERIFY VOLTAGE RECORDERS TO FINE RANGE.	
	S	14	CLTC	,	SEQUENCE RECORDERS TO MINUTE SPEED AND TIME PULSE ON.	
	S	15	CLTC	CDCG	CHECK VEHICLE GROUND POWER GENERATOR SETTINGS FOR POWER TRANSFER.	
	S	16	CLTC	B4TM	TELEMETER RECORDING ON.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-515 °						
	S	17	CLTC	C4DP	TEST S=IV EBW S FOR READY	
	S	18	CLTC	C4DP	VERIFY S&A "SAFE"	
	S	19	CLTC	CSRP	VERIFY S-I AND IU FLIGHT SEQUENCER ZERO INDICATIONS.	
	S	20	CLTC	C4DP	S=IV CDR NO. 1 AND NO. 2 "ON".	
	S	21	CLTC	C4DP	VERIFY S-IV EBW NO. 1 AND NO. 2 "OFF" AND "NOT READY".	
	S	22	CLTC	CSPC	S=IV EBW EXTERNAL "ON".	
	S	23	CLTC	C4DP	S=IV CDR NO. 1 AND NO. 2 TO "INTERNAL".	
	S	24	CLTC	C40P	VERIFY ALL S-IV NOT READY INDICATIONS	
	S	25	CLTC	CUNC	COOLING TO PREFLIGHT.	
	S	26	CLTC	SUVN	VERIFY ST-124 BLOWER "ON".	
	S	27	CLTC	CUNC	COOLING TO "INFLIGHT".	
	S	28	CLTC	SUVN	VERIFY PRIMARY BLOWER ON (AFTER 40 SECONDS).	
	S	29	CLTC	RBHM	SANBORN RECORDER TO 25 MM/SEC.	
	SRO	30	CNTS	ORNG	CUTOFF COMMAND AND RELEASE AT TEST SUPERVISOR REQUEST.	
	S	31	CLTC	C1DP C4DP	VERIFY S=I AND S=IV EBW #1 AND #2 ON "INTERNAL" AND CHARGED.	
	s	32	CLTC		S=IV EBW #1 AND #2 TO EXTERNAL AND VERIFY MOFFM.	
	S	33	CLTC		S-I EBW #1 AND #2 TO EXTERNAL AND VERIFY "OFF".	
510 °						
	sc	1			VERIFY COMPLETION OF POWER TRANSFER TEST.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-510°						
	S	2	CNTS	ORNG	VERIFY 0=18 RADAR AWAY FROM THE PAD (AFTER BEACON NO. 1 READOUT).	
	S	3	CSTC	CICM	SPACECRAFT RADAR BEACON NO. 1 "OFF".	
	S	4	CSTC	CICM	SPACECRAFT RADAR BEACON NO, 2 "ON".	
	S	5	CNTS	ORNG	ONE (1) MINUTE AFTER NOTIFICATION OF BEACON ON, BEGIN INTERROGATION AND PROCEED WITH READOUT, REPORT READOUT WHEN COMPLETE.	
	RD4	6		PRPF	SET UP RP#1 SYSTEM PER LVO#L#1031.	
	SRO	7	CNTS	ORNG	MONITOR ALL RF SYSTEMS AND REPORT ANY CHANGE IN INTERNAL POWER.	
	S	8	CLTC	B1TM OTMD	TELEMETER RECORDINGS "ON".	
	S	9	CLTC	CUNP	VERIFY IU READY FOR POWER TRANSFER ON,	
		10	CLTC	RVFC	HYDRAULIC PUMPS ENABLE "ON" .	
	S	11	CLTC	RVFC	S=I HYDRAULIC PUMPS ON.	
	S	12	CLTC	RGCC	INITIATE SEI STEERING COMMANDS (APSE104	<u> </u>
	S	13	CLTC		PERSONNEL CONNECTED WITH S-IV POWER TRANSFER SWITCH TO S-IV CIRCUIT.	
	SIV	14	C4TC	CIPC	INSTRUMENTATION SYSTEM TO "INTERNAL" .	
	SIV	14-1	C4TC	CSPC	STAGE POWER SYSTEMS TO "INTERNAL" .	
		14-2	C4TC	CSPC CIPC	READOUT BATTERIES.	
	SIV	14-3	C4TC	CIPC	PERFORM AUTOMATIC VCO CALIBRATION.	
	SIV	14-4	C4TC	CSPC CIPC B4TM	VERIFY SATISFACTORY ON INTERNAL POWER.	
	SIV	14-5	C4TC	CSPC	STAGE POWER SYSTEMS "EXTERNAL".	

PAGE TEST NO. VEHICLE

REVISION					VEHICLE	SA-
TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
~510°						
	SIV	14-6	C4TC	CIPC	INSTRUMENTATION SYSTEMS "EXTERNAL".	
	SIV	14-7	C4TC	C4DP	S=IV CDR S #1 AND #2 "EXTERNAL" AND "OFF".	
	SIV	14-8	C4TC	C4DP	Selv EBW°S #1 AND #2 "EXTERNAL" AND VERIFY "OFF",	
	SIV	14-9		C4TC	S=IV POWER TRANSFER PERSONNEL SWITCH TO S=CIRCUIT,	
	S	14-10		C4TC	VERIFY SEIV POWER TRANSFER SATISFACTORY	9
	S	15	CLTC	C1CM	CAMERA LIGHTS "ON" (HOLD ON).	
,	S	16	CLTC	C1NP	S=I STAGE POWER TEST #1 "ON",	
	S	17	CLTC	CSRP	NOTE SEI POWER TRANSFER OK,	
	S	18	CLTC	C1CM	CAMERA LIGHTS "OFF".	
	S	19	CLTC	CUNP	IU STAGE POWER TEST #1 "ON" .	
	S	20	CLTC	CSRP	NOTE IU POWER TRANSFER OK.	
	S	21		CSRP	VERIFY SATISFACTORY S=I AND IU POWER TRANSFER.	
	S	22		C1NP	Sel Stage Power Test #1 "Off",	
	S	23		CUNP	IU STAGE POWER TEST #1 "OFF".	
	S	24	CLTC	_	S-I CDR #1 AND #2 TO "EXTERNAL" AND "OFF".	
	S	25	CLTC	CSPC	Salv EBW EXTERNAL "OFF".	
	S	26	CLTC	RBHM	SANBORN RECORDER TO 2 MM/SEC.	
	S	27	CLTC	R4VC	VISICORDER TO "OFF".	
	S	28	CLTC	RVFC	S=I HYDRAULIC PUMPS "OFF".	
		29	CLTC	RVFC	HYDRAULIC PUMPS ENABLE "OFF",	
	2	29	CLTC	RALC	HYDRAULIC PUMPS ENABLE "OFF",	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
510°						
	S	30	CLTC	CUNP	HORIZON SENSOR MOFF".	
	S	31	CLTC	CUMP	AZUSA "OFF".	
	S	32	CLTC	ORNG	COMMAND CARRIER "OFF".	
	S	33	CLTC	CUMP	C-BAND BEACON (RADAR) "OFF".	
	S	34	CLTC	CUMP	ALTIMETER "OFF" .	
	S	35	CLTC	CUMP	MISTRAM "OFF",	
	S	36	CLTC	CUMP	GUIDANCE COMMAND RECEIVER "OFF".	
	S	37	CLTC	CUMP	ODOP "OFF",	
	S	38	CLTC	C1CM	TV CAMERA "OFF".	
		39		CLTC	PERSONNEL CONNECTED WITH S-IV CONTROL CHECKS SWITCH TO YW2 CIRCUIT.	
	YW2	39=1		C4TC	CLEAR S-IV ENGINE AREA OF UNNECESSARY PERSONNEL AND EQUIPMENT.	
	YW2	39-2		RVFC	STAGE SELECTOR SWITCH TO "S-IV".	
	YW2	40		RVFC	HYDRAULIC PUMPS ENABLE "ON".	
	YWZ	40-1	RVFC	R4FC	SHIV HYDRAULIC PUMPS "ON".	
	YW2	40-2	RVFC	RGCC	INITIATE S=IV STEERING COMMANDS (APS-105)	
	YWZ	40-3		RVFC	IU RATE GYRO TORQUER COMMANDS.	
	YW2	40-4	RVFC	R4FC	"CLOSE" S-IV HYDRAULIC ACCUMULATOR VALVES.	
	YW2	40-5	RVFC	R4FC	VERIFY SEIV HYDRAULIC PUMPS "OFF",	
	YW2	40-6		RVFC	FLIGHT CONTROL SWITCH TO "S-I".	
	YW2	40-7		RVFC	CONTROL VOLTAGE "OFF".	
	YW2	40-8		RVFC	CONTROL COMPUTER "OFF".	
	YW2	40-9		RVFC	IU RATE GYROS "OFF".	
	YW2	40-10		RVFC	CONTROL ACCELEROMETERS "OFF".	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-510°						
	YW2	40-11		RVFC	S-I RATE GYROS "OFF".	
	YW2	40-12		RVFC	Q=BALL SYSTEM "OFF".	
	YW2	40-13		RVFC	Q=BALL HEATERS #OFF#;	
	YW2	40-14		RVFC	ACCELERATION SWITCH "OFF",	
	GY1	40-15	RGCC	B1TM OTMD	TELEMETER RECORDING "OFF".	
	S	41	CLTC	CUNP	HORIZON SENSOR POWER "OFF".	
	S	42	CLTC	C1MP	TELEMETERS F1, F2, F3, P2, S1 AND S2	
	S	43	CLTC	C1MP CUMP	TELEMETER AUXILIARY EQUIPMENT "OFF".	
		44	CLTC	C1MP CUMP	SWEEP FREQUENCY CALIBRATOR "OFF".	
		45	CLTC	CUMP	TELEMETERS F5, F6, P1, S3 AND MINITRACK "OFF",	
	S	46	CLTC	CLVN	DESTRUCT ENABLE "OFF".	
	S	47	CLTC	RSEQ	SEQUENCE AND E+1 RECORDERS TO "HOUR SPEED" AND TIME PULSE "OFF".	
	S	48	CLTC		SEQUENCE RECORDERS TO "HOUR SPEED" AND TIME PULSE "OFF".	
	S	49		C1MP CUMP	TELEMETER CALIBRATION TO PREFLIGHT.	
	S	50	CLTC		VERIFY VCO CALIBRATION "OFF" AND "AUTO".	
	S	51	CLTC	RFSR	SEQUENCE RECORDERS TO 2 MM/SEC.	
	S	52	CLTC	CUNC	COOLING TO PREFLIGHT.	
	S	53	CNTS		REPORT ANY CHANGE OR DEVIATIONS OF REEQUIPMENTS DURING INTERNAL RUN.	

PAGE

176 TEST NO. 7=LSVI=300
VEHICLE SA=7 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
~505°						
	YW2	1.		RVFC	S-IV CONTROL CHECKS PERSONNEL SWITCH TO S-CIRCUIT.	
	S	2	CLTC	RVFC	VERIFY COMPLETION OF S-IV CONTROL CHECKS.	
		3		CIPC	TELEMETER D1, D2 AND D3 OFF AFTER COMPLETION OF S-IV CONTROL CHECKS.	
		4		B4TM	TELEMETER RECORDING OFF AFTER S-IV TELEMETERS MOFFM.	
		5		C4TC	OPEN SHIV ENGINE AREA FOR NORMAL ACCESS	
	YW1	6		RS4C	SECURE ST=124 PER PROCEDURE LVO-G=7040 AFTER CLEARANCE FROM RGCC AND RVFC.	
	YW1	7		RATC	SECURE AUTO THEODOLITE AFTER CLEARANCE FROM RS4C PER PROCEDURE LVO-G-7044.	
		8		CUNC	COOLING "OFF",	
		9		BDOP	ODOP GROUND TRANSMITTER NO. 1 "OFF",	
7500 0						
		1		RGCC	FLIGHT COMPUTER SYSTEM POWER "OFF".	
		2		SVMP	INSTALL DRAIN SCREW ACCESS COVERS ON OUTBOARD ENGINES.	
		3		SVMP	RECORD GEAR CASE PRESSURE FOR EACH ENGINE; POSITION 1 POSITION 3 POSITION 4 POSITION 5 POSITION 6 POSITION 7 POSITION 8	
		4		SVMP	REMOVE GEAR CASE MONITOR GAGES AND INSTALL AN 929-4C CAP ON PORT OF CROSS.	
		5		S1TM SUTM S4ME	CONNECT TELEMETER SYSTEMS TO VEHICLE TELEMETER ANTENNAS.	

PAGE

177 TEST NO. 7~LSVI-500 VEHICLE SA-7 SA-7

TIME	COMM.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
~495°						
	RD4	1.		PLOF	SET UP LOX SYSTEM PER PROCEDURE LV0=L=1002.	
-490°						
	SRO	1	CNTS	ORNG	VERIFY 0-18 RADAR AWAY FROM THE PAD (AFTER BEACON #2 READOUT).	
	SC	2	CSTC	CICM	SPACECRAFT RADAR BEACON #1 "ON".	
	SRO	3	CNTS	ORNG	ONE (1) MINUTE AFTER NOTIFICATION OF BEACON "ON", BEGIN INTERROGATION OF BOTI BEACONS AND PROCEED WITH READOUT. REPORT READOUT WHEN COMPLETE.	4
	PA	4		CNTS	ANNOUNCE & COUNTDOWN READOUTS WILL BE INTERRUPTED.	
	,	5		CGCM	PERFORM LIFTOFF CHECKS WITH BDOP, ANEMOMETER RECORDERS, HANGAR D AND COUNTDOWN CLOCK,	
	BL1	6		CGCM	PERFORM RCA SEQUENCER CHECKS.	
-485°						
	RD3	1		SGMP	PREPARE THE HOLDDOWN ARMS FOR LAUNCH PER PROCEDURE LVO=E=7016.	
	RD1	2	C1TC	S1VM	SECURE INSTRUMENT CONTAINER #2.	
	RD1	3	CLTC	SUVM	SECURE IU TUBE 4 FOR FLIGHT,	
	S	4	CLTC		VERIFY TELEMETER SYSTEMS CONNECTED TO VEHICLE TELEMETER ANTENNAS.	
-480°						
		1		ANAG	CLOSE ENGINE HEATER CIRCUIT BREAKERS.	
	RD3	2		RECS	START ECS TO S-IV ENGINE COMPARTMENT.	
	S	3	CLTC	C2MP	TELEMETER S1 AND S2 ON.	

PAGE TEST NO. VEHICLE

TIME	сомм.	SEQUENCE	COMMAND	RESPONSE	DESCRIPTION	SA = 7
	CH.		STA.	STA.		
=480°						
	S	4	CLTC	CUMP	TELEMETER S3 ON.	
	S	5	CLTC	C1PC	TELEMETER D2 ON.	
-476°						
	S	1	CLTC	B1TM B4TM	VERIFY RECEPTION OF TELEMETER SIGNALS S1, S2, S3 AND D2,	
	S	2	CLTC	C1MP	TELEMETER S1 AND S2 OFF.	
	S	3	CLTC	CUMP	TELEMETER S3 OFF.	
	S	4	CLTC	C1PC	TELEMETER D2 OFF.	
					NOTE: IU TUBE DOOR #3 CAN BE SECURED FOR FLIGHT AFTER COMPLETION OF STEPS AT T=476?	
					NOTE: S-I CANISTER 13 CAN BE SECURED FOR FLIGHT AFTER COMPLETION OF STEPS AT T=476?	
-475°						
	RD1	1			REQUEST S=I COMPONENT HEATERS ON AND PERFORM HEATER CHECKS PER PROCEDURE LVO=MV=P+1003;	
		2			CHECK SETTING OF CONTROL REG (750 PSIG) AND THEN SECURE FOR LAUNCH.	
		3			INSTALL AN 929-4C CAP ON 750 PRESSURE MEASUREMENT CALIBRATION TEE AND SAFETY WIRE.	
		4			OPEN 750 OK PRESSURE SWITCH CALIBRATION HAND VALVE AND SAFETY WIRE "OPEN".	
	BR1	5	C1TC		VERIFY CONTROL SUPPLY AND REGULATOR MEASUREMENTS HAND VALVE OPEN AND SAFETY WIRED (FUEL BAY 3).	
		6		C1CM	BEGIN LOX TANK CAMERAS LENS PURGE.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-475°						
	8R1	7	C1TC	SSMS	VERIFY MEASURING READY TO SECURE FUEL BAYS.	
	BL1	8	C1TC	S1NA	VERIFY NETWORKS READY TO SECURE FUEL BAYS.	
-470 0						
		1	C1TC	SVMP	INSTALL FUEL BAY DOORS 1, 2 AND 4 (REMOVE INTERNAL ACCESS LADDERS).	
	SRO	2	CNTS	ORNG	VERIFY RADAR AWAY FROM PAD.	
	SC	3	CSTC	CICM	SPACECRAFT RADAR BEACON NO. 1 "OFF".	
	SC	4	CSTC	CICM	SPACECRAFT RADAR BEACON NO. 2 "OFF".	
	SC	5	CSTC	CICM	SPACECRAFT TELEMETER A (LINK 1) "OFF".	
	SC	6	CSTC	CICM	SPACECRAFT TELEMETER B (LINK 6) "OFF".	
	SC	7	CSTC	CICM	SPACECRAFT TELEMETER C (LINK 14) "OFF".	
	PA	8		CNTS	RF SILENCE "ON".	
-465°						
	BR1	1	C1TC	RFDM	VERIFY FIRE DETECTION CALIBRATION PER PROCEDURE 2-LSII-707	
	BR1	2	C1TC	RÇSM	VERIFY CSM CALIBRATION PER PROCEDURE 2-LSII-706,	
		3	C1TC	SVMP	VERIFY REMOVAL OF STRIPPABLE COATING FROM FIN BASE.	
	SC	4	CSTC	CPWR	VERIFY POWER IS REMOVED FROM THE SPACECRAFT.	
	BL1	5	C1TC	S1NF	BEGIN S-I RETROROCKET INITIATOR CONNECTION PER PROCEDURE 1-LSII-707.	
	BL1	6	CLTC	SUVN	BEGIN CONNECTION OF HORIZON SENSOR DOME THRUSTER SQUIB PER PROCEDURE 1-LIUI-502	1
		7		SGMP	SECURE DOOR ON ENGINE SERVICE PANEL AT NW PEDESTAL LEG OF LAUNCHER.	

PAGE TEST NO. VEHICLE

180 7=LSVI=300 SA=7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
465 9						
		8		CSTC	VERIFY SPACECRAFT BATTERIES DISCONNECTE)
	BL1	9	CNTS	ANAG	TURN BREAKER 66A5A10-1 MOFFM (G-BALL POWER BREAKER)	
455°					TOPPALL FOREN BREAKEN).	
	RD2	1	C4TC	CPTE	VERIFY COMPLETION OF S-IV LOX AND LH2 SYSTEM PURGES AND COMPONENT CHECKS PER DAC-P-7009, PART IV.	
	RD2	2	C4TC	CPTE	PERFORM AUTOMATIC VALVE CONTROL FUNCTIONAL CHECK PER DAC-P=7009, PART V	
	S	3	CSTC	CNTS	VERIFY SPACE VEHICLE AND LAUNCH COMPLEX RF SILENCE AND Q=BALL HEATER BREAKER "OFF".	
	SC	4	CSTC	CPYE	CONNECT SPACECRAFT LES IGNITER PER PROCEDURE C-10001.	
450 9						
					THE THE HERITAGE STATES	
		1		SUVM	INSTALL HORIZON SENSOR COVER.	
435 0						
	BL1	1	C1TC	S1NF	VERIFY COMPLETION OF RETROPOCKET	
				2	INITIATOR CONNECTION.	
430 0				, ,		
	RD1	1	C1TC	SVMP	VERIFY COMPLETION OF HYPERGOL INSTALLATION.	
	RD3	2	SVMP	RECS	BEGIN ECS AIR TO S-I ENGINE COMPARTMENT	
	BR1	3	C1TC	SVME	END MANUAL PREFLIGHT PHOTOGRAPHIC INSTRUMENTATION.	
	BR1	4	C1TC		REMOVE PREFLIGHT MOD A EJECT TUBE COVERS.	

EMERGENCY PROCEDURE (T-410' to T-0')

(AFTER S-I LOX TANKING)

PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-425°						
		1	PPSO	ORNG	BRING UP 4 PUMPS TOTAL ON 36-INCH WATER SYSTEM (175 PSI).	
		2		PPSQ	ESTABLISH COMPLEX ROADBLOCKS (ACCESS TO PAD AFTER THIS TIME BY ACCESS LIST).	
		3		C1FP	VERIFY FUEL LOADING PANEL FUEL BUBBLING ON THEN OPERATE FUEL BUBBLING SWITCH ON	
	RD2	4	C4TC	CPTE	VERIFY COMPLETION OF AUTOMATIC VALVE CONTROL FUNCTIONAL CHECK, DAC-P-7009, PART V, INCLUDING SECURING OPERATIONS.	
	SC	5	CSTC	CPYR	VERIFY COMPLETION OF SPACECRAFT LES	
	PA	6		CNTS	ALL PERSONNEL CLEAR PAD AREA EXCEPT THOSE PERSONNEL SPECIFICALLY LISTED ON ACCESS. PERSONNEL ON ACCESS LIST REPORT TO STANDBY POSITIONS.	
-420 9						
		1.		RWCP	ARM WATER SYSTEM.	
	BL1	2	CNTS	ANAG	TURN BREAKER 66A5A10-1 "ON" (Q-BALL POWER BREAKER).	
	SÇ	3	CNTS	CSTC	VERIFY SPACECRAFT PERSONNEL ARE CLEARING THE PAD.	
	SÇ	4	CSTC	CPWR	APPLY SPACECRAFT POWER PER PROCEDURE C-0007.	
	YW3	5	CLTC	BSOC	VERIFY CONSTANTS AND SCAN RATE OF ONE MINUTE INSERTED INTO PROPELLANT SCAN PROGRAM.	
-410°						
	RD4	1		PPSO	VERIFY AREA CLEAR FOR S-I LOX LOADING.	
	YW3	2	CLTC	BSOC	INITI/TE PROPELLANT LOADING SCAN PROGRAM.	
	RD4	3	CLTC	RLOO	LOAD S=1 LOX TO 18 PERCENT PER PROCEDURE LVO=L=1004.	

EMERGENCY PROCEDURE (T-410' to T-0')

(AFTER S-I LOX TANKING)

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PAGE
TEST NO.
VEHICLE

7 - LSVI - 300
SA - 7 SA = 7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
×410 °						
					NOTE	
					IN EVENT OF ACCIDENTAL LOX PREVALVE CLOSURE, REFER TO EMERGENCY PROCEDURE,	
-395 ⁰						
		1		S1VM	S=I LOX LOADING CREW RETURN TO SERVICE STRUCTURE AND PEDESTAL (AFTER TANK PRESSURE IS BELOW 2 PSI).	
	PA	2		CNTS	RF SILENCE MOFFM.	
≈390°						
		1		SVMP	BEGIN S-I LOX LEAK CHECKS PER PROCEDURE LVO=MV-P=1007.	
	PA	2		CNTS	AFTER VERIFICATION OF NO GROSS LEAKS ON S-I, ANNOUNCE CLEARANCE FOR OPERATIONAL PERSONNEL TO RETURN TO THE PAD AND VEHICLE (BY ACCESS LIST).	
-385 ⁰						
	BR1	1	C1TC	SSMS	VERIFY MEASURING READY TO SECURE LOX. BAYS.	
	BL1	2	C1TC	S1NA	VERIFY NETWORKS READY TO SECURE LOX BAYS.	
-3801						
	RD4	1.	C1TC	RLOO	VERIFY S=I LOX LOADED TO 18 PERCENT.	
		2		SVMP	AFTER LOX LEAK CHECK IS COMPLETE REMOVE INTERNAL ACCESS LADDERS. SECURE LOX BAYS: #1, 2, 3 AND 4 DOOR.	
		3		SVMP	AFTER LEAK CHECKING (LEAK TEC) LOX DIFFERENTIAL STEP PRESSURE SWITCH SENSING LINE, REMOVE INTERNAL ACCESS LADDER, SECURE FUEL BAY 3 DOOR,	
	RD3	4			BEGIN PREPARATIONS OF THE UMBILICAL SWING ARMS FOR LAUNCH PER PROCEDURE	

EMERGENCY PROCEDURE (T-410' to T-0')

(AFTER S-I LOX TANKING)

PREMATURE CLOSURE OF S-I PREVALVES

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- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-380°						
					LV0-E-7016.	
	RD3	5		UQBL	BEGIN PREPARATIONS OF Q-BALL COVER RETRACT SYSTEM PER PROCEDURE LVO-E=7019	
- 375 ⁰						
		1		SVMP	INSTALL CENTER BARREL COVER.	
		2		ANAG	INSTALL JUMPER FOR CDR LIFTOFF (84A1J17/LITTLE J TO 84A1J20/BIG F).	
		3		SVOR		
				,	SEQUENCE RECORDERS TO MINUTE SPEED AND TIME PULSE "ON";	
	PA	4		CNTS	ANNOUNCEMENT = ALL PERSONNEL CONNECTED WITH COMMAND CHECKS, MAN APPOINTED STATIONS (NORCOM S CIRCUIT).	
	S	5		C1MP	TELEMETER F2 "ON",	
	S	6		C1MP	AUXILIARY EQUIPMENT "ON".	
	s	7		C1MP	TM CALIBRATION TO INFLIGHT.	
-370 9						
	BL2	1	C4TC	SMTE	PERFORM S-IV AFT ACCESS KIT REMOVAL PER DAC-L-7002.	
		2			COMMAND CHECKS,	
	S	2-1	CNTS	S40E	VERIFY MODEL 243 PULSE SENSORS CONNECTED TO 410A14P2 AND 410A19P2.	
	S	2-2	CNTS	S1NF S40E	VERIFY INERT DESTRUCT S&A UNITS ARE ELECTRICALLY CONNECTED.	
	S	2-3		CNTS	VERIFY PSO ON HAND TO OBSERVE DESTRUCT CHECKS ON VEHICLE AND IN BLOCKHOUSE.	
	S	2-4	CNTS	CSPC	VERIFY 400 CYCLE POWER "ON",	
	S	2~5	CNTS	C4DP	VERIFY MODULE POWER SUPPLY "ON" AND CHECK FOR 5 PLUS OR MINUS 0,2 VOLTS.	

EMERGENCY PROCEDURE (T-410' to T-0')

(AFTER S-I LOX TANKING)

PREMATURE CLOSURE OF S-I PREVALVES

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PAGE TEST NO. VEHICLE

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COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
S	2-6	CNTS	C10P	VERIFY MODULE POWER SUPPLY "ON".	
S	2-7	CNTS	C4DP	TEST S=IV EBW(S FOR READY.	
S	2-8	CNTS	C4DP	VERIFY STIV SEA SAFE.	
S	2 == 9	CNTS	SVOC	VERIFY PULSE SENSOR POWER "ON",	
S	2-10	CNTS	CSSP	VERIFY MSFC TALKBACK ENABLE "ON".	
S	2-11	CNTS	CLVN	DESTRUCT ENABLE "ON".	
S	2-12	CNTS	C4DP	VERIFY SAFETY BUS INDICATION "ON".	
S	2-13	CNTS	CIPC	TELEMETER D2 "ON",	
S	2-14	CNTS	B4TM	VERIFY RASTER.	
S	2-15	CNTS	B4TM OTMD	TELEMETER RECORDERS "ON".	
S	2-16	CNTS	R4VC	VERIFY VISICORDER "ON",	
S	2-17	CNTS	RFSR	SEQUENCE RECORDERS TO 10 MM/SEC.	
S	2-18	CNTS		FUNCTION SELECTOR TO FIRING UNIT VOLTAGE CHECK.	
S	2-19	CNTS	ORNG	COMMAND TRANSMITTER NO. 1 "ON".	
S	2-20	CNTS		VERIFY PANEL SELECT TO AUXILIARY RECORDER,	
S	2-21	CNTS	CiDP	S=I CDR OS #1 AND #2 PONP,	
S	2-22	CNTS	RHBM	SANBORN RECORDER TO 5 MM/SEC.	
S	2-23	CNTS	C1DP	Sel CDROS #1 AND #2 TO "INTERNAL".	
S	2-24	CNTS	C4DP	S=IV CDR(S #1 AND #2 "ON",	
S	2~25	CNTS		VERIFY SmIV EBW'S #1 AND #2 "OFF" AND "NOT READY".	
S	2-26	CNTS	C4DP	S-IV CDR'S #1 AND #2 TO "INTERNAL".	
s	2-27	CNTS	RBHM	SANBORN RECORDER TO 25 MM/SEC.	
S	2~28	CNTS	ORNG	CUTOFF COMMAND AND RELEASE UPON REQUEST	
	SSSSSSSSSSSSSSSSSSSSSSSSSSSSSSSSSSSSSS	SEQUENCE Sequen	SHOPPICE STA. SEQUENCE STA. SEQUENCE STA. SALANDA CONTS SALAND	CH. SENDENCE STA. STA. S 2-6 CNTS C10P S 2-7 CNTS C4DP S 2-8 CNTS C4DP S 2-9 CNTS SVOC S 2-10 CNTS CSSP S 2-11 CNTS CLVN S 2-12 CNTS C4DP S 2-12 CNTS C4DP S 2-13 CNTS C1PC S 2-14 CNTS B4TM OTMD S 2-15 CNTS B4TM OTMD S 2-15 CNTS RFSR S 2-16 CNTS RFSR S 2-17 CNTS ORNG S 2-19 CNTS C40P S 2-20 CNTS C40P S 2-21 CNTS C4DP S 2-23 CNTS C4DP S 2-24 CNTS C4DP S 2-25 CNTS	S 2-6 CNTS C10P VERIFY MODULE POWER SUPPLY "ON". S 2-7 CNTS C4DP TEST S-IV EBM'S FOR READY. S 2-8 CNTS C4DP VERIFY PULSE SENSOR POWER "ON". S 2-9 CNTS SVOC VERIFY PULSE SENSOR POWER "ON". S 2-10 CNTS CSSP VERIFY MSFC TALKBACK ENABLE "ON". S 2-11 CNTS CLVN DESTRUCT ENABLE "ON". S 2-12 CNTS C4DP VERIFY SAFETY BUS INDICATION "ON". S 2-13 CNTS CIPC TELEMETER D2 "ON". S 2-14 CNTS B4TM VERIFY RASTER. S 2-15 CNTS B4TM VERIFY VISICORDER "ON". S 2-16 CNTS R4VC VERIFY VISICORDER "ON". S 2-17 CNTS RFSR SEQUENCE RECORDERS TO 10 MM/SEC. S 2-18 CNTS C10P FUNCTION SELECTOR TO FIRING UNIT VOLTAGE CHECK. S 2-19 CNTS C40P VERIFY PANEL SELECT TO AUXILIARY RECORDER. S 2-21 CNTS C1DP S-I CDR'S #1 AND #2 "ON". S 2-22 CNTS RHBM SANBORN RECORDER TO 5 MM/SEC. S 2-23 CNTS C4DP S-IV CDR'S #1 AND #2 "ON". S 2-26 CNTS C4DP S-IV CDR'S #1 AND #2 "ON". S 2-27 CNTS C4DP S-IV CDR'S #1 AND #2 "ON". S 2-26 CNTS C4DP S-IV CDR'S #1 AND #2 TO "INTERNAL". S 2-27 CNTS C4DP S-IV CDR'S #1 AND #2 TO "INTERNAL". S 2-26 CNTS C4DP S-IV CDR'S #1 AND #2 TO "INTERNAL". S 2-27 CNTS C4DP S-IV CDR'S #1 AND #2 TO "INTERNAL".

EMERGENCY PROCEDURE (T-410' to T-0')

(AFTER S-I LOX TANKING)

PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

185 7=LSVI=300 SA=7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-370°						
	S	2=29	CNTS	C1DP C4DP	VERIFY CDR°S #1 AND #2 CUTOFF.	
	S	2-30	CNTS	C1DP C4DP	VERIFY S-I AND S-IV EBWIS #1 AND #2 ON INTERNAL AND CHARGED.	
	S	2-31		B4TM RFSR	NOTE ALL ENGINE CUTOFF NOT RECEIVED.	
	S	2-32		RFSR	NOTE HELIUM HEATER CUTOFF NOT RECEIVED.	
	S	2-433	CNTS	CPSO	VERIFY SOI AND S-IV EBWSS #1 AND #2 VOLTAGE INDICATIONS.	
	S	2-34	CNTS	ORNG	DESTRUCT COMMAND.	
	S	2-35	CNTS	svoç	VERIFY S-I AND S-IV DESTRUCT DOES NOT OCCUR ON VEHICLE.	
	S	2-36	CNTS	ORNG	RELEASE DESTRUCT COMMAND.	
	S	2-37	CNTS	RBHM	SANBORN RECORDER TO 5 MM/SEC.	
	S	2~38	CNTS	C1DP C4DP	S=I AND S=IV EBW°S #1 AND #2 TO "EXTERNAL",	
	S	2-39	CNTS	C1DP	VERIFY Sal EBWS #1 AND #2 VOLTAGES #OFF#.	
	S	2-40	CNTS	C4DP	VERIFY SHIV EBWIS #1 AND #2 "OFF" AND NOT READY.	,
	S	2-41	CNTS	C1DP	ARM S-I DESTRUCT S&A (INERT).	
	S	2-42	CNTS	C4DP	ARM S-IV DESTRUCT S&A (INERT).	
	S	2-43	CNTS	CLVN	LIFTOFF ENABLE "ON".	
	S	2-44	CNTS	C1DP C4DP	SIMULATED LIFTOFF "ON",	
	S	2-45	CNTS	RBHM	SANBORN RECORDER TO 25 MM/SEC.	
	S	2=46	CNTS	ORNG	CUTOFF COMMAND AND RELEASE UPON REQUEST	
	S	2=47	CNTS	C1DP C4DP	VERIFY S=I AND S-IV EBW#S #1 AND #2 ON #INTERNAL# AND #CHARGED#.	

EMERGENCY PROCEDURE (T-410' to T-0')

(AFTER S-I LOX TANKING)

PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
3701						
	s	2-48	CNTS	CPSO	VERIFY S=I AND S-IV EBW S #1 AND #2 VOLTAGE INDICATIONS.	
	S	2-49		CSRP	NOTE S=I ENGINES CUTOFF DOES NOT OCCUR.	
	S	2-50		RFSR B4TM	NOTE ALL ENGINE CUTOFF COMMAND. CUTOFF COMMAND.	
	S	2-51	CNTS	RFSR	VERIFY HELIUM HEATER THERMAL CUTOFF COMMAND INDICATION.	
	S	2-52	CNTS	ORNG	DESTRUCT COMMAND.	
	S	2=53	CNTS	SVOC	VERIFY S=I AND S-IV DESTRUCT ON VEHICLE	
	S	2-54	CNTS	ORNG	RELEASE DESTRUCT COMMAND.	
	S	2-55	CNTS	RBHM	SANBORN RECORDER TO 5 MM/SEC,	
	S	2-56	CNTS	ORNG	SWITCH TO COMMAND TRANSMITTER #2.	
	S	2-57	CNTS	SVOC	RESET PULSE SENSORS.	
	S	2~58	CNTS	C1DP C4DP	S=I AND S=IV EBW S #1 AND #2 TO "EXTERNAL"	
	S	2=59	CNTS	RFSR B4TM	VERIFY ALL ENGINE CUTOFF COMMAND REMOVED.	
	S	2-60	CNTS	C1DP C4DP	VERIFY S=I AND S=IV EBW*S #1 AND #2	
	S	2-61	CNTS	CSPC	SEQUENCER RELAY POWER "OFF", THEN "ON".	
	S	2-62	CNTS		VERIFY LOSS OF HELIUM HEATER CUTOFF COMMAND INDICATION.	
	S	2-63	CNTS	RBHM	SANBORN RECORDER TO 25 MM/SEC.	
	S	2-64	CNTS	ORNG	CUTOFF COMMAND AND RELEASE UPON REQUEST	
	S	2=65	CNTS		VERIFY S-I AND S-IV EBWIS #1 AND #2 ON "INTERNAL" AND "CHARGED".	

EMERGENCY PROCEDURE (T-410' to T-0')

(AFTER S-I LOX TANKING)

PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-370°						
	S	2=66	CNTS	CPSO	VERIFY S=I AND S=IV EBW S #1 AND #2 VOLTAGE INDICATIONS.	
	S	2-67		CSRP	NOTE Sal ENGINES CUTOFF DOES NOT OCCUR.	
	S	2=68	CNTS	RFSR B4TM	VERIFY ALL ENGINE CUTOFF COMMAND.	
	S	2-69	CNTS	RFSR	VERIFY HELIUM HEATER CUTOFF COMMAND INDICATION.	
	S	2-70	CNTS	ORNG	DESTRUCT COMMAND.	
	S	2-71	CNTS	SVOÇ	VERIFY S=I AND S=IV DESTRUCT ON VEHICLE	
	S	2-72	CNTS	ORNG	RELEASE DESTRUCT COMMAND.	
	S	2=73	CNTS	RBHM	SANBORN RECORDER TO 5 MM/SEC.	
	S	2-74	CNTS	SVOC	RESET PULSE SENSORS.	
	S	2-75	CNTS	C1DP C4DP	SIMULATED LIFTOFFS "OFF".	
	S	2=76	CNTS	RFSR B4TM	VERIFY ALL ENGINE CUTOFF REMOVED.	
	S	2=77	CNTS	C1DP C4DP	S=I AND S=IV EBW S #1 AND #2 TO PEXTERNAL 99 .	
	S	2=78	CNTS	C1DP C4DP	VERIFY S=I AND S=IV EBW S #1 AND #2 "OFF" AND "NOT READY".	
	S	2-79	CNTS	CSPC	SEQUENCER RELAY POWER "OFF", THEN "ON",	
	S	2=80	CNTS		VERIFY LOSS OF HELIUM HEATER CUTOFF COMMAND INDICATION.	
	S	2-81	CNTS	RBHM	SANBORN RECORDER TO 25 MM/SEC.	
	S	2~82	CNTS	ORNG	CUTOFF COMMAND AND RELEASE UPON REQUEST	
	S	2-83	CNTS		VERIFY S=I AND S-IV EBW \$ #1 AND #2 TO "INTERNAL" AND "CHARGED".	
	S	2 = 84		RFSR B4TM	NOTE ALL ENGINE CUTOFF NOT RECEIVED.	

EMERGENCY PROCEDURE (T-410' to T-0')

(AFTER S-I LOX TANKING)

PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-370 ⁰						
	S	2=85	CNTS	RFSR	NOTE HELIUM HEATER CUTOFF COMMAND NOT RECEIVED.	
	S	2-86	CNTS	CPSO	VERIFY S≈I AND S≈IV EBW®S #1 AND #2 VOLTAGE INDICATIONS.	
	S	2-87	CNTS	ORNG	DESTRUCT COMMAND.	
	S	2~88	CNTS	syoc	VERIFY S-I AND S-IV DESTRUCT DOES NOT OCCUR ON VEHICLE.	
	S	2-89	CNTS	ORNG	RELEASE DESTRUCT COMMAND.	
	S	2-90	CNTS	RBHM	SANBORN RECORDER TO 5 MM/SEC.	
	S	2-91	CNTS	C4DP	Smiv CDR SWITCHES #1 AND #2 "OFF".	
	S	2-92	CNTS	C4DP	VERIFY S-IV CDROS #1 AND #2 REMAIN "ON"	
	S	2-93	CNTS	ORNG	SAFE COMMAND AND RELEASE UPON REQUEST.	
	S	2=94	CNTS	C4DP	VERIFY SAFE COMMAND RECEIVED,	
	S	2=95	CNTS	C4DP B4TM	VERIFY SEIV CDR'S #1 AND #2 AND EBW'S #1 AND #2 REMAIN "ON".	
	S	2~96	CNTS	C1DP C4DP	SRI AND SRIV EBW'S #1 AND #2 TO	
	S	2~97	CNTS	C1DP C4DP	VERIFY S-I AND S-IV EBW'S #1 AND #2	
	S	2=98	CNTS	RBHM	SANBORN RECORDER TO 2 MM/SEC.	
	S	2-99	CNTS	C1PC	TELEMETER D2 "OFF".	
	S	2-100	CNTS	B4TM	TELEMETER RECORDING "OFF".	
	S	2-101	CNTS	C1MP	TELEMETER F2 "OFF"	
	S	2-102	CNTS	C1MP	AUXILIARY EQUIPMENT "OFF".	
	S	2-103	CNTS	ОТМО	TELEMETER RECORDING "OFF".	
	S	2=104	CNTS	C1MP CUMP	TELEMETER CAL TO "PREFLIGHT".	

EMERGENCY PROCEDURE (T-410' to T-0')

(AFTER S-I LOX TANKING)

PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-370'						
	S	2-105	CNTS	C1DP	SAFE S-I DESTRUCT S&A (INERT).	
	S	2-106	CNTS	C4DP	SAFE S-IV DESTRUCT S&A (INERT).	
	S	2-107	CNTS	CLVN	LIFTOFF ENABLE HOFFH.	
	S	2-108	CNTS	C4DP	S=IV CDR'S #1 AND #2 TO "EXTERNAL" AND VERIFY "OFF".	
	S	2-109	CNTS	C1DP	S=I CDR'S #1 AND #2 TO "EXTERNAL".	
	S	2-110	CNTS	C1DP	S=I CDR+S #1 AND #2 "OFF",	
	S	2-111	CNTS	ORNG	COMMAND CARRIER "OFF".	
	S	2-112	CNTS	CLVN	DESTRUCT ENABLE "OFF".	
	S	2-113	CNTS	C4DP	VERIFY LOSS OF SAFETY BUS ON INDICATION	
	S	2-114	CNTS	RFSR	SEQUENCE RECORDERS TO 2 MM/SEC.	
	S	2-115	CNTS	R4VC	VISICORDER "OFF".	
	S	2-116	CNTS	CLVN	REMOVE DESTRUCT AND LIFTOFF ENABLE SWITCH KEYS AND GIVE TO THE PAD SAFETY REPRESENTATIVE IN THE BLOCKHOUSE.	7
	S	2-117	CNTS	ANAG	REMOVE JUMPER 84A1J17/LITTLE J TO 84A1J20/BIG F.	
	S	2-118	CNTS		SEQUENCE RECORDERS TO HOUR SPEED, TIME PULSE "OFF",	
	S	3		CNTS	VERIFY COMMAND CHECKS COMPLETE.	
		4			REMOVE PRELAUNCH COVERS FROM STROBE LIGHTS AND FIBEROPTIC LENSES AFTER S-IV AFT ACCESS KIT REMOVAL.	
3651						
		1			REMOVE LOX VENT LINES (DO NOT REMOVE VENT II-III UNTIL IMMEDIATELY AFTER S=I S&A CONNECTION).	
		2			INSTALL HEAT SHEILD PANELS 30M03497 (FOUR EACH).	

EMERGENCY PROCEDURE (T-410' to T-0')

(AFTER S-I LOX TANKING)

PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

190 7=LSVI=300 SA=7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
3651						
		3		SVMP	INSTALL ESCAPE CHUTE COVER AND CONNECT CALORIMETER PURGE LINE.	
		4		S1VM	STOW WORK PLATFORMS ON ADJUSTABLE ONE.	
		5	C1TC	SVME	VERIFY ESCAPE CHUTE AND HEAT SHEILD PANEL MEASUREMENTS ARE CONNECTED AND PROTECTIVE COVERS REMOVED:	
×360°						
		1	CLTC	RGCC	REQUEST APS 100 FLIGHT COMPUTER SYSTEM POWER "ON",	
		5	¥	C1NP	VERIFY LCC ESE PER PROCEDURE 1-LLVI-706	
		3		SVOR	SEQUENCE RECORDERS "OFF", TIME PULSE	
		4		CPSP	AFTER ROGER FROM SVOC, TURN SERVICE STRUCTURE OAT POWER "OFF".	
355°						
		1		SVMP	VERIFY ALL BAY DOORS SECURED FOR LAUNCH	
		5		PLOF	TOP OFF LN2 150-GALLON TANK.	
		3		SGMP	VERIFY ALL LAUNCHER AUXILIARY PLATFORMS ARE REMOVED AND SECURED FOR LAUNCH.	
	SC	4		CSTC	VERIFY FLIGHT BATTERIES CONNECTIONS COMPLETE.	
-350 *						
	PA	4		CNTS	RF SILENCE "ON" AND CONTROLLED SWITCHING.	
		2		SGNP	AFTER ROGER FROM SVOC. C4TC, BCDC AND SCTC, DISCONNECT SERVICE STRUCTURE TEST CABLES,	
		3		SGMP	SECURE TEST CABLE AND LOX REPLENISH ACCESS COVER ON SOUTH SIDE OF LAUNCHER.	

EMERGENCY PROCEDURE (T-410' to T-0')

(AFTER S-I LOX TANKING)

PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
≈350¹						
		4		SUVN S1NF	DISCONNECT PULSE SENSORS AND ASSOCIATED CABLING FOR S=I AND S=IV DESTRUCT.	
		5		SVOC	SECURE 136 FOOT LEVEL OAT ROOM.	
	BL1	6	C1TC	S1NF	CONNECT Sal DESTRUCT S&A UNIT PER PROCEDURE 1-LSII-710, PART II.	
	BL1	7	C1TC	S1NF	CONNECT Sal DESTRUCT DETONATORS PER PROCEDURE 1-LSII-708.	
	RD2	8	C4TC	CVTE	CONNECT S-IV DESTRUCT S&A PER PROCEDURE DAC-0-7012,	
		9		SUMP	CHARGE HYDRAULIC ACCUMULATORS TO 1600 PSIG, LEAK CHECK SHRADER FITTING AND SECURE FOR LAUNCH: POSITION 1 POSITION 2 POSITION 3 POSITION 4	
		10		SVMP	INSTALL AN 929-12C CAPS ON HYDRAULIC SYSTEM LOW PRESSURE RELIEF VALVES: POSITION 1 POSITION 2 POSITION 3 POSITION 4	
		11		SVMP	INSTALL FINAL HEAT SHIELD PANELS.	
	SIV	12	C4TC	S4ME	PREPARE SAIV AREAS FOR SERVICE STRUCTURE REMOVAL.	
		13			VERIFY ALL LOOSE ITEMS ON ADJUSTABLE LEVELS 1, 2 AND 3 REMOVED OR MOVED BACK AND TIED DOWN SECURELY.	
		14		S4ME	VERIFY ALL ELECTRICAL, MECHANICAL AND PNEUMATIC CONNECTIONS BETWEEN FIXED AND MOVEABLE PLATFORMS ON ADJUSTABLE LEVELS 1, 2 AND 3 DISCONNECTED.	
		15	C4TC		VERIFY EJECT PNEUMATIC LINES CONNECTED TO EJECT MECHANISMS ON SWING ARM 2 AND GH2 VENT COUPLING.	
		16		S4ME	REMOVE SAFETY WIRE FROM SWING ARM 2 AND GH2 VENT COUPLING EJECT	

EMERGENCY PROCEDURE (T-110' TO T-0')

(AFTER S-I LOX TANKING)

PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
350'						
					MECHANISMS.	
		17		S4ME	REPLACE STEEL SHEAR PINS WITH ALUMINUM SHEAR PINS ON SWING ARM 2 EJECT MECHANISM.	
		18		S4ME	REMOVE SAFETY BLOCK FROM GH2 VENT COUPLING EJECT MECHANISM.	
		19		S4ME	VERIFY SWING ARM 2 AND GH2 VENT COUPLING SOLENOIDS ELECTRICALLY CONNECTED.	
		20		S4ME	CONNECT BUNGEE LANYARD TO GH2 COUPLING.	
	7	21		S4ME	CONNECT RETRACT LANYARDS TO SWING ARM 2 AND GH2 VENT COUPLING CARRIERS.	
	BL1	22	C4TC	CSPC	SERVICE TOWER POWER ENABLE "OFF".	
		23		RECS	START ECS AIR TO THE S-IV AFT SECTION:	
-345 '						
	RD3	1		APCD	PREPARE THE PNEUMATIC DISTRIBUTION SYSTEM FOR LAUNCH PER PROCEDURE LVO-E-7020.	
	RD3	2	C1TC		COMPLETE SWING ARM PREPARATIONS PER LV0-E=7016,	
	RD3	3	C1TC		COMPLETE Q=BALL RETRACT SYSTEM PREPARATIONS PER LVO-E-7019.	
		4		SVMP	REMOVE BAY EXTERNAL ACCESS LADDERS.	
340 !						
	YW4	1		SSSC	OPEN SERVICE STRUCTURE SILO C=1 AND RETRACT PLATFORM #4.	
-335'						
	YW3	1	CLTC	RGCC	INITIATE APS 101, PSEUDO FLIGHT TEST.	
		2			REMOVE PROTECTIVE COVERS FROM TEMP. MEASUREMENTS ON BLAST PLATES AND LOX/SOX SYSTEM.	

EMERGENCY PROCEDURE (T-410' to T-0')

(AFTER S-I LOX TANKING)

PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

LAUNCH COUNTDOWN DATE SEPTEMBER 5, 1964 LAUNCH OPERATIONS REVISION

SATURN/APOLLO

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-335°						
		3		CSTC	VERIFY COMMAND MODULE INSPECTION COMPLETED,	
		,				
				s.		

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

·330 '			STA.	STA.	DESCRIPTION	REMARKS
	BL1	1	C1TC	S1NA	CONNECT S.P.G.G. INITIATORS PER PROCEDURE 1-LSII-709, PART II.	
					NOTE	
					IF CUTOFF IS REQUIRED BECAUSE OF AN ACCIDENTAL IGNITION, HOLD THE CUTOFF BUTTON UNTIL PREVALVES 1 THRU 8 ARE "OFF".	
		2		SVMP	INSTALL INITIATOR ACCESS HAND HOLE COVERS AS INITIATOR CONNECTIONS ARE COMPLETED.	
E	BL1	3	C1TC	S1NF	VERIFY SEI DESTRUCT S&A UNIT AND DETONATORS CONNECTED.	
F	RD2	4	C4TC	CVTE	VERIFY S=IV S&A CONNECTED FOR FLIGHT.	
		5	CVTE		REMOVE THE REMAINING 2 LH2 TANK PROTECTIVE PADS FROM THE FORWARD INTERSTAGE,	
		6	CVTE	SUVM	REMOVE IU/S=IV FORWARD ACCESS KIT DOOR PAN,	
		7		CVTE	VERIFY S-IV FORWARD INTERSTAGE HAS BEEN INSPECTED FOR FLIGHT CONFIGURATION	
		8			INSTALL AND SECURE THE 5863 S=IV FORWARD INTERSTAGE DOOR.	
		9		S1VM	REMOVE LOX TANK CAMERA PURGE VENT LINE.	,
		10			REMOVE REMAINING LOX VENT LINE BETWEEN II=III AFTER S-I S&A CONNECTION IS COMPLETED.	
	1	11			VERIFY ECS AIR ON IN THE S=IV AFT SECTION. OPEN ENGINE INJECTOR PURGE HAND VALVE AND LOX VENT PILOT HAND VALVE AND SAFETY WIRE BOTH IN OPEN POSITION.	
В	3L2 1	L2	C4TC		VERIFY COMPLETION OF S-IV AFT ACCESS KIT REMOVAL PER PROCEDURE DAC-L-7002.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-325°						
		1		SVMP	SECURE GN2 PANELS ON ENGINE SERVICE PLATFORM.	
		2		SGMP	SECURE AUXILIARY GN2 CONNECTIONS INSIDE LAUNCHER RING FOR LAUNCH.	
-315°						
	BL1	1	RS4C	ANAG	ST-124 HEATER SWITCH TO NORMAL POSITION.	
-310 °						
	s	1	CLTC	SVME	VERIFY MEASURING READY TO OPEN SILO B=3	
	RD2	2	C4TC	CVTE	VERIFY COMPLETION OF S-IV DESTRUCT BLOCK CONNECTION PER PROCEDURE DAC-0-7012.	
	SIV	3		S4ME	VERIFY S-IV AREAS READY FOR SERVICE STRUCTURE REMOVAL.	
·305°				A .		
	BL1	1.		I	SECURE UMBILICAL TOWER ESE PER PROCEDURE 1=LLVI=715.	
	Y W 4	2	CLTC	SSSC SMTE	OPEN SERVICE STRUCTURE SILO B=3 AND RETRACT PLATFORMS #2 AND #3.	
	sc	3		CSTC	BEGIN SPACECRAFT ECS PURGE.	
-295°						
		1		SGMP	BEGIN PREPARATIONS TO SECURE ENGINE SERVICE PLATFORM BENEATH LAUNCHER PER PROCEDURE LVO-E=7020.	
· 290 °						
	BL1	1	C1TC	S1NA	VERIFY S.P.G.G. INITIATOR CONNECTION COMPLETE.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

SATURN/APOLLO

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
~290 °						
		2		SVMP	FINAL S-I TAIL SECURING OPERATIONS,	
-285						
		1		C1TC RBHM	PERFORM LOX BUBBLING CHECK.	
		2		SVMP	COMPLETE INITIATOR ACCESS HAND HOLE COVER INSTALLATION.	
		3		C1FP SVMP	INITIATE CALORIMETER PURGE AND VERIFY FLOW AT ALL 10 LOCATIONS.	
~280°						
	S	1	CLTC		VERIFY ALL NETWORK CABLES ARE CLEAR FOR STRUCTURE REMOVAL.	
-275 ⁰						
	BL1	1			SECURE LAUNCHER ESE PER PROCEDURE 1-LLVI-714.	
	S	2	CLTC	RWCP	VERIFY LAUNCH VALVE "OPEN";	
-270°						
		1		SVMP	COMPLETE S=1 TAIL SECURING OPERATIONS.	
	RD3	2	C1TC	SGMP	SECURE ENGINE SERVICE PLATFORM FOR LAUNCH PER PROCEDURE LV0-E=7021:	
		3	CLTC		OBTAIN PPSO VERIFICATION THAT INSIDE OF LAUNCHER IS CLEAR OF ALL PERSONNEL AND SECURE LAUNCHER BLAST DOOR AND WINDOW FOR LAUNCH.	
	SC	4			VERIFY COMPLETION OF SPACECRAFT HATCH INSTALLATION.	
	sc	5			VERIFY COMPLETION OF SPACECRAFT ECS PURGE.	

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- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	сомм.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-265°						
	SC	1		CSTC	VERIFY ALL SPACECRAFT PERSONNEL ARE CLEARING THE PAD AREA.	
	Y W 3	2	CLTC	SSSC	OPEN SERVICE STRUCTURE SILO C-2 AND RETRACT PLATFORMS NO. 5 AND 6. NOTE SPACECRAFT PAD LEADER WILL REMAIN UNTIL SILO CLEARS SPACECRAFT.	
~260°	PA	3		CNTS	ANNOUNCE TO CLEAR PAD FOR S-IV LOX TANKING.	
200						
	BL1	1		ANAG	IGNITION CIRCUIT BREAKER "ON",	
	BL1	2		ANAG	IGNITION CONTACTOR ON AND SAFETY WIRE,	
	BL1	3		ANAG	VERIFY IGNITION RACK SWITCH IN SAFE POSITION:	
	BL1	4		ANAG	VERIFY AGCS ESE SECURED PER PROCEDURE 1=LLV1=713.	
- 245 °						
	PA	1		CNTS	RF SILENCE "OFF",	
	SRO	2	CNTS	ORNG	FREQUENCY CLEARANCE FOR THE FOLLOWING TM LINKS = 244.8, 251.5, AND 258.5 MC.	
	S	3	CLTC	PPS0	VERIFY PAD AREA CLEAR FOR S-IV LOX LOADING.	
	RD4	4	CLTC	RLOO	LOAD S-IV LOX PER PROCEDURE LVO-L-1004.	
					NOTE TM SYSTEMS ON AND OFF AS REQUIRED.	
-215	RD1	5	C1TC	C1VP	PRESSURIZE CONTROL SPHERES TO 3000 PSIG	
	RD4	1		RLOO	VERIFY Salv LOX LOADING COMPLETE.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

KSC FORM 23-81 B (5/64)

SATURN/APOLLO LAUNCH OPERATIONS

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
~215'						
		2		CIPC	TELEMETERS D1. D2 AND D3 "OFF".	
	RD4	3		S4PE U4PE	S-IV LEAK CHECK CREW RETURN TO SERVICE STRUCTURE AND UMBILICAL TOWER AND LEAK CHECK PER PROCEDURE DAC-P-7007.	
-1951		×				
		1		C4TC	VERIFY S=IV LEAK CHECKS COMPLETE PER PROCEDURE DAC=P=7007.	
		2		S4SE	CLOSE S-IV AFT INTERSTAGE.	
-1851			2			
		1		C4TC	VERIFY AFT INTERSTAGE SECURE FOR FLIGHT PER PROCEDURE DAC-P=7007.	
	YW4	2	CLTC	SSSC SMTE	OPEN SERVICE STRUCTURE SILO B-2 AND RETRACT PLATFORM #1.	
≈180°						
		1		CUNC	COOLING TO PREFLIGHT.	
	YW1	2	CLTC	RS4C	BRING ST=124 TO OPERATING CONDITIONS PER PROCEDURE LVO=G=7041.	
	YW1	3	CLTC	RATC	BRING AUTO THEODOLITE UP TO OPERATING CONDITIONS PER PROCEDURE LVO-G-7044.	
-170'				v		
	YW1	1	CLTC		MOVE SERVICE STRUCTURE TO LAUNCH LOCATION PER PROCEDURE LSO-F-1001.	
-165°						
	PA	1			CLEAR BLOCKHOUSE AND PARKING LOT OF NON-ASSIGNED PERSONNEL TO THE ROAD BLOCKS.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE 199 7=LSVI=300 SA=7

TIME	сомм.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-150°						
		1		CPSO	SET BLAST DANGER AREA ROAD BLOCKS. AREA TO BE CLEAR BY T-115.	
	SRO	2	CNTS	ORNG	DO NOT INTERROGATE C-BAND BEACON WITH 1,16 RADAR UNTIL REQUESTED.	
145 9						
	S	1	CLTC	RWCP	VERIFY WATER PRESSURE TO 175 PSIG.	
		2	,		RF AND TM CHECKS.	
	S	2-1	CLTC	BDOP	ODOP GROUND TRANSMITTER NO. 2 "ON".	
	S	2-2	CLTC	C1CM	TV TO FILAMENT.	
	S	2-3	CLTC	CUMP	ALTIMETER "ON".	
	S	2-4	CLTC	CUMP	ODOP MONO,	
	S	2-5	CLTC	CUMP	MINITRACK "ON".	
	S	2-6	CLTC	CUMP	AZUSA "ON",	
	S	2-7	CLTC	CUMP	C-BAND BEACON (RADAR) "ON",	
	S	2-8	CLTC	CUMP	MISTRAM "ON".	
	S	2-9	CLTC		TELEMETERS F1, F2, F3, P2, S1 AND S2 "ON".	
	S	2-10	CLTC	C1MP	VERIFY TAPE RECORDER READY INDICATION.	
	S	2-11	CLTC	CUMP	TELEMETERS F5, F6, S3, AND P1 "ON",	
	S	2-12	CLTC	CUMP	VERIFY TAPE RECORDER READY INDICATION.	
	S	2-13	CLTC	C1MP CUMP	AUXILIARY EQUIPMENTS "ON",	
	S	2-14		C1MP CUMP	SWEEP FREQUENCY CALIBRATOR "ON",	
	S	2-15	CLTC	CIPC	TELEMETERS D1, D2 AND D3 "ON",	
	S	2-16		C1MP	TELEMETER CALIBRATIONS TO "PREFLIGHT".	

- 1. C1NP: Give cutoff (Caution: Hold ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

	DESCRIPTION	STA.	COMMAND STA.	SEQUENCE	COMM. CH.	TIME
						-145°
FY	CALIBRATION SWITCH TO MANUAL (VERIFY VCO CALIBRATION LIGHT OFF).	CIPC	CLTC	2-17	S	
	PREFLIGHT CALIBRATIONS TO "50%".	C1MP CUMP	CLTC	2~18	S	
	INTERROGATE MISTRAM AND AZUSA AND REPORT READOUT.	ORNG	CNTS	2-19	SRO	
						-140 0
	RF AND TM CHECKS.			1		
	TV TO B.	C1CM	CLTC	1-1	S	
	TELEMETER RECORDINGS "ON",	B1TM B4TM OTMD	CLTC	1-2	S	
AND 99 。	TELEMETER CALIBRATIONS TO "AC COMMAND".	C1MP CUMP	CLTC	1-3	S	
	INTERROGATE IU C-BAND BEACON WITH 1,16 RADAR AND REPORT READOUT.		CNTS	<u>1</u> = 4	SRO	
						-139 0
	RF AND TM CHECKS.	1		1		
	TELEMETER CALIBRATION COMMANDS	C1MP		1-1	S	
						-138 ° 45 "
	RF AND TM CHECKS.	F		1		
HT " .	TELEMETER CALIBRATIONS TO "PREFLIGHT".	C1MP CUMP		1-1	S	
	TELEMETER CALIBRATION COMMANDS.	C1MP CUMP		1-2	S	
	TV TO B TELEMETER RECORDINGS "ON". TELEMETER CALIBRATIONS TO "AC COMMA INTERROGATE IU C-BAND BEACON WITH 1.16 RADAR AND REPORT READOUT. RF AND TM CHECKS. TELEMETER CALIBRATION COMMANDS "ON" AND "OFF". RF AND TM CHECKS. TELEMETER CALIBRATIONS TO "PREFLIGH	C1CM B1TM B4TM OTMD C1MP ORNG C1MP CUMP C1MP C1MP C1MP	CLTC CLTC CLTC	1 = 1 1 = 2 1 = 3 1 = 4	S SRO S	~138°

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
~138° 45"	S	1-3	CLTC	C1MP CUMP	PREFLIGHT CALIBRATION TO "0%".	
-137º 30º		1		,	RF AND TM CHECKS.	
136 °	S	1-1	CLTC	C1MP CUMP	PREFLIGHT CALIBRATIONS TO "100%".	
30"		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1MP CUMP	RECORDER TRANSFERS "ON",	
	S	1-2	CLTG	B1TM OTMD	8 KC OSCILLATORS "ON".	
	S	1-3	CLTC	C1MP CUMP	TELEMETER CALIBRATIONS COMMANDS.	
	S	1 - 4	CLTC	C1MP CUMP	PREFLIGHT CALIBRATIONS TO 0%, 25%, 50%, 75%, AND 100% (IN 2 SECOND INCREMENTS).	
	S	1-5	CLTC	CIPC	STEP THROUGH MANUAL CALIBRATION IN 2 SECOND INCREMENTS.	
	S	1-6	CLTC	C1MP CUMP	TELEMETER CALIBRATION TO "INFLIGHT".	
	S	1-7	CLTC	RBHM	S=I AND IU HI CAL, COMMAND FOR 15 SEC.	
	S	1-8	CLTC	RBHM	S-I AND IU LOW CAL COMMAND FOR 15 SEC.	
	S	1-9	CLTC	RBHM	S-I AND IU RUN COMMAND.	
	S	1-10	CLTC	CIPC	CALIBRATION TO AUTOMATIC.	
	S	1-11	CLTC	CIPC	GIVE CAL START.	
	S	1-12	CLTC		TELEMETER CALIBRATION COMMANDS "ON" FOR 5 SECONDS.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

	ſ		7			3 A =
TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-136' 30"						
	S	1-13	CLTC	CIPC	BRIDGE CALIBRATION HIGH FOR 10 SECONDS.	
	S	1-14	CLTC	CIPC	BRIDGE CALIBRATION LOW FOR 10 SECONDS.	
		1-15	CLTC	CUMP	P1 CALIBRATION COMMAND "ON" AND "OFF".	
	S	1=16	CLTC	B1TM, OTMD	8 KC OSCILLATOR OFF.	
-135'		g.				
	RD4	1	C1TC	PPSO	VERIFY PAD AREA CLEAR TO BEGIN S-I LOX PRECOOL.	
	RD4	2	C1TC	RLOO	BEGIN S-I FINAL LOX TANKING PER PROSCEDURE LVOSL=1004 (AFTER RECORDER TRANSFER OFF).	
=134' 30"						
		1.			RF AND TM CHECKS.	
	S	1-1	CLTC	C1MP CUMP	RECORDER TRANSFERS "OFF".	
	S	1-2	CLTC	CIPC	GIVE CAL START.	
	S	1 - 3	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMAND "ON" FOR 5 SECONDS.	
	S	1-4	CLTC	CUMP	P1 CALIBRATION COMMAND "ON" AND "OFF".	
,	S	1-5	CLTC	C1MP	TELEMETER CALIBRATION TO "PREFLIGHT".	
-134'						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	CUMP	ODOP "OFF",	
	S	1-2	CLTC	BDOP	ODOP GROUND TRANSMITTERS #2 "OFF".	
	S	1-3	CLTC		TELEMETERS F1, F2, F3, P2, S1 AND S2 "OFF",	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

SATURN/APOLLO

PAGE

203 TEST NO. 7 = L SVI = 300
VEHICLE SA = 7 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
=134 °						
	S	1-4	CLTC	CUMP	TELEMETERS F5, S3 AND MINITRACK "OFF",	
	S	1-5	CLTC	C1MP	TELEMETER AUXILIARY EQUIPMENT "OFF",	
	S	1-6	CLTC	CIPC	TELEMETERS D1, D2, AND D3 "OFF",	
	S	1-7	CLTC	В4ТМ	S-IV TELEMETER RECORDING "OFF".	
130						
		1			RF AND TM CHECKS.	
	S	1 = 1	CLTC	C1CM	TV "OFF";	
	S	1-2	CLTC	CUMP	C-BAND BEACON (RADAR) "OFF" WHEN READOUT COMPLETE.	
	S	1-3	CLTC	CUMP	AZUSA "OFF" WHEN READOUT IS COMPLETE.	
	S	1-4	CLTC	CUMP	MISTRAM "OFF" WHEN READOUT IS COMPLETE.	
120 °						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	CUMP	ALTIMETER "OFF",	
	S	1-2	CLTC	CUMP	TELEMETERS F6 AND P1 "OFF".	
	S	1-3	CLTC	CUMP	TELEMETER AUXILIARY EQUIPMENT "OFF".	
	S	1-4	CLTC		TELEMETER CALIBRATION TO "PREFLIGHT".	
	S	1-5		OTMD	TELEMETER RECORDINGS "OFF".	
	YW1	2		RATC	ST-124 TO FLIGHT AZIMUTH PER PROCEDURE LV0-G-7042.	
	S	3	CLTC		CONTROL COMPUTER "ON" (VERIFY "CAT ZERO"),	
	S	4	CLTC	RVFC	CONTROL ACCELEROMETERS "ON".	
	S	5	CLTC	RVFC	IU RATE GYROS "ON" .	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
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PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-120'						
	S	6	CLTC	RVEC	S=I RATE GYROS HONH.	
	S	7	CLTC	RVFC		
	S				Q-BALL SYSTEM "ON"	
		8	CLTC	RVFC	Q-BALL HEATERS "ON".	
	S	9	CLTC	RVFC	ACCELERATION SWITCH "ON".	
≈115'						
	S	1	CLTC	RGCC	INITIATE APS 103, FLIGHT COMPUTER SYSTEM FINAL LAUNCH PREPARATION.	
-100°						
	S	1		RECS	SWITCH ECS MODE FROM AIR TO GN2 PER PROCEDURE LVO-E-7018.	
	YW4	2			VERIFY SERVICE STRUCTURE SECURED AT LAUNCH LOCATION. PERSONNEL CLEAR AREA TO ROADBLOCK OR BLOCKHOUSE AS ASSIGNED.	
-901						
		1		CPSQ	VERIFY COMPLEX CLEAR OF ALL PERSONNEL AND SECURE BLOCKHOUSE DOOR.	
	S	2	CLTC	RVFC	CONTROL VOLTAGE "ON".	
-85!						
		1		C1FP	DELUGE PURGE TO AUTO.	
a	S	2	CLTC	CPSP	GENERATOR COMMIT ON ALL GENERATORS.	
	S	3	CLTC	CPSO	RELEASE DESTRUCT ENABLE KEY TO CLVN.	
	S	4	CLTC	RBHM	SANBORN RECORDER TO 5 MM/SEC.	
	S	5	CLTC	C4DP	VERIFY S&A SAFE.	
	S	6	CLTC	CLVN	DESTRUCT ENABLE ON.	
	S	7	CLTC	C40P	PANEL SELECT TO AUXILIARY RECORDER	

- 1. C1NP: Give cutoff (Caution: Hold ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
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PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
≈85°						
	S	8	CLTC	R4VC	VERIFY VISICORDER ON.	
	S	9	CLTC	C1DP C4DP	S=I AND S=IV EBW S #1 AND #2 ON.	
	S	10	CLTC	C1DP C4DP	VERIFY S-I AND S-IV EBW #1 AND #2 VOLTAGES OK.	
	S	11	CLTC	C1DP C4DP	S=I AND S=IV EBW S #1 AND #2 "OFF",	
	S	12	CLTC	RBHM	SANBORN RECORDERS TO 2 MM/SEC.	
	S	13	CLTC	CLVN	DESTRUCT ENABLE OFF AND DELIVER DESTRUCT ENABLE KEY TO CPSO.	
		14		CPSO	VERIFY BLAST DANGER AREA SECURE FOR LH2 TANKING.	
-801						
	RD4	1		RLOO	COMPLETE S-I LOX LOADING TO 95%.	
	RD4	2		RLOO	BEGIN S-I AND S-IV LOX REPLENISH.	
	RD4	3	RLHC	RECS	VERIFY VEHICLE HAS BEEN PURGED WITH GN2 FOR AT LEAST 15 MINUTES PRIOR TO LH2 LOADING.	
	RD4	4			BEGIN S-IV LH2 LOADING PER PROCEDURE LVO-L-1033 (NOTE: TM SYSTEMS "ON" AND "OFF" AS REQUIRED).	
		5			SET LAUNCH DANGER AREA ROADBLOCKS (CAPE ROAD TO REMAIN OPEN UNTIL T-50).	
-721						
	sc	1		CSTC	SPACECRAFT EXTERNAL POWER ON.	
-70°						
	RD1	1			PERFORM S=I LOX BUBBLING CHECK (10 MINUTES AFTER S=I LOX LOADING COMPLETE)	

- 1. C1NP: Give cutoff (Caution: Hold ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

	CH.	SEQUENCE	STA.	RESPONSE STA.	DESCRIPTION	REMARKS
701						
,						
	RD1	2	C1TC	C1VP	PRESSURIZE HELIUM SPHERE TO 3000 PSIG.	
	RD1	3	C1TC	C1VP	PRESSURIZE FUEL SPHERES AND CAMERA PURGE SPHERES TO 3000 PSIG.	
	RD1	4	C1TC	C1VP	PRESSURIZE CAMERA EJECT SPHERE TO 3000 PSIG.	
	RD1	5	CLTC	CUNC	OPERATE GAS BEARING SWITCH "ON".	
621						
	SC	1	CSTC	CICM	SPACECRAFT TELEMETER A (LINK 1) "ON".	
	SC	2	CSTC	CICM	SPACECRAFT TELEMETER B (LINK 6) "ON".	
	SC	3	CSTC	CICM	SPACECRAFT TELEMETER C (LINK 14) "ON".	
-60 9						
	SRO	1	CNTS	ORNG	VERIFY PATRICK 0-18 RADAR AWAY FROM PAD	
	SC	2	CSTC	CICM	SPACECRAFT RADAR BEACON #1 "ON".	
	sc	3	CSTC	CICM	SPACECRAFT RADAR BEACON #2 "ON".	
	SRO	4	CNTS		ONE (1) MINUTE AFTER NOTIFICATION OF BEACON ON, BEGIN INTERROGATION OF BOTH BEACONS AND PROCEED WITH READOUT, REPORT READOUT WHEN COMPLETE.	
551						
	SRO	1	CNTS	ORNG	BEGIN READOUT OF SPACECRAFT TELEMETERS.	
50 9						
		1			START INCREASING COLD HELIUM SPHERES PRESSURE TO 3000 PSIG WHEN LH2 LEVEL REACHES 70%.	
	S	2	CLTC	C1NP CUNP	VERIFY FLIGHT SIMULATION SWITCHES "OFF"	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

SATURN/APOLLO

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSI	DESCRIPTION	REMARKS
-40°						
	SRO	1	CNTS	ORNG	REPORT STATUS OF SPACECRAFT RADAR BEACONS (GO/NO GO).	
= 35 1						
	RD4	1		RLHC	COMPLETE LH2 LOADING.	
		2		CIPC	TELEMETERS D1, D2 AND D3 OFF.	
	SIV	3		C4TC	PERFORM S-IV TERMINAL COUNTDOWN PER PROCEDURE DAC-N-1004.	
	RD4	4		CPTE RLOO	PERFORM S=IV LOX BUBBLING CHECK PER PROCEDURE DAC=P=7010 (AFTER LH2 LOADING IS COMPLETE).	
-321						
	SRO	1	CNTS	ORNG	REPORT STATUS OF SPACECRAFT TELEMETERS (GO/NO GO).	
-301						
	S	1	CLTC		VERIFY FLIGHT COMPUTER CHECK-OUT COMPLETE AND READY FOR S-IV AND S-I STEERING.	
	S	2	CLTC	C1FP RBHM	PERFORM GG LOX INJECTOR PURGE TEST.	
	S	3	CLTC		FILL INFLIGHT COOLER SYSTEM.	
	S	4	CLTC		ALTIMETER "ON".	
	S	5	CLTC	CPSO	VERIFY HOLD FIRE CHECKS COMPLETE.	
	S	6	CLTC	CISP	GODDARD SYSTEMS STATUS CHECK.	
	GY1	7			MAKE COMMUNICATION CHECK WITH HANGAR D TM STATION IN GRAY 1 CIRCUIT.	
~25!						
	S	1	CLTC	C1CM	TELEVISION TO "FILAMENT".	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
251						
	S	2	CLTC	CUMP	AZUSA "ON" ,	
	S	3	CLTC	C1FP	G.G. LOX INJECTOR PURGE TO "AUTOMATIC".	
	S	4	CLTC	C1FP	LOX DOME PURGE TO "AUTOMATIC".	
	S	5	7	C1FR	NOTE PURGES ARMED INDICATION.	
	S	6	CLTC	CIPC	TELEMETERS D1, D2 AND D3 "ON".	
	S	7	CLTC	CIPC	VERIFY VCO CALIBRATION OFF AND AUTO.	
	S	8	CLTC	B4TM	S-IV TELEMETER RECORDER "ON".	
	S	9	CLTC	R4VC	VISICORDER "ON",	
	S	10	CLTC	B4TM	PERFORM TM CHANNEL READOUTS (NOTE: VERIFY CHANNELS 21-24 OF TELEMETER D2 HIGH LEVEL PDM READING 100%).	
	S	11		CLTC	PERSONNEL CONNECTED WITH Smiv STEERING COMMANDS SWITCH TO YW2 CIRCUIT.	
		12			S-IV STEERING COMMANDS.	
	YW2	12-1		RVFC	FLIGHT CONTROL SWITCH TO "S-IV".	
	YW2	12-2		RVFC	HYDRAULIC PUMP ENABLE "ON".	
	YW2	12-3	RVFC	R4FC	S-IV HYDRAULIC PUMPS "ON".	
	YW2	12-4	RVFC		INITIATE S-IV STEERING COMMANDS (APS 105)	
	Y W 2	12-5		RVFC	IU RATE GYRO TORQUER COMMANDS.	
	YW2	12-6	RVFC	R4FC	CLOSE S-IV ACCUMULATOR VALVES.	
	YM5	12-7	RVFC	R4FC	VERIFY S-IV HYDRAULIC PUMPS "OFF"	
	YW2	12-8		RVFC	HYDRAULIC PUMPS ENABLE "OFF".	
	YW2	12-9	RVFC		CHECK ACTUATOR ACCUMULATOR PRESSURES AT 2950 PSIA PLUS OR MINUS 150.	
	YW2	12-10		RVFC	STAGE SELECTOR SWITCH TO "S-I".	
	S	13	CLTC		MAKE RP-1 LEVEL ADJUSTMENT PER PROCEDURE LVO-L-1030.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-251						
	S	14	CLTC	C4TC	REPORT COMPLETION OF S-IV LOX BUBBLING CHECK	
9	S	15	CLTC	C4TC	REPORT COMPLETION OF S-IV TERMINAL COUNTDOWN PREPARATIONS.	
	SRO	16	CNTS	ORNG	VERIFY 0.18 AND 1.16 RADARS AWAY FROM PAD.	
	SRO	17	CNTS	ORNG	DO NOT INTERROGATE C-BAND BEACON WITH 1.16 RADAR UNTIL REQUESTED.	
-241						
	S	1	CLTC	C1MP	TELEMETERS F1, F2, F3, P2, S1 AND S2 "ON".	
	S	2	CLTC	CUMP	C-BAND BEACON (RADAR) "ON".	
	S	3	CLTC	C1MP	TELEMETER AUXILIARY EQUIPMENT "ON".	
	S	4	CLTC	CUMP	TELEMETERS F5, F6, S3, P1 AND MINITRACK "ON".	
	S	5	CLTC	CUMP	TELEMETER AUXILIARY EQUIPMENT "ON".	
	S	6	CLTC	C1CM	CYCLE ALL S-I CAMERAS AS REQUIRED.	
	S	7	CLTC	C1MP CUMP	TELEMETER CALIBRATIONS TO "PREFLIGHT".	
	S	8	CLTC	C1MP CUMP	PREFLIGHT CALIBRATIONS TO 50%.	
	S	9	CLTC	OTMD	SPECIAL RECORDER "ON".	
	S	10	CLTC	C1MP CIPC CUMP	VERIFY TAPE RECORDER READY INDICATIONS.	
-21°	S	1	CLTC	BSOC	VERIFY SCAN RATE OF 10 SEC INSERTED INTO PROPELLANT SCAN PROGRAM.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-201						
	6					
	S	1	CLTC	BDOP	ODOP GROUND TRANSMITTER "ON".	
	S	2	CLTC	C1MP CUMP	PREFLIGHT CALIBRATIONS TO 0%.	
	S	3	CLTC	CUMP	MISTRAM "ON".	
	S	4	CLTC	CUMP	ODOP "ON".	
	SRO	5	CNTS	ORNG	INTERROGATE AZUSA AND MISTRAM.	
		6	CNTS	CPSO	VERIFY LES ARMING KEY HAS BEEN RELEASED TO CSTC.	
	S	7	CLTC	BSOC	INITIATE PROPELLANT LOADING SCAN RATE OF 10 SEC.	
-19 1						
	S	1	CLTC	RSEQ	SEQUENCE AND E&I RECORDERS TO "MINUTE SPEED" AND TIME PULSE "ON".	
	S	2	CLTC	RSEQ	VERIFY VOLTAGE RECORDERS TO "FINE RANGE".	
	SRO	3	CNTS	ORNG	INTERROGATE IU C-BAND BEACON WITH 1.16 RADAR.	
	SRO	4	CNTS	ORNG	INTERROGATE SC C-BAND BEACON WITH 0.18 RADAR.	
	S	5	CLTC	CPSO	RELEASE DESTRUCT ENABLE KEY TO CLVN.	
18 9						
	S	1	CLTC	C4DP	MODULE POWER SUPPLY ON AND CHECK FOR 5 PLUS OR MINUS 0.2 VOLTS.	
	S	2	CLTC	C4DP	TEST S-IV EBWIS FOR "READY".	
	S	3	CLTC	C4DP	VERIFY S-IV S&A "SAFE".	
	S	4	CLTC	CSRP	VERIFY S-I AND IU FLIGHT SEQUENCERS	
	S	5	CLTC	CSPC	S-IV EBW EXTERNAL "ON".	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-181						
	S	6	CLTC	C40P	VERIFY ALL READY INDICATIONS "OFF" AND NOT READY INDICATIONS "ON".	
	S	7	CLTC	CSSP	VERIFY MSFC TALKBACK ENABLE "ON".	
-171						
	S	1	CLTC	CLVN	DESTRUCT ENABLE "ON",	
	S	2	CLTC	C4DP	VERIFY SAFETY BUSS "ON" INDICATION.	
	S	3	CLTC	C1MP CUMP	PREFLIGHT CALIBRATION TO 100%.	
	S	4	CLTC	ORNG	BRING UP WATER PUMPS 5 AND 6 TO	
-161			7			
	S	1	CLTC	CUMP	HORIZON SENSOR POWER "ON".	
	S	2	CLTC	CDCG	CHECK VEHICLE GROUND POWER GENERATOR SETTINGS FOR POWER TRANSFER.	
-15						
	SRO	1	CNTS	ORNG	GUIDANCE COMMAND CARRIER "ON".	
	SRO	2	CNTS	ORNG	RANGE SAFETY COMMAND CARRIER "ON",	
	YWS	3		RVFC	S-IV STEERING COMMAND PERSONNEL SWITCH TO S-CIRCUIT.	
	S	4	CLTC	RVFC	VERIFY S-IV STEERING COMMANDS COMPLETE.	
	S	5	CLTC	C1CM	TELEVISION TO "B+".	
	S	6	CLTC	CUMP	GUIDANCE COMMAND RECEIVER "ON".	
	S	7	CLTC	RRPO	RP-1 LINE INERT PER PROCEDURE LVO-L-1030,	
	SC	8	CNTS	CSTC	PROCEED WITH PREPARATIONS FOR SPACECRAFT INTERNAL POWER.	
				,		

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	сомм.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-14730"						
	S	1	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS.	
	S	2	CLTC	RBHM	SANBORN RECORDER TO 5 MM/SEC.	
	S	3	CLTC	C10P	FUNCTION SELECTOR TO FIRING UNIT VOLTAGE CHECK.	
	S	4	CLTC	C1DP	S-I CDR #1 AND #2 TO "INTERNAL".	
	S	5	CLTC	C1MP CUMP	TELEMETER CALIBRATIONS TO "INFLIGHT".	
	S	6	CLTC	C4DP	S-IV CDR #1 AND #2 "ON".	
	S	7	CLTC	C4DP	VERIFY S-IV EBW #1 AND #2 "OFF" AND "NOT READY".	
	S	8	CLTC	C4DP	S-IV CDR #1 AND #2 TO "INTERNAL".	
-149						
	S	1	CLTC	CIPC	GIVE CAL START.	
	S	2	CLTC	CUNP	VERIFY IU READY FOR POWER TRANSFER "ON"	
	SRO	3	CNTS	ORNG	CUTOFF COMMAND.	
	S	4	CLTC	C1DP C4DP	VERIFY EBW #1 AND #2 "INTERNAL" AND "CHARGED".	
	S	5		B4TM RFSR	NOTE ALL ENGINE CUTOFF NOT RECEIVED.	
		6		RFSR	NOTE HELIUM HEATER CUTOFF NOT RECEIVED.	
	SRO	7	CNTS	ORNG	RELEASE CUTOFF COMMAND.	
	S	8	CLTG	C1DP C4DP	S=I AND S=IV EBW #1 AND #2 TO "EXTERNAL" a	
,	S	9	CLTC	C1DP C4DP	VERIFY S-I AND S-IV EBW'S #1 AND #2	
	S	10	CLTC	CUNP	HORIZON SENSOR "ON".	

- 1. C1NP: Give cutoff (Caution: Hold ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-14 9						
	S	11	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS "ON" FOR 5 SECONDS.	
	S	12	CLTC	CUMP	P1 CALIBRATE COMMAND "ON" AND "OFF".	
-131						
	S	1	CLTC	ORNG	MONITOR ALL RF SYSTEMS AND REPORT ANY CHANGE ON INTERNAL POWER.	
	S	2	CLTC	RVFC	HYDRAULIC PUMP ENABLE "ON",	
	S	3	CLTC	RVFC	S-I HYDRAULIC PUMPS "ON".	
	S	4		CLTC	PERSONNEL CONNECTED WITH S-IV POWER TRANSFER SWITCH TO S-IV CIRCUIT.	
	SIV	5	C4TC	CIPC	INSTRUMENTATION SYSTEMS TO "INTERNAL" .	
	SIV	5-1	C4TC	CSPC	STAGE POWER SYSTEM TO "INTERNAL".	
	SIV	5≂2	C4TC	CSPC	READOUT BATTERIES.	
	SIA	5-3	C4TC	CIPC	PERFORM AUTOMATIC VCO CALIBRATION.	
	SIA	5-4	CATC	CSPC CIPC B4TM	VERIFY SATISFACTORY ON INTERNAL POWER.	
	SIV	5=5	C4TC	CSPC	STAGE POWER SYSTEM TO "EXTERNAL".	
	SIA	5-6	C4TC	CIPC	INSTRUMENTATION SYSTEMS TO "EXTERNAL" .	
	SIV	5-7	C4TC	C4DP	CDR S #1 AND #2 TO "EXTERNAL".	
	S	6	CLTC	RGCC	INITIATE S-I STEERING COMMANDS (APS-104).	
	S	7	CLTC	C1CM	CAMERA LIGHTS "ON" (HOLD ON).	
	S	8	CLTC	CINP	ON MARK S-I STAGE POWER TEST #2 "ON".	

- 1. C1NP: Give cutoff (Caution: Hold ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-131						
	S	9		CSRP	NOTE S-I POWER TRANSFER OK.	
	S	10	CLTC	C1CM	CAMERA LIGHTS "OFF".	
	S	11	CLTC	CUNP	ON MARK IU STAGE POWER TEST #2 "ON" 0	
	S	12		CSRP	NOTE IU POWER TRANSFER OK.	
	S	13	CLTC	CSRP	VERIFY SATISFACTORY S-I AND IU POWER TRANSFER.	
	S	14		C1NP	S=I STAGE POWER TEST #2 "OFF".	
	S	15		CUNP	IU STAGE POWER TEST #2 "OFF" .	
	SIV	16		C4TC	S=IV POWER TRANSFER PERSONNEL SWITCH TO S-CIRCUIT.	
	S	17	CLTC	C4TC	VERIFY SATISFACTORY S-IV POWER TRANSFER	
	S	18	CLTC	CUNP	HORIZON SENSOR "OFF".	
	S	19	CLTC	CPSC	VERIFY TEMPERATURES OK FOR LOX BUBBLING.	
-12 130"						
	SC	1	CNTS	CSTC	BEGIN TRANSFER OF SPACECRAFT TO "INTERNAL POWER".	
-11'						
	S	1	CLTC	RFDM	ARM FIRE DETECTION AND RECORDERS TO SLOW SPEED.	
	S	2	CLTC	RCSM	CSM POWER "ON".	
	S	3	CLTC	RCSM	CSM ACTIVATE "ON" .	
	S	4		C1FR	NOTE CSM READY "ON".	
	S	5		CLTC	PERSONNEL CONNECTED WITH TELEMETER PREFLIGHT CALIBRATION SWITCH TO GRAY 1 CIRCUIT.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

SATURN/APOLLO

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
10!30"						
		1			TELEMETER PREFLIGHT CALIBRATION.	
	GY1	1~1	C1TC	B1TM OTMD	TELEMETER RECORDINGS "ON",	
	SRO	1-2	CNTS	ORNG	TELEMETER RECORDINGS "ON".	
	GY1	1-3	C1TC	C1MP CUMP	TELEMETER CALIBRATIONS TO AC COMMAND.	
	GY1	1-4	C1TC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS "ON" AND 10 SECONDS BEFORE NEXT STEP).	
	GY1	1-5	CITC	CIPC	VCO CALIBRATION TO "MANUAL".	
	GY1	1-6	C1TC	C1MP CUMP	TELEMETER CALIBRATIONS TO "PREFLIGHT".	
	GY1	1-7	C1TC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS.	
	GY1	1-8	C1TC	C1MP CUMP	PREFLIGHT CALIBRATIONS TO 0%, 25%, 50%, 75%, AND 100% IN 2 SECOND INCREMENTS.	
	GY1	1-9	C1TC	CIPC	STEP THROUGH MANUAL CALIBRATION IN 2 SECOND INCREMENTS.	
	GY1	1-10	C1TC	C1MP CUMP	TELEMETER CALIBRATIONS TO "INFLIGHT".	
	GY1	1-11	C1TC	C1MP	RECORDER TRANSFERS "ON".	
	GY1	1=12	CITC		TELEMETER CALIBRATION COMMANDS "ON" FOR 5 SECONDS.	
	GY1	1-13	C1TC		P-1 TELEMETER CALIBRATION COMMAND "ON" AND "OFF".	
	GY1	1-14	C1TC	CIPC	VCO CALIBRATION TO "AUTOMATIC",	
	GY1	1-15	C1TC	CIPC	GIVE CAL START.	
	GY1	1-16	C1TC	RBHM	S-I AND IU HIGH CAL COMMAND FOR 15 SEC.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-10/30"						
	GY1	1-17	C1TC	RBHM	S-I AND IU LOW CAL COMMAND FOR 15 SEC.	
	GY1	1-18	C1TC	RBHM	S-I AND IU RUN COMMAND.	
	GY1	1-19	C1TC	C1MP CUMP	RECORDER TRANSFER "OFF".	
	GY1	1-20	C1TC	B1TM OTMD B4TM	TELEMETER RECORDING "OFF" (AND RELOAD IF REQUIRED).	
	SRO	1-21	CNTS	ORNG	TELEMETER RECORDING "OFF",	
	GY1	1-22		C1TC	TELEMETER CALIBRATION PERSONNEL RETURN TO S CIRCUIT.	
	S	2		C1TC	TELEMETER PREFLIGHT CALIBRATION COMPLETE.	
	S	3	CLTC	CPPC	VERIFY ALL HELIUM SPHERE PRESSURES NORMAL AND SUFFICIENT COLD HELIUM MASS ABOARD.	
-10'10"						
	S	1	CLTC	CPSC	LOX BUBBLING TO "MANIFOLD COOLDOWN".	
-10 '						
	S	1	CLTC	RVFC	S-I HYDRAULIC PUMPS "OFF".	
	s	2	CLTC	RVFC	HYDRAULIC PUMPS ENABLE "OFF".	
	S	3	CLTC		VERIFY STEERING COMPLETE AND READY TO LAUNCH.	
-8'10"						
	S	1	CLTC	CPSÇ	LOX BUBBLING TO "BUBBLE SUPPLY".	
-8 0						
	s	1	CLTC	CISP	GODDARD SYSTEMS STATUS CHECK.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-81						
	S	2	CLTC	RUAC	CONDUCT SWING ARM HYDRAULIC TEST PER LV0=E-7016.	
	S	3	CLTC	RVFC	REMOVE Q-BALL COVER.	
-713011						
	SRO	1	CNTS	ORNG	TELEMETER RECORDING "ON".	
	sc	2		CSTC	PROCEDE WITH SPACECRAFT R AND Z CALIBRATIONS.	
=7º						
	S	1	CLTC	CPSO	PERMISSION TO ARM IGNITION "ON".	
		2	CLTC	C1LS	LAUNCH SEQUENCER ARM "ON".	
		3	CLTC	C1LS	TIME PULSE "ON".	
		4		C1FR	NOTE LAUNCH SEQUENCER READY.	
		5	CLTC	C4DP	VERIFY S-IV EBW'S #1 AND #2 "OFF" AND "NOT READY".	
		6	CLTC	RRPO	VERIFY RP-1 LEVEL ADJUST & LINE INERT COMPLETE.	
	SC	7		CSTC	VERIFY SPACECRAFT SATISFACTORY POWER TRANSFER.	
	S	8	CNTS	CSTC	PROCEDE WITH LES ARM SEQUENCE.	
-6'30"						
	SRO	1	CNTS	ORNG	TELEMETER RECORDING "OFF".	
6 ¹				-		
	S	1	CLTC	CUNP	ARM IU SAFETY SWITCHES.	
	S	2	CLTC	C1NP	ARM S-I SAFETY SWITCHES.	
	S	3	CLTC	CGCM	CAMERA START TO "ARM".	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

			STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-6'						
	S	4	CLTC	C4DP	S-IV CDR #1 AND #2 TO "INTERNAL".	
	S	5	CLTC	C4DP	S-IV CDR #1 AND #2 SWITCHES "OFF" (VERIFY CDR'S REMAIN "ON").	
-51						
	S	1	CLTC	CPER	VERIFY Q-BALL COVER REMOVED.	
	S	2	CLTC	C1FR	VERIFY RANGE SAFE.	
	S	3	CLTC	RGCC	VERIFY FLIGHT COMPUTER SYSTEM IN READY TO LAUNCH MODE.	
	S	4	CLTC	C1NP	FUNCTION SELECTOR TO "LAUNCH".	
	S	5	CLTC	CINP	IGNITION ARMING "ON".	
	S	6		C1FR	NOTE IGNITION SEQUENCER READY.	
	S	7	CLTC	C4TC	VERIFY ALL STATUS SELECTOR SWITCHES TO "READY TO FIRE".	
	S	8	CLTC	RFSR	SEQUENCE RECORDERS TO 10 MM/SEC.	
-4 30 "						
	sc	1		CSTC	SPACECRAFT TO "READY".	
	S	2	CLTC	RVFC	HYDRAULIC PUMPS ENABLE "ON".	
	S	3	CLTC	RVFC	S-I HYDRAULIC PUMPS "ON".	2
-41						
	S	1	CLTC	CUNC	SWITCH TO "INFLIGHT COOLING".	
	SRO	2	CNTS	ORNG	CLEARANCE FOR LAUNCH.	
	s	3	CLTC	C1CM	ZERO THE CAMERA TIMER.	
	S	4	CLTC	RVFC	RATE GYRO SIMULATED COMMANDS,	
	S	5	CLTC	RPSQ	SEQUENCE RECORDERS TO "MINUTE SPEED".	

- 1. C1NP: Give cutoff (Caution: Hold ON)
- 2. C1FP: Prevalves 1-8 OFF
- 3. C1NP: Release cutoff
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-3 ° 40 "						
	S	1	CLTC	B4TM	S-IV TELEMETER RECORDING "ON".	
	S	2	CLTC	C1DP C4DP	ARM DESTRUCT S&A.	
	S	3		C1FR	NOTE S-IV SYSTEMS READY INDICATION "ON"	
	S	4	-	C1FR	NOTE SC SYSTEMS READY INDICATION "ON".	
		5	CLTC	BSOC	CDC INHIBIT "OFF".	
		6	CLTC	BSOC	DISCRETE INHIBIT "OFF".	
-3'10"						
	S	1	CLTC	CPSC	LOX BUBBLING "OFF".	
-3 0						
	S	1	CLTC	BSDC	ENABLE GUIDANCE RELEASE.	
-2153"						
	0		01 20	Dagg	VEDIEW OHIDANGE DELEGOE	
	S	1	CLTC		VERIFY GUIDANCE RELEASE.	
	S S	3	CLTC		NOTE "IU READY". VERIFY "PREP COMPLETE".	
-2145"	3	3	0610	CIPK	VERIFY "FREE COMPLETE".	
2 45"						
	SRO	1	CNTS	ORNG	TELEMETER RECORDING "ON".	
	S	2	CLTC	OTMD B1TM	TELEMETER RECORDINGS "ON".	
-213811						2
	S	1	CLTC	RSEQ	SEAHENCE RECORDEDS TO MEAST SDEED!	
	3	Τ.	U I U	NJEW	SEQUENCE RECORDERS TO "FAST SPEED".	

EMERGENCY PROCEDURE (T-2'33" to T-3")
Note: Spacecraft has indefinite recycle capability until T-18", limited thereafter

To Interrupt Automatic Launch Sequence

- 1. C1FR: GIVE CUTOFF
- 2. C1NP: Ignition arming OFF
- 3. CPSO: IGNITION PERMISSION OFF
- L. CPSO: HOLD FIRE ON
- 5. RSEQ: RECORDERS TO SLOW SPEED
- 6. RWCP: Operate water control system as req!d
- 7. RVFC Hydraulic pumps 1, 2, 3, and 4 OFF
- 8. RBHM: VERIFY FUEL AND LOX TANKS VENTED
- 9. RPLC: VERIFY LH2 AND LOX TANKS VENTED
- 10. C1NP: Function selector to prelaunch
- 11. C1LS: Launch sequencer arm OFF
- 12. C1DP, C4DP: SAFE DESTRUCT
- 13. C1NP, CUNP: SAFETY SWITCH ARMING OFF
- 14. C1NP: Reset cutoff
- 15. C1NP: RESET IGNITION SEQUENCER
- 16. CNTS: Secure all other equip. As Req'd and MAKE ANNOUNCEMENT FOR ALL PERSONNEL TO STAND BY UNTIL DECISION IS MADE TO RECYCLE OR SCRUB

PREMATURE CLOSURE OF PREVALVES

- 1. Note: The prevalves may remain closed FOR 10 MINUTES DURING WHICH THE LOX TANKS MUST BE PRESSURIZED, PREVALVES OPENED AND LOX BUBBLING INITIATED. THE LOX TANKS SHOULD THEN BE VENTED AND BUBBLING DIS-CONTINUED WHEN THE PUMP INLET TEMPERATURES REACH -275°F.
- Note: Upon exceeding 10 minutes the pre-VALVES MUST REMAIN CLOSED, LOX DRAINED AND STANDBY OBSERVED UNTIL ALL SUCTION LINE FROST IS MELTED. THE PUMP INLET PRESSURES MUST BE MONITORED DURING THIS PERIOD AND IF THE PRESSURE EXCEEDS 200 PSIG (FULL CHART SCALE) THE PREVALVES MUST BE OPENED.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
~2 1 33 11						
	S	1	CLTC	C1FR	LAUNCH SEQUENCE START.	
	S	2		C1FR	MONITOR FUEL PRESSURIZING.	
	S	3		C1FR	MONITOR LOX BUBBLING.	
	S	4		CSSP	MONITOR S-IV STAGE POWER AND INSTRUMENTATION POWER SYSTEMS INTERNAL.	
-2128#		2				
	S	1		C1FR	MONITOR S-I FUEL TANKS PRESSURIZED.	
	S	2		CSSP	MONITOR LO2 & LH2 SYSTEMS READY INDICATION.	
-2'23"						
	S	1		RSEQ	SEQUENCE RECORDERS TO SLOW SPEED.	
2 1 3 11						
	S	1		CSSP	REPORT S-IV READY FOR LAUNCH WHEN INDICATION APPEARS AND REMAINS FOR AT LEAST 5 SECONDS.	
	S	2		RPLC	S-IV UMBILICAL LINE VENT AND PURGE SWITCHES TO OPEN (AFTER S-IV READY FOR LAUNCH).	
-1'53"						
	S	1			MONITOR S=IV LOX SYSTEM CONTROL TO LOCAL (AFTER LOX SYSTEM READY).	
-1'48"						
1 9 4 3 #	S	1	9	RSEQ	SEQUENCE RECORDERS TO FAST SPEED.	
	S	1		C1FR	MONITOR S=I LOX BUBBLING "OFF".	

EMERGENCY PROCEDURE (T-2'33" to T-3")
Note: Spacecraft has indefinite recycle capability until T-18", limited thereafter

To Interrupt Automatic Launch Sequence

- 1. C1FR: GIVE CUTOFF
- 2. C1NP: Ignition arming OFF
- 3. CPSO: IGNITION PERMISSION OFF
- L. CPSO: HOLD FIRE ON
- 5. RSEQ: RECORDERS TO SLOW SPEED
- 6. RWCP: Operate water control system as req!d
- 7. RVFC Hydraulic pumps 1, 2, 3, and 4 OFF
- 8. RBHM: VERIFY FUEL AND LOX TANKS VENTED
- 9. RPLC: VERIFY LH2 AND LOX TANKS VENTED
- 10. C1NP: Function selector to prelaunch
- 11. C1LS: Launch sequencer arm OFF
- 12. C1DP, C4DP: SAFE DESTRUCT
- 13. C1NP, CUNP: SAFETY SWITCH ARMING OFF
- 14. C1NP: Reset cutoff
- 15. C1NP: RESET IGNITION SEQUENCER
- 16. CNTS: Secure all other equip. As Req'd and MAKE ANNOUNCEMENT FOR ALL PERSONNEL TO STAND BY UNTIL DECISION IS MADE TO RECYCLE OR SCRUB

PREMATURE CLOSURE OF PREVALVES

- 1. Note: The prevalves may remain closed FOR 10 MINUTES DURING WHICH THE LOX TANKS MUST BE PRESSURIZED, PREVALVES OPENED AND LOX BUBBLING INITIATED. THE LOX TANKS SHOULD THEN BE VENTED AND BUBBLING DIS-CONTINUED WHEN THE PUMP INLET TEMPERATURES REACH -275°F.
- Note: Upon exceeding 10 minutes the pre-VALVES MUST REMAIN CLOSED, LOX DRAINED AND STANDBY OBSERVED UNTIL ALL SUCTION LINE FROST IS MELTED. THE PUMP INLET PRESSURES MUST BE MONITORED DURING THIS PERIOD AND IF THE PRESSURE EXCEEDS 200 PSIG (FULL CHART SCALE) THE PREVALVES MUST BE OPENED.

SATURN/APOLLO

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-1 43"						
	S	2	z - 2	C1FR	MONITOR S-I LOX VENT AND RELIEF VALVES "CLOSE".	
	S	3		C1FR	MONITOR S-I LOX PRESSURIZING.	
	S	4		RBHM	REPORT S=I LOX TANK PRESSURES DURING LOX TANK PRESSURIZATION.	
-1º0"						
	S	1		RCSM	RECORDERS "ON".	
	S	2		RFDM	RECORDERS TO "FAST SPEED".	
	S	3		RBHM	OSCILLOGRAPH "ON".	
~38 [†]						2
	S	1		RWCP	MONITOR PAD FLUSH OPENING.	
	S	2	i.	OTMD, B1TM	8 KC OSCILLATOR "ON".	
		3		ORNG	WATER PUMPS 5 AND 6 TO FULL SPEED.	
-28#						
	S	1		CINP	VERIFY S-I POWER TRANSFER.	
	S	2		CUNP	VERIFY IU POWER TRANSFER.	
	S	3		C1FR	MONITOR TO FUEL INJECTOR PURGE "ON".	
	S	4		C1FR	MONITOR GG LOX INJECTOR PURGE "ON".	
	S	5		C1FR	MONITOR S-I LOX DOME PURGE "ON".	
	S	6		C1FR	MONITOR S-I LOX TANKS PRESSURIZED.	
-18"						
	S	1		CSRP	MONITOR READY FOR IGNITION "ON",	
	S	2		CSRP	MONITOR SC UMBILICAL ARM RELEASE.	

EMERGENCY PROCEDURE (T-2'33" to T-3")
Note: Spacecraft has indefinite recycle capability until T-18", limited thereafter

To Interrupt Automatic Launch Sequence

- 1. C1FR: GIVE CUTOFF
- 2. C1NP: Ignition arming OFF
- 3. CPSO: IGNITION PERMISSION OFF
- L. CPSO: HOLD FIRE ON
- 5. RSEQ: RECORDERS TO SLOW SPEED
- 6. RWCP: Operate water control system as req!d
- 7. RVFC Hydraulic pumps 1, 2, 3, and 4 OFF
- 8. RBHM: VERIFY FUEL AND LOX TANKS VENTED
- 9. RPLC: VERIFY LH2 AND LOX TANKS VENTED
- 10. C1NP: Function selector to prelaunch
- 11. C1LS: Launch sequencer arm OFF
- 12. C1DP, C4DP: SAFE DESTRUCT
- 13. C1NP, CUNP: SAFETY SWITCH ARMING OFF
- 14. C1NP: Reset cutoff
- 15. C1NP: RESET IGNITION SEQUENCER
- 16. CNTS: Secure all other equip. As Req'd and MAKE ANNOUNCEMENT FOR ALL PERSONNEL TO STAND BY UNTIL DECISION IS MADE TO RECYCLE OR SCRUB

PREMATURE CLOSURE OF PREVALVES

- 1. Note: The prevalves may remain closed FOR 10 MINUTES DURING WHICH THE LOX TANKS MUST BE PRESSURIZED, PREVALVES OPENED AND LOX BUBBLING INITIATED. THE LOX TANKS SHOULD THEN BE VENTED AND BUBBLING DIS-CONTINUED WHEN THE PUMP INLET TEMPERATURES REACH -275°F.
- Note: Upon exceeding 10 minutes the pre-VALVES MUST REMAIN CLOSED, LOX DRAINED AND STANDBY OBSERVED UNTIL ALL SUCTION LINE FROST IS MELTED. THE PUMP INLET PRESSURES MUST BE MONITORED DURING THIS PERIOD AND IF THE PRESSURE EXCEEDS 200 PSIG (FULL CHART SCALE) THE PREVALVES MUST BE OPENED.

LAUNCH COUNTDOWN
DATE SEPTEMBER 5, 1964
REVISION

SATURN/APOLLO LAUNCH OPERATIONS

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
15"						
	S	1	2	RCSM	CSM RECORDERS TO "FAST SPEED".	
10"						
	S	1.		CSRP	MONITOR SC UMBILICAL ARM RETRACTED.	
8"						
	S	1		CSRP	MONITOR SHORT MAST 2 AND 4 PRESSURIZING VALVES #1 "OPEN".	
		9				
				8		
	2					

EMERGENCY PROCEDURE (T-3" TO T-0")

AUTOMATIC CUTOFF

- 1. C1FR: OBSERVE AUTOMATIC CUTOFF
- 2. C1FP: Verify deluge purge ON if Not ON, operate deluge purge to Manual
- 3. RSEQ: Sequence recorders to slow speed
- 4. RWCP: Operate water control system as required
- 5. RVFC: Hydraulic pumps 1, 2, 3, and 4 OFF
- 6. C1NP: Function selector to prelaunch
- 7. CGCM: CAMERA ARMING OFF
- 8. C1LS: Launch sequencer arm OFF
- 9. C1DP, C4DP: SAFE DESTRUCT
- 10. RBHM: VERIFY FUEL AND LOX TANKS VENTED TO ZERO PSI
- 11. RPLC: VERIFY LH2 AND LOX TANKS VENTED TO ZERO PSI
- 12. C1NP, CUNP: SAFETY SWITCH ARMING OFF
- 13. The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks

- SHOULD THEN BE VENTED AND BUBBLING DISCONTINUED WHEN THE PUMP INLET TEMPERATURES REACH -275° F.
- 14. Upon exceeding 10 minutes, the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.
- 15. <u>Caution</u>: Prior to resetting cutoff, verify the following:
 - A. C1FP: ALL PREVALVE SWITCHES TO OFF POSITION
 - B. C1FP: Lox dome purge ON (REQUIRED FOR 15 SECONDS MINIMUM AFTER CUTOFF)
 - c. C1FP: GG Lox injector purge ON (required for 10 minute minimum after cutoff)
- 16. C1NP: IGNITION ARMING OFF
- 17. CNTS: Secure ALL OTHER EQUIPMENT AS REQUIRED

SATURN/APOLLO

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
= 3 ¹¹						
=2,98"	S	1		C1FR	VERIFY IGNITION COMMAND.	
-2.88"	S	1		CSRP	MONITOR IGNITION OF ENGINES #5 AND #7.	
_2 79#	S	1		CSRP	MONITOR IGNITION OF ENGINES #6 AND #8.	
~2.78¶						
-2.68#	S	1		CSRP	MONITOR IGNITION OF ENGINES #2 AND #4.	
	S	1		CSRP	MONITOR IGNITION OF ENGINES #1 AND #3.	
	S	2		CSRP	ALL ENGINES RUNNING "ON".	
	S	3		C1FR	TC FUEL INJECTOR PURGE "OFF".	

EMERGENCY PROCEDURE (T-0" TO T+0")

<u>AUTOMATIC CUTOFF</u>

1.	C1FR:	Observe automatic cutoff	13.	CINP,	CUNP: SAFETY SWITCH ARMING OFF
2.	CIFP:	ALL PREVALVE SWITCHES TO OFF POSITION	14.	RBHM:	Verify S-I fuel and Lox tanks vented to zero PSI
3.	CIFP:	VERIFY DELUGE PURGE ON - IF NOT ON, OPERATE DELUGE PURGE TO MANUAL	15.	RPLC:	VERIFY S-IV LH2 AND LOX TANKS VENTED TO ZERO PSI
4.	RWCP:	Operate water control system as required	16.	CILS:	Launch sequencer arm OFF
5.	CIMP:	RECORDER TRANSFER SWITCH TO EMERGENCY POSITION	17.		The prevalves may remain closed
6.	RVFC:	Hydraulic pumps 1, 2, 3 and 4 OFF			FOR 10 MINUTES DURING WHICH THE LOX TANKS MUST BE PRESSURIZED,
7.	CINP:	Function selector to prelaunch			PREVALVES OPENED AND LOX BUBBLING INITIATED. THE LOX TANKS SHOULD
8.	CINP:	Verify prevalve switches OFF then reset cutoff			THEN BE VENTED AND BUBBLING DIS- CONTINUED WHEN THE PUMP INLET TEMPERATURES REACH -275° F.
9.	CIFP:	GG Lox injector purge to manual	18.	NOTE:	Upon exceeding 10 minutes the pre- valves must remain closed, Lox

VALVES MUST REMAIN CLOSED, LOX
DRAINED AND STANDBY OBSERVED UNTIL
ALL SUCTION LINE FROST IS MELTED.
THE PUMP INLET PRESSURES MUST BE
MONITORED DURING THIS PERIOD AND IF
THE PRESSURE EXCEEDS 200 PSIG (FULL
CHART SCALE) THE PREVALVES MUST BE

OPENED.

19. CNTS: Secure ALL OTHER EQUIPMENT AS REQUIRED.

10. CIFP: Lox dome purge to manual

12. CIDP, CADP: SAFE DESTRUCT

11. RSEQ: Sequence recorders to slow speed

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-0 0 0 11						
	S	1		CSRP GSSP	MONITOR COMMIT.	
	S	2		CSRP	HOLDDOWNS RELEASE.	
	S	3		CSRP	S-I LOX AND FUEL MASTS EJECT VALVES	
+0 0 0 **						
	S	1		CSRP	VERIFY LIFTOFF.	
	S	2		CSRP	SHORT MAST 2 AND 4 RELEASE.	
	S	3		CSRP	UMBILICAL ARMS 1, 2 AND 3 RELEASE AND RETRACT.	
	S	4		CVMP	"CLOSE" VACUUM SUPPLY VALVES.	
	S	5		CVMP	"OPEN" PURGE SUPPLY.	
	S	6		RECS	PERFORM ECS POST LAUNCH OPERATIONS PER PROCEDURE LVO=E-7018.	
	SRO	7		ORNG	START GUIDANCE COMMAND TRANSMISSION.	
+5"						
	S	1		RWCP	TORUS RING WATER VALVES "OPEN".	
	S	2		RWCP	PAD FLUSH VALVE FAST "CLOSES".	
+ 6 *!						
	S	1		CPSP	LAUNCHER POWER "OFF".	
+8"						
	S	1		RSEQ	SEQUENCE RECORDERS TO "SLOW SPEED".	
+10 **						
	S	1		RCSM	RECORDERS TO "SLOW SPEED".	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
+10"						
	S	2		RBHM	SANBORN RECORDER "OFF".	
÷30 ¹¹						
		1			511 CAMERA OPERATION.	
	S	2		OCCF	SHUT OFF ECS LOW PRESSURE GN2 SUPPLY.	
	S	3	8	BSOC	TEST END SENSE SWITCH "ON".	
+1 0						
	SIV	1		CSSP	RESET TIMER.	
	SIV	2		RFSR	SEQUENCE RECORDER TO 2 MM/SEC.	
	SIV	3	2	CPPC	CLOSE CONTROL HELIUM SUPPLY.	
	SIV	4		CPSC	DECREASE COLD HELIUM SUPPLY DOME PRESSURE TO 0 PSIG.	
	SIV	5		CPSC	SWITCH COLD HELIUM SUPPLY MOMENTARILY TO VENT, THEN SUPPLY.	
	SIV	6		RPLC	SWITCH S-IV LO2 CONTROL TO "MANUAL".	
	SIV	7		CPSC	CLOSE LO2 NOZZLE PURGE.	
	SIV	8		CPSC	CLOSE LH2 NOZZLE PURGE.	
	SIA	9		RPLC	CHECK OR POSITION THE FOLLOWING SWITCHES:	
					A. LO2 UMBILICAL LINE PURGE "OPEN"	
					B. LO2 UMBILICAL LINE VFNT "OPEN"	
					C. LH2 UMBILICAL LINE PURGE "OPEN"	
					D. LH2 UMBILICAL LINE VENT "OPEN"	
			,		E. LH2 TRANSFER (MOMENTARILY) "SECURE"	
					F. LH2 TRANSFER "MANUAL"	
					G. LH2 REPLENISH "OPEN"	
					H. LH2 MAIN FILL "OPEN"	

PAGE TEST NO. VEHICLE

REVISION					VEHICLE	SA-
TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
+1 †						
	SIV	10		RPLC	SWITCH COMPUTER READOUT LOCK "ON".	
	SIV	11		RPLC	SWITCH LOX AND LH2 COMPUTERS "OFF".	
	SIA	12		C4DP C4OP CSSP	NORMALIZE ALL SWITCHES,	
	SIV	13		C4TC	COMPLETE S=IV SECURING PER PROCEDURE DAC-P-7011.	
+1' 47.8"						
		1			SEPARATION CAMERA OPERATION.	
		2			LH2 PRESTART.	
+2 ! 14 . 9 "						
		1			S-I STAGE LEVEL SENSORS ARMED.	
+2 ¹						
		1		7	FIRE VENT PORTS.	
-2'21"						
		1			S-I INBOARD ENGINE CUTOFF (NOMINAL).	
2126"						
26,9"						
		1			S-I OUTBOARD ENGINE CUTOFF (NOMINAL).	
27.6"						
		1			S-IV ULLAGE ROCKET IGNITION (NOMINAL).	
,		2			SEPARATION (NOMINAL).	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
+2' 27,72"						
2/9/2		1			S=I RETROROCKET IGNITION (NOMINAL).	
+2° 29.4°						
		1			S-IV ENGINE START (NOMINAL).	
+3°0"						
	RD4	1		RLHC	INERT LH2 SYSTEM PER LVO-L-1006.	
+600"						
	RD3	1			SECURE WATER SYSTEM PER PROCEDURE LVO-E-7014.	
÷7°30"						
		1		ORNG	AZUSA SWITCHOVER.	
+9 ¹ 43 . 5 ¹¹						
-		1			S-IV CUTOFF ARM (NOMINAL).	
+10 119"						
		1			S-IV CUTOFF (NOMINAL).	
		1			ACCORDING TO PREARRANGED CRITERIA AND TIME SEND "SAFE" COMMAND.	
		2			PERFORM ORBITAL LOAD ACCORDING TO INFORMATION IN OD 2460.	
			2		NOTE	
					UPON RETURN TO PAD.	

LAUNCH COUNTDOWN
DATE SEPTEMBER 5, 1964
REVISION

SATURN/APOLLO LAUNCH OPERATIONS

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARK
		1		PPS0	KEEP LAUNCHER CLEAR OF ALL NON DESIGNATED PERSONNEL UNTIL SGMP VERIFIES ITEMS FOLLOWING ARE COMPLETE.	
		1-1		SGMP	INSTALL LAUNCHER SAFETY NET.	
		1-2		SGMP	OPEN LAUNCHER ACCESS DOOR AND VERIFY LAUNCHER INTERIOR CLEAR OF GN2.	
		1-3		SGMP	SAFETY WIRE HOLDDOWN ARM RELEASE PANEL SUPPLY VALVE (4444) "CLOSED" AND VENT VALVE (4442) "OPEN".	
		1-4		SGMP	VERIFY APCD HAS SECURED ALL LAUNCHER PNEUMATICS.	
		1=5			PERFORM POST LAUNCH PNEUMATIC OPERATIONS PER PROCEDURE LVO-E-7022.	
		2			PERFORM USA POST LAUNCH OPERATIONS PER PROCEDURE LVO-E-7023.	