

### VOLUME II SATURN I COUNTDOWN MANUAL

JOHN F. KENNEDY SPACE CENTER, NASA

HTTP://HEROICRELICS.ORG

SA-7

### SATURN I COUNTDOWN MANUAL

(VOL. 11)

SA-7

TEST NUMBER

7-LSVI-300

LAUNCH COUNTDOWN

SEPTEMBER 5, 1964

### SATURN I COUNTDOWN MANUAL

(VOL. | & VOL. ||)

PREPARED BY:

TOST SUPERVISOR

APPROVED BY:

LAUNCH VEHICLE TEST CONDUCTOR

Souglas Operations

CHRYSLER OPERATIONS

APPROVED BY:

SPACECRAFT TEST CONDUCTOR

NORTH AMERICAN OPERATIONS

APOLLO SATURN 1/18 TEST AND SYSTEMS ENGINEERING OFFICE

APPROVED BY:

DIRECTOR, LAUNCH VEHICLE OPERATIONS

LAUNCH VEHICLE OPERATIONS

APPROVED BY:

CECRAFT OPERATIONS

### INDEX

In	11	
VO		

	INTRODUC	TION	24 (794) 1																1
	TV CAMER	A LOCAT	LONS			•	•	•	•	•	•	•	•	•	•	•	•	•	
	COMPLEX	37 COMM	INIC	AT 10	NS.	•	•	*		*		•		•					2346
	NORCOM C	HANNEL	Assi	SUME	NITS	· :	0677	* 5	28 <b>.</b> 07 1585		DAV	30	530k	1 <b>9</b> 0	10.000 12.000	*		*	Si.
	OPERATIN	G STATI	one (	ODE	Inc	·			· A T		IAI		•	•		•	•	•	2
	TEST DES	CRIPTIO	1 (5.	_1 F	HEL	10	AD	10	(6)	10	/14	*	(0 <b>9</b> 13)	*	0.00				11
	S-I FUEL	LOADING	RAI	DCDA	DH	LU	מח	TIN	10 /	ii.		•	55	101			5	•	12
	S-1 RP-1	LOADING	a Ppi	FDAD	AT 10	· ·		•	•	•	•	•	0. <b>1</b> (1)	•		•	•	*	13
	(PART I		3 0000	LEAR	A110	JIVS		•		*	•	•	•	٠	•	•		*	')
	S-I RP-1	LOADIN	G.	• •		٠	2.0	•	•	•	(3 <b>*</b> )	•		*	٠	٠	•	٠	18
	TEST DES		, (C	011N1T	D O W/	. D		^.			-		.1						2)1
																			24
	POWER FA																		25 26
																			20
		11, T-10	0351	TO	T-52	151	)												
	Countdow (Part	n Demon:	5451	T I ON	TES T+39	90 !	;	•	•	•	•	•	٠	٠	•	Š	٠	•	51
VOL.	II (Thi	s Volum	E)																
	TEST DES	CRIPTIO	۷ (L	AUNC	н Сс	NUC	TD	OW	M)	ě		2	٠						125
	TV CAMER	A LOCAT	ION	• •			•	٠		٠	٠	٠		٠	•	*	0.00	.5)	126
	ENGINE C																		
	COMPLEX	<b>37</b> Соммі	JNICA	ATIO	NS	٠		٠				•					•	•	131
	Norcom C	HANNEL /	Assi	SNME	NTS	٠		٠	•	٠	٠		•			٠	•		132
	OPERATIN	G STATI	ON CO	DDE	DEN	ITI	FΙ	CA	TI	01	1								134
	COUNTDOW	N BARGR	APH												5.00			•	139
	LAUNCH P	RECOUNT IV, T-8						٠	٠	٠	٠	٠	٠	٠	٠	*	•	*	140
	Launch C (Part	OUNTDOWN	у. 51 тн	 HROU	 Gн l	.AU	NC	н)	( <b>.</b> )	*	٠	٠		٠	٠	*	٠	*	159

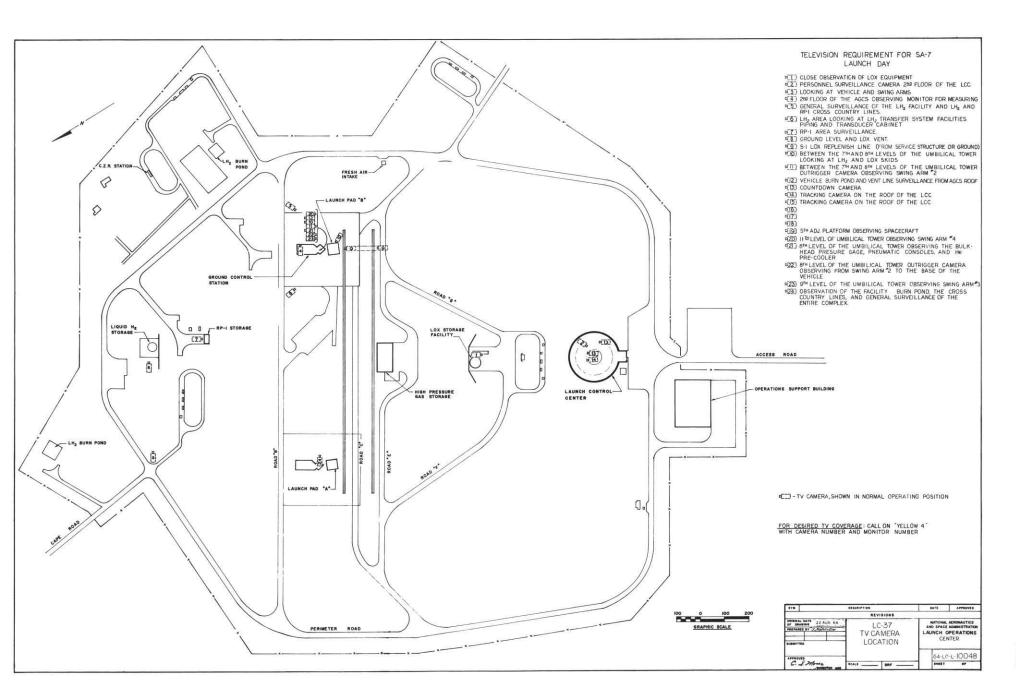
LAUNCH OPERATIONS TES	125 <b>T</b>
1. TITLE:	2. TEST NUMBER
LAUNCH COUNTDOWN	7-LSVI-300
	sA-7
4. OBJECTIVES: TO PROVIDE A SEQUENCE OF OPERATION REQUIRE LAUNCH VEHICLE TO THE POINT OF FIRING COMMAND (INFIRING SEQUENCE); TO PROVIDE EMERGENCY PROCEDURES IN THE EVENT OF STANDARD MALFUNCTIONS; TO PROVIDE OTHER NASA CENTERS WITH A LISTING OF INTERRELATED MENT BEFORE AND AFTER LIFTOFF; AND TO PROVIDE THE THE IMMEDIATE POST-LAUNCH SECURING OPERATIONS.	FOR SAFING OF VEHICLE THE AMR RANGE AND ITEMS FOR ACCOMPLISH-
5. TEST DESCRIPTION:	
1. THE COUNTDOWN INCLUDES OR REFERENCES ALL ACTIVE TO ACCOMPLISH THE LAUNCH OF THE SATURN I VEHICLE. DIVIDED INTO TWO PARTS (SPLIT COUNT) NORMALLY TO ECONSECUTIVE DAYS AFTER THE COMPLETION OF FIRST STA	This operation is be performed on two

- AND ASSOCIATED OPERATIONS.
- 2. EMERGENCY PROCEDURES FOR STANDARD SAFING OPERATIONS ARE INCLUDED AS A PART OF THE COUNTDOWN MANUAL.
- 3. Post-Liftoff operations terminate with payload injection into orbit.

#### 6. STATUS:

1. ALL PRELAUNCH PREPARATIONS AND CHECKOUT SATISFACTORILY COMPLETED.

7. PREPARED BY:	9. NASA SYSTEMS ENGINEER	11, DATE
		SEPTEMBER 4, 1964
8. CONTRACTOR GROUP ENGINEER	10. NASA SYSTEMS SUPERVISOR	12. CONTRACTOR REFERENCE NR.
	Com Stone	



### ENGINE CUTOFF AND EMERGENCY OPERATIONS

### 1. NETWORK PANEL CUTOFF

CUTOFF FROM THE NETWORK PANEL SHOULD BE INITIATED ONLY
IN A SITUATION WHERE A TURBINE SPINNER SQUIB INADVERTENTLY
IGNITES PRIOR TO THE IGNITION COMMAND. THE CUTOFF BUTTON
SHOULD BE DEPRESSED UNTIL PREVALVES 1-8 SWITCHES ARE
TURNED OFF. (FOR DETAILED PROCEDURE SEE "EMERGENCY
PROCEDURE" INCORPORATED IN THE LAUNCH COUNTDOWN.)

### 2. FIRING PANEL CUTOFF

ONCE FIRING COMMAND HAS BEEN INITIATED AT T-2'33" AND THE AUTOMATIC SEQUENCE BEGINS, IT IS NECESSARY TO GIVE CUTOFF FROM THE FIRING PANEL TO STOP THE SEQUENCE AND INTERRUPT THE ASSOCIATED FIRING CIRCUITRY. THIS CUTOFF WILL
NOT FIRE THE CONAX VALVE SQUIBS NOR CLOSE THE PREVALVES
UNLESS IGNITION COMMAND HAS BEEN INITIATED. (FOR
DETAILED PROCEDURE SEE "EMERGENCY PROCEDURE" INCORPORATED
IN THE LAUNCH COUNTDOWN.)

### 3. AUTOMATIC MALFUNCTION CUTOFFS

	Ac	TIVE		
TYPE OF CUTOFF	From	То	DESCRIPTION	
Premature Commit	Power ON	TIME FOR COMMIT	This circuit will detect premature voltage on the +1D14 bus.	
SEQUENCE FAILURE	Time for Ignition	IGNITION COMMAND	THIS CUTOFF WILL COME AT "TIME FOR IGNITION" IF ANY ONE OR MORE OF THE PREREQUISITES IN THE "READY FOR IG- NITION" CHAIN IS NOT MET.	
S-I VOLTAGE FAILURE	IGNITION COMMAND	Сомміт	THIS CIRCUIT DETECTS LOSS OF THE S-I INVERTER BUS.	
S-IV FAILURE	IGNITION COMMAND	Сомміт	This cutoff will be initiated if any one or more of the prerequisites for "S-IV Ready to Launch" are not met by the time S-I ignition command occurs.	
IU FAILURE	IGNITION COMMAND	Сомміт	THIS CUTOFF WILL BE INITIATED IF ANY OF THE PREREQUISITES IN THE "IU READY" CHAIN ARE NOT MET BY THE TIME S-I IGNITION OCCURS.	

	Аст	IVE	
TYPE OF CUTOFF	FROM	То	DESCRIPTION
Fire Detection	IGNITION COMMAND	Сомміт	THIS CIRCUIT WILL INITIATE CUTOFF IF THERMOCOUPLES MOUNTED ON THE HEAT SHIELD DETECT A PREDETERMINED RATE OF HEAT RISE.
ROUGH COMBUSTION	IGNITION COMMAND	Сомміт	THIS CIRCUIT WILL DETECT ROUGH COM- BUSTION OF ANY ENGINE, CUTOFF THIS ENGINE IMME- DIATELY AND INI- TIATE SEQUENCE CUT- OFF TO THE REMAIN- ING ENGINES.
THRUST FAILURE	Time for Commit	Сомміт	THIS CIRCUIT WILL INITIATE CUTOFF IF ANY ONE OR MORE OF THE S-I ENGINES IS NOT RUNNING BY "TIME FOR COMMIT"
Launch Failure	IGNITION COMMAND	LIFTOFF	THIS CIRCUIT WILL INITIATE CUTOFF 8 SECONDS AFTER IGNITION IF THE VEHICLE HAS NOT LIFTED OFF BY THIS TIME. (NOMINAL TIME IS APPROXI- MATELY 3.4")

### 4. PRIMARY (60 CYCLE AC) POWER FAILURES

A MOMENTARY LOSS OF AC POWER (CAPE CRITICAL OR INDUSTRIAL) WILL

CAUSE ALL GENERATORS TO DROP OUT; HOWEVER, BACKUP BATTERIES

(160 AMPERE-HOUR) ACROSS THE GENERATOR OUTPUT LINES PROVIDE

28 VDC POWER FOR MAINTAINING THE VEHICLE IN A SAFE CONDITION.

DURING ANY EXTENDED LOSS OF AC POWER, ANY UNNECESSARY LOADS

WILL BE REMOVED FROM THE BUSES.

PAGE

TEST NO.

VEHICLE

COMPLEX 37 COMMUNICATIONS

INTERCOM: (IC)

TO BE UTILIZED FOR "T" COUNT ANNOUNCEMENTS, OPERATIONAL AND EMERGENCY ANNOUNCEMENTS, AND FOR PAD NORCOM REQUEST ANNOUNCEMENTS.

PAGING: (PA)

TO BE UTILIZED FOR ADMINISTRATIVE PHONE CALL ANNOUNCEMENTS, "T" COUNT ANNOUNCEMENTS, AND PAGING SPECIFIC PERSONNEL TO A HEADSET CHANNEL.

NORCOM: (OIS)

THE TEST AND CHECKOUT OPERATIONAL COMMUNICATIONS
UTILIZED AS ASSIGNED OR INDICATED IN THE
PROCEDURE FOR THE TEST OPERATIONS. TEST
SUPERVISORY PERSONNEL SHOULD ALWAYS BE AVAILABLE
IN THE FOLLOWING CIRCUITS.
S-I CHIEF OPERATIONS ENGINEER (CHRYSLER) SI
S-IV CHIEF OPERATIONS ENGINEER (DOUGLAS) SIV
LAUNCH VEHICLE TEST CONDUCTOR (NASA-KSC) S
SPACECRAFT TEST CONDUCTOR (NASA-MSC) SC
VEHICLE TEST SUPERVISOR S.SRO

#### RANGE PHONE: (SRO)

A SPECIAL COMMUNICATIONS NETWORK IS ESTABLISHED BETWEEN THE TEST SUPERVISOR AND THE EASTERN TEST RANGE SUPERVISOR OF RANGE OPERATIONS. THIS CIRCUIT WILL BE UTILIZED FOR ALL TEST RELATED COMMUNICATIONS BETWEEN THE LAUNCH COMPLEX AND CENTRAL CONTROL. IT WILL BE BUSSED TO THE OIS AS REQUIRED BY THE TEST PROCEDURE OR AT THE DISCRETION OF THE TEST SUPERVISOR.

SA-7

### NORCOM CHANNEL ASSIGNMENTS

CHANNEL		PRIMARY FUNCTION
*RED 1	(RD1)	S-I VEHICLE MECHANICAL
RED 2	(RD2)	S-IV VEHICLE MECHANICAL
*RED 3	(RU3)	ALL LAUNCH VEHICLE GSE (MECHANI
RED 4	(RD4)	PROPELLANTS
YELLOW 1	(YW1)	GYRO AND LAYING
YELLOW 2	(YW2)	LAUNCH VEHICLE CONTROL
YELLOW 3	(YW3)	GUIDANCE AND RCA 110 COMPUTER
#YELLOW 4	(YW4)	PAD SAFETY, TV AND COMMUNICATIO
YELLOW 5	(YW5)	( NOT ASSIGNED)
BROWN 1	(881)	S-I MEASURING AND BH MONITOR CI
BROWN 2	(BR2)	S-IV MEASURING
BROWN 3	(BR3)	FACILITY AND GSE MEASURING
BROWN 4	(RR4)	SERVICE STRUCTURE MEASURING STA
*BLUE 1	(BL1)	NASA GROUND ELECTRICAL NETWORK
BLUE 2	(BF5)	DAC GROUND ELECTRICAL NETWORK
*BLUE 3	(BL3)	S-I, IU VEHICLE NETWORK
BLUE 4	(BL4)	DAC S-IV VEHICLE NETWORK
GRAY 1	(GY1)	TELEMETER S=I AND S=IV
GRAY 2	(GY2)	ALL OTHER RF VEHICLE SYSTEMS
GREEN 1	(GN1)	SC TEST PROJECT ENGINEER
GREEN 2	(GN2)	SC LAUNCH ESCAPE SYSTEM/ORDNANC
GREEN 3	(GN3)	MSC QUALITY CONTROL
GREEN 4	(GN4)	NAA QUALITY CONTROL
GREEN 5	(GN5)	(NOT ASSIGNED)

SA-7

WHITE 3 (	WH3)	SC FACILITIES
WHITE 4 (	WH4)	SC ENVIRONMENTAL CONTROL
WHITE 5 (	WH5)	SC ELECTRICAL POWER SYSTEM
PURPLE 1 (	PR1)	SC INSTRUMENTATION
PURPLE 2 (	PR2)	SC RF SYSTEMS
PURPLE 3 (	PR3)	SC GROUND SUPPORT EQUIPMENT
*S-I (WHITE 1)	(81)	S-I, IU TEST CONDUCTOR
#S-IV (WHITE 2	(SIV)	S-IV TEST CONDUCTOR
sc (	SC)	SPACECRAFT TEST CONDUCTOR
*S (	S)	SPACE VEHICLE TEST SUPERVISOR

. CHANNELS LOCATED ON "ALL" END UNITS

SA-7

### OPERATING STATIONS CODE IDENTIFICATION \*\*\*\*\*\*\*\*\*\*\*

### BLOCKHOUSE CONSOLES

C1CM	S-I CAMERA PANEL
Company of the Compan	S-I COMPONENT TEST PANEL
	S-I DESTRUCT PANEL
CIFP	S-I FIRING PREP PANEL
CIFR	S-I FIRING PREP PANEL S-I FIRING PANEL
CILS	LAUNCH SEQUENCER PANEL
	S-I MEASURING PANEL
	S-I NETWORKS PANEL
CIOP	S-I ORDNANCE MONITOR PANEL
CITC	S-I ORDNANCE MONITOR PANEL S-I TEST CONDUCTOR
CIVP	S-I VEHICLE PRESSURE PANEL
C4DP	S-IV DESTRUCT PANEL
C40P	S-IV ORDNANCE MONITOR PANEL
CATC	S-IV TEST CONDUCTOR
CATE	S-IV TEST CONDUCTOR S-IV ASTRIONICS TEST ENGINEER
CDCG	DC GENERATOR PANELS
	DIRECTOR KSC
CETE	S-IV ELECTRICAL EQUIPMENT TEST ENGINEER
CGCM	GROUND CAMERA PANEL
CINV	INVERTER PANELS S-IV INSTRUMENTATION POWER CONTROL AND MONITOR PANEL
CIPC	S-IV INSTRUMENTATION POWER CONTROL AND MONITOR PANEL
CISP	INSTRUMENTATION STATUS PANEL
CLTC	LAUNCH VEHICLE TEST CONDUCTOR
CLVN	VEHICLE NETWORKS PANEL
CNTS	NASA TEST SUPERVISOR PERISCOPE
CPER	PERISCOPE
	S-IV PROPULSION SYSTEM PREPARATION AND CONTROL PANEL
	S-IV PNEUMATIC SYSTEM CONTROL PANEL
CPSO	PAD SAFETY OFFICER
CPSP	POWER SWITCHING PANEL
CPTE	S-IV PROPULSION TEST ENGINEER
CSPC	S-IV STAGE POWER CONTROL AND MONITOR PANEL
CSRP	SEQUENCE RECORDER READOUT PANEL
	S-IV STAGE SYSTEM STATUS PANEL
CSTC	SPACECRAFT TEST CONDUCTOR
CUMP	IU MEASURING PANEL
CUNC	IU NITROGEN CONTROL PANEL
	IU NETWORKS PANEL
	S-IV VACUUM MONITOR PANEL
CVTE	S-IV VEHICLE DESIGN TEST ENGINEER

SA-7

#### BLOCKHOUSE RACKS

R4FC	HYDRAULIC CONTROL AND MONITOR PANEL
RATC	AUTOMATIC THEODOLITE CONTROL
RBHM	BLOCKHOUSE MEASURING
	COMBUSTION STABILITY MONITOR
	ENVIRONMENTAL CONTROL SYSTEM
RECE	S-IV FLIGHT CONTROL ENGINEER
RFDM	FIRE DETECTION MONITOR
RFSR	S-IV FLIGHT SEQUENCE RECORDER
RGCC	GUIDANCE COMPUTER CONTROL
RGMC	GUIDANCE MONITOR CONTROL
RGPP	GROUND PRESSURE PANEL
RGRM	GROUND FACILITY MEASURING
RLFO	LIFTOFF CLOCK
RLHC	LH2 CONTROL
RLHM	LH2 MONITOR
RLOC	LOX COMPUTER
RLOO	LOX CONTROL
RPLC	S-IV PROPELLANT LOADING CONTROL
	PROPELLANT RECORDER READOUTS
	PROPELLANT SEQUENCE RECORDERS
RPUP	S-IV PROPELLANT UTILIZATION PANEL
RRPC	RP-1 COMPUTER
RRPO	RP-1 CONTROL
RS4C	ST-124 CONTROL PANEL
RSCR	S-IV STRIP CHART RECORDERS
RSEQ	S-I AND IU SEQUENCE RECORDER
	TRAFFIC LIGHT CONTROL PANEL
RUAC	UMBILICAL ARM CONTROL
RVFC	VEHICLE FLIGHT CONTROL
	S-I AND IU VISICORDER
	WATER CONTROL PANEL
R4VC	S-IV VISICORDER

### SERVICE STRUCTURE, LAUNCH PEDESTAL AND VEHICLE

```
5124
        ST-124 IU
SIFC
         FLIGHT CONTROL S-I ENGINE
SINA
         NETWORKS UNIT 1-9
SINF
         NETWORKS UNIT 10-12
S1TM
         S-I TM
         S-I VEHICLE MECHANICAL PLATFORM NO. 1
SIVM
         S-IV VEHICLE ASTRIONICS ENGINEER
S4AE
S4CC
        S-IV CREW CHIEF
S4EE
        S-IV VEHICLE ELECTRICAL EQUIPMENT ENGINEER
S4FC
         FLIGHT CONTROL S-IV STAGE
         S-IV MECHANICAL EQUIPMENT ENGINEER
S4ME
         S-IV ORDNANCE ENGINEER
540E
        S-IV PROPULSION ENGINEER
SAPE
S4SE
        S-IV STRUCTURES ENGINEER
        S-IV VEHICLE DESIGN ENGINEER
SAVE
SESP
        ENGINE SERVICE PLATFORM
        GROUND NETWORKS ON PEDESTAL
SGNP
SGMP
         GSE MECHANICAL ON PEDESTAL
         HORIZON SENSOR
SHZN
         S-IV MATERIAL RESEARCH AND PROCESS ENGINEER
SMRP
         S-IV MECHANICAL TEST ENGINEER
SMTE
SORR
        OVERALL TEST BATTERY ROOM
SOPS
        S-IV OPERATIONS ENGINEER
SQFC
         FLIGHT CONTROL Q-BALL
SSM1
         SHORT CABLE MAST NO. 1
SSM2
         SHORT CABLE MAST NO. 2
SSMS
         STRUCTURE MEASURING STATION
SSSC
         SERVICE STRUCTURE CONTROL
        FLIGHT CONTROL IU
SUFC
         GUIDANCE IU
SUGU
SUTM
        IU TM
SUVM
         IU VEHICLE MECHANICAL PLATFORM NO. 3
         NETWORKS IU
SUVN
SVME
         VEHICLE MEASURING
SVMP
         VEHICLE MECHANICAL ON PEDESTAL
SVOC
         VEHICLE NETWORKS DAT CONSOLE
         VEHICLE NETWORKS RECORDERS
SVOR
SVRF
        VEHICLE RF
```

SA - 7

AGCS

AAAE	S-IV AGCS ELECTRICAL
AEBR	EMERGENCY BATTERY ROOM
AGAA	ST-124 ALIGNMENT AMPLIFIER
ANAG	AGCS NETWORKS
APCD	PNEUMATIC CONTROL DISTRIBUTOR
AFCS	ECS ON AGCS ROOF

#### BLOCKHOUSE FIRST FLOOR

BITM S-1 BLOCKHOUSE TELEMETER S	TATION
B4TM S-IV BLOCKHOUSE TELEMETER	STATION
BCDC COUNTDOWN CLOCK	
BCOM COMMUNICATIONS AND TV CONT	ROL
BDOP ODOP GROUND STATION	
BGRN LCC NETWORKS	
BRCA RCA SEQUENCER	
BRFS BLOCKHOUSE RF STATION	
BSOC SATURN OPERATIONAL COMPUTE	R

#### UMBILICAL TOWER

S-IV PROPULSION E	INGINEER	1		
ECS ON UMBILICAL	TOWER			
UMBILICAL GROUND	NETWORK	S		
UMBILICAL GSE				
S-IV PNEUMATIC CO	NSOLES	A	AND B	
Q-BALL GSE				
UMBILICAL SWING A	RM NO,	1	CONTROL	PANEL
UMBILICAL SWING A	RM NO.	2	CONTROL	PANEL
UMBILICAL SWING A	ARM NO.	4	CONTROL	PANEL
	ECS ON UMBILICAL UMBILICAL GROUND UMBILICAL GSE S-IV PNEUMATIC CO Q-BALL GSE UMBILICAL SWING A UMBILICAL SWING A UMBILICAL SWING A	ECS ON UMBILICAL TOWER UMBILICAL GROUND NETWORK UMBILICAL GSE S-IV PNEUMATIC CONSOLES Q-BALL GSE UMBILICAL SWING ARM NO. UMBILICAL SWING ARM NO. UMBILICAL SWING ARM NO.	UMBILICAL GROUND NETWORKS UMBILICAL GSE S-IV PNEUMATIC CONSOLES A Q-BALL GSE UMBILICAL SWING ARM NO, 1 UMBILICAL SWING ARM NO, 2 UMBILICAL SWING ARM NO, 3	ECS ON UMBILICAL TOWER UMBILICAL GROUND NETWORKS UMBILICAL GSE S-IV PNEUMATIC CONSOLES A AND B

### PAD AND OTHER COMPLEX AREAS

PALN	VEHICLE ALIGNMENT STATIONS
PATB	AUTOMATIC THEODOLITE BLDG.
PHPF	HIGH PRESSURE FACILITY
PLHF	LH2 FACILITY
PLOF	LOX FACILITY
PPSO	PSO ON PAD
PRPF	RP-1 FACILITY
PVP1	VALVE PIT NO. 1

SA-7

REVISION

PAGE TEST NO. VEHICLE

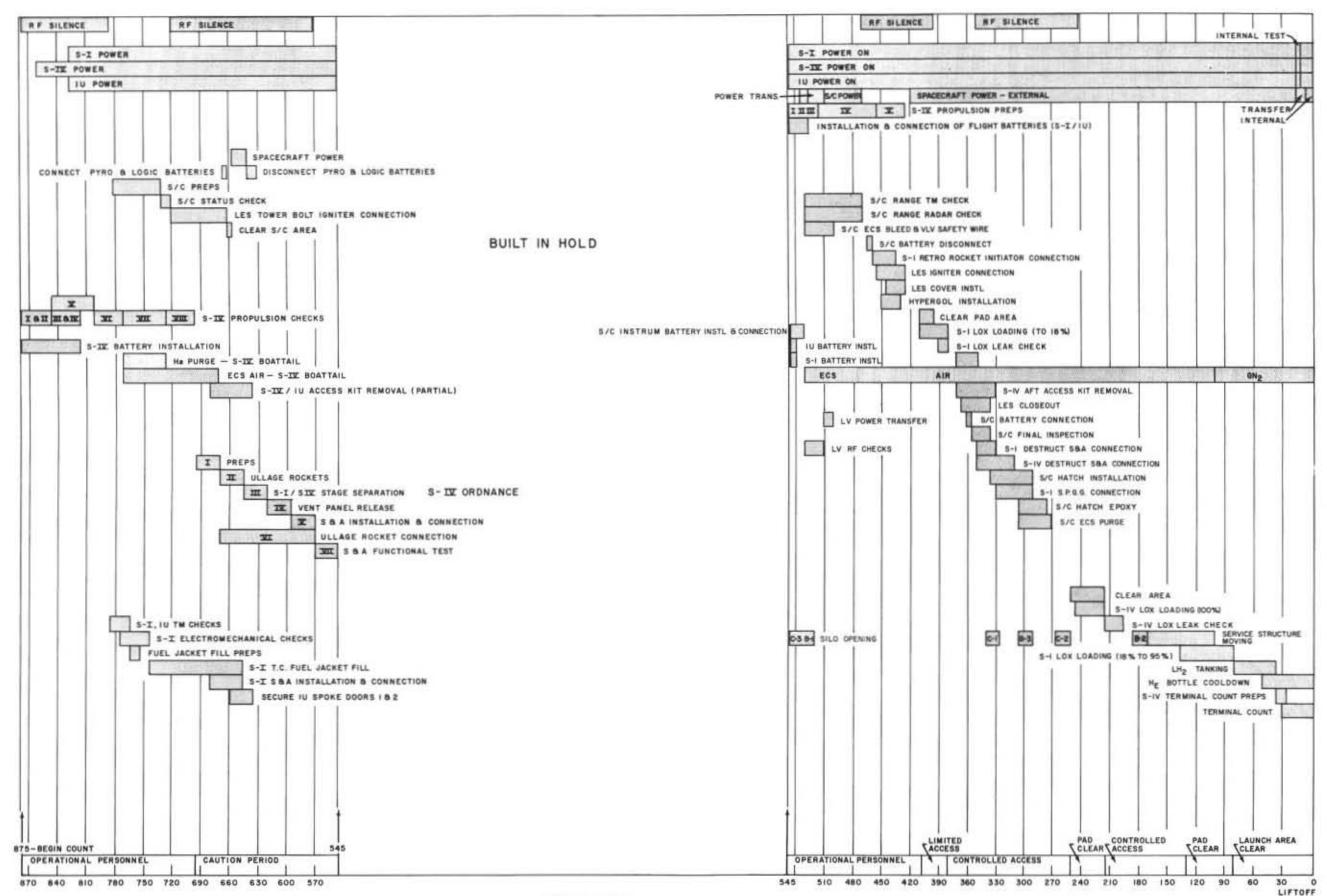
STATIONS EXTERNAL TO COMPLEX 37

OCCF CONVERTER COMPRESSOR FACILITY RANGE OPERATIONS THROUGH THE SRO

ORNG

TELEMETER HANGAR D STATION

# PRELAUNCH PREPARATIONS - L-I DAY PART IX



PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
					PART IV	
					LAUNCH PRECOUNT	
875 1						
	SRO	1	CNTS	ORNG	VERIFY COMPLEX IS ON CRITICAL POWER.	
		2		S1VM SUVM	OPEN VEHICLE.	
	BL1	3		CINV	GROUND INVERTER 2 AND 60 KW GENERATOR	
		4		SVME	REMOVE ALL PNEUMATIC CALIBRATION LINES AND SAFETY WIRE CALIBRATION VALVES.	
		5			PREPARE THE PNEUMATIC DISTRIBUTION SYSTEM FOR PRELAUNCH OPERATIONS PER PROCEDURE LVO-E-7015.	
	PA	6		CNTS	RF SILENCE "ON".	
	RD3	7			PREPARE WATER SYSTEM PER PROCEDURE LVO-E-7014, PART I.	
	RD4	8			SET UP LH2 TRANSFER SYSTEM PER PROCEDURE LVO-L-1034.	
	SIV	9	C4TC		APPLY POWER TO S-IV PER PROCEDURE DAC-N-7001.	
	SIV	10	C4TC		VERIFY UMBILICAL TOWER WATER SYSTEM "ON",	
	BL4	11	C4TC		BEGIN S-IV BATTERY INSTALLATION PER PROCEDURE DAC-N-7005.	
	RD2	12	C4TC		BEGIN S-IV PROPULSION PRELIMINARY PREPS PER PROCEDURE DAC-P-7005, PART I.	
	RD1	13	S4PE		REMOVE CHILLDOWN DUCT DESSICATORS AND STUB FIN MAST ADAPTERS.	
	RD2	14	C4TC		PERFORM PROPELLANT LINES VACUUM CHECK PER DAC-P=7005, PART II.	
	RD3	15			PERFORM ECS PREPARATIONS AND OPERATIONS PER PROCEDURE LVO-E-7018.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
875 1						
	GY1	16	B1TM	SUTM S1TM B1TM	PREPARE S1 & IU TELEMETER SYSTEMS PER PROCEDURE LVO-R-3001.	
	GY2	17	BRFS	SVRF	PREPARE S1 & IU RF SYSTEMS PER PROCEDURE LVO-R-3002.	
	WH2	18	C4TC	S4AE	PERFORM RF & TM SYSTEMS PRE-TEST PREPARATIONS PER PROCEDURE LVO-R-3003.	
8651						
	SIV	1	C4TC	SMTE	VERIFY QUICK DISCONNECTS AND UMBILICAL CARRIERS PROPERLY INSTALLED, (SWING ARM NO. 2 AND GH2 VENT COUPLING).	
					NOTE	
					STEEL SHEAR PINS, SAFETY BLOCKS AND SAFETY WIRE INSTALLED, LANYARDS NOT CONNECTED, AND EJECT ACTUATOR PNEUMATIC LINES NOT CONNECTED.	
	s	2	CLTC	C47C	VERIFY POWER ON S-IV.	
	SIV	3	C4TC		ALL STATUS SELECT SWITCHES TO "ON SCHEDULE",	
8451						
	RD2	1	C4TC		VERIFY COMPLETION OF S-IV PROPULSION PRELIMINARY PREPS PER DAC-P-7005. PART I.	
	RD2	2	C4TC	CPTE	VACUUM PUMP HELIUM HEATER AND ENGINE IGNITERS PER PROCEDURE DAC-P-7005, PART III.	
	KD2	3	CATC	CPTE	PERFORM LH2 AND LOX SYSTEMS PURGES AND SAMPLING PER PROCEDURE DAC-P-7005, PART V.	
	KD2	4	C4TC		PERFORM BULKHEAD SAMPLING PER PROCEDURE DAC-P-7005, PART IV.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-830 0						
-825†		1		S1NF SVOC	VERIFY DESTRUCT MONITOR SYSTEM READY.	
-825						
	BL1	1		CINP	APPLY POWER TO S-I.	
	BL1	2		CUNP	APPLY POWER TO IU.	
		3		SVMP	CHECK MAIN FUEL VALVES FOR LEAKAGE AND RECORD IF LEAKAGE IS NOTED.	
		4		SVMP	INSTALL FUEL DRAIN SCREWS/WASHERS.	
		5		SVMP	CHECK THAT THE CONTROL 750 REGULATOR IS STILL ADJUSTED TO 750 PSIG.	
	BL1	6		CPSP	SERVICE STRUCTURE DAT POWER "ON",	
	BL1	7		SVOC	PULSE SENSOR POWER "ON".	
	BL1	8		SVOC	TEST DESTRUCT PULSE SENSORS AND RESET.	
	BL1	9	C4TC	S1NA	BEGIN NO VOLTAGE AND CONTINUITY CHECKS ON TURBINE SPINNER INITIATOR CIRCUITS PER PART I OF PROCEDURE 1-LSII-709.	
-820 V						
	RD1	1	C1CT	RBHM	VERIFY SOI LOX TANK PRESSURE MEASURE- MENTS INDICATE TANK PRESSURE BEFORE VENTING.	
-815						
	BL1	1			STANDBY TO TURN IGNITION CIRCUIT BREAKER AND CONTACTOR "ON",	
	RD1	2	C1TC	SVMP SGMP	PREPARE TO RUN ELECTRO-MECHANICAL TEST PER PROCEDURE LVO-MV-P-1006.	
		3		SHZN	HORIZON SENSOR CHECKS.	
	KD2	4	C4TC	CPTE	VERIFY COMPLETION OF HELIUM HEATER AND ENGINE IGNITER VACUUM PUMPING PER DAC-P=7005, PART III.	

LAUNCH COUNTDOWN SATURN/APOLLO
DATE SEPTEMBER 5, 1964 LAUNCH OPERATIONS REVISION

# SATURN/APOLLO

PAGE 143
TEST NO. 7-LSV1-300
VEHICLE SA-7 143 SA-7

TIME	COMM.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
8151						
	RD2	5	C4TC	CPTE	VERIFY COMPLETION OF PROPELLANT LINES VACUUM CHECK PER DAC-P-7005, PART II	
	BL4	6	CATC	CATE	VERIFY COMPLETION OF S-IV BATTERY INSTALLATION PER DAC-N-7005.	
	RD2	7	C4TC	CPTE	VERIFY COMPLETION OF BULKHEAD SAMPLING PER DAC-P-7005, PART IV.	
8001						
	RD2	1	C4TC	CPTE	VERIFY COMPLETION OF LH2 AND LOX SYSTEM PURGES AND SAMPLING PER DAC-P-7005, PART V.	
	RD2	2	C4TC	CPTE	PERFORM PROPELLANT VALVES AND IGNITERS FUNCTIONAL CHECK PER PROCEDURE DAC-P-7005, PART VI.	
790 *						
	SRO	1	CNTS	ORNG	DO NOT INTERROGATE C-BAND BEACON WITH 1.16 RADAR UNTIL REQUESTED.	
785						
	BL1	1	C1TC	S1NA	VERIFY NO VOLTAGE AND CONTINUITY CHECKS COMPLETE ON S.P.G.G. INITIATOR CIRCUITS PRIOR TO RF CHECKS.	
		2	Citc	CINP	RESET IGNITION SEQUENCER	
		3	CATE	S4CC	START FORWARD INTERSTAGE AUXILIARY BLOWER.	
	SRO	4	CNTS	ORNG	RF CLEARANCE FOR ALL SYSTEMS AND VERIFY READY FOR READOUTS.	
	SRO	5	CNTS	ORNG	DO NOT INTERROGATE C-BAND BEACON WITH 1,16 RADAR UNTIL REQUESTED.	
		6			RF AND TM CHECKS.	
	PA	6-1		CNTS	RF SILENCE "OFF".	

## SATURN/APOLLO

PAGE 144
TEST NO. 7-LSV I = 300
VEHICLE SA = 7 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARK
7851						
	S	6-2	CLTC	C1MP	TELEMETERS F1, F2, F3, P2, S1 AND S2 "ON",	
	S	6-3	CLTC	CUMP	TELEMETERS F5, F6, S3 AND P1 "ON",	
	S	6-4	CLTC	C1MP,	AUXILIARY EQUIPMENT "ON".	
	S	7	CLTC	C1MP CUMP	MEAS. ON DDAS "ON".	
	S	7-1	CLTC	C1MP CUMP	SWEEP FREQUENCY CALIBRATOR "ON".	
	S	7-2	CLTC	BDOP	ODOP GROUND TRANSMITTER NO. 2 "ON".	
	S	7~3	CLTC	CUMP	ALTIMETER "ON".	
	S	7-4	CLTC	CUMP	GUIDANCE COMMAND RECEIVER "ON".	
	S	7~5	CLTC	CUMP	ODOP "ON",	
	S	7~6	CLTC	CUMP	MINITRACK "ON".	
	S	7-7	CLTC	CUMP	AZUSA "ON",	
	S	7=8	CLTC	CUMP	C=BAND BEACON (RADAR) TO "ON",	
	S	7-9	CLTC	CUMP	MISTRAM "ON",	
	SRO	7-10	CNTS		CHECK MISTRAM AND AZUSA AND REPORT READOUT.	
	S	7-11	CLTC	CIPC	TELEMETER D1, D2, AND D3 "ON".	
	S	7-12	CLTC		VERIFY TAPE RECORDER READY INDICATIONS.	
	S	7-13	CLTC		TELEMETER CALIBRATIONS TO "PREFLIGHT",	
	S	7-14	CLTC		CALIBRATION SWITCH TO MANUAL (VERIFY VCO CALIBRATION LIGHT OUT).	
	S	7-15		C1MP CUMP	PRE-FLIGHT CALIBRATION TO 50%.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-785°						
	S	7-16	CLTC	B4TM	TELEMETER RECORDINGS "ON".	
	S	7-17	CLTC	B4TM	PERFORM TELEMETER CHANNEL READOUTS.	
					NOTE	
					VERIFY CHANNEL 21-24 OF TELEMETER D2 HIGH LEVEL PDM READING 100%.	
782						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	ОТМО	TELEMETER RECORDINGS "ON".	
	S	1-2	CLTC	ORNG	COMMAND CARRIER "ON".	
	S	1-3	CLTC	RBHM	SANBORN RECORDER TO 2 MM/SEC.	
	S	1~4	CLTC	CSSP	VERIFY MSFC TALKBACK ENABLE.	
	S	1~5	CLTC	CLVN	DESTRUCT ENABLE PONP.	
	S	1~6	CLTC	C4DP	VERIFY SAFETY BUS INDICATION "ON".	
	S	1~7	CLTC	CiDP	S-I CDR S NO. 1 AND NO. 2 "ON".	
	S	1-8	CLTC	C10P	MODULE POWER SUPPLY "ON".	+
	S	1=9	CLTC		FUNCTION SELECTOR TO FIRING UNIT VOLTAGE CHECK.	
	S	1-10	CLTV		PANEL SELECT TO "AUXILIARY RECORDER".	
	S	1-11	CLTC	R4VC	VISICORDER MON"	
	S	1-12	CLTC	RFSR	SEQUENCE RECORDERS TO 10 MM/SEC.	
	S	1-13	CLTC	C4DP	S-IV CDR #1 AND #2 "ON".	
	S	1-14	CLTC		VERIFY EBW S #1 AND #2 "OFF" AND "NOT READY",	

PAGE TEST NO. VEHICLE

TIME	COMM.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
780'						
		1			RF AND TM CHECKS.	
	SRO	1-1	CNTS	ORNG	INTERROGATE IU AND SPACECRAFT C-BAND BEACONS WITH 1.16 RADAR AND 0.18 RADAR RESPECTIVELY.	
	S	1-2	CLTC	RSEQ	S=I RECORDERS TO MINUTE SPEED AND TIME PULSE "ON",	
	S	1-3	CLTC	C1CM	TELEVISION TO "FILAMENT".	
	S	1-4	CLTC	C4DP	S-IV CDR #1 AND #2 TO "INTERNAL",	
	S	1-5	CLTC	RBHM	SANBORN RECORDER TO 25 MM/SEC.	
	S	1-6	CLTC	RSEQ	RECORDERS FAST.	
	S	1-7	CLTC	ORNG	DESTRUCT COMMAND (MOMENTARILY THROUGH CUTOFF).	
	S	1-8	CLTC	RSEQ	RECORDERS SLOW.	
	S	1-9	CLTC	RBHM	SANBORN RECORDER TO 5 MM/SEC.	
	S	1-10	CLTC	C4DP	VERIFY S-IV EBW NO. 1 AND NO. 2 GO TO INTERNAL AND CHARGE.	
	s	1-11	CLTC	C1DP	VERIFY S-I EBW'S NO. 1 AND NO. 2 LATCH INTERNAL AND CHARGE ON GROUND POWER.	
	S	1-12	CLTC	CNTS	VERIFY CUTOFF COMMAND FOR 2.0 SECONDS.	
	S	1-13	CLTC	CNTS	VERIFY DESTRUCT COMMAND 2.0 SECONDS AFTER CUTOFF RECEIVED.	
	S	1-14	CLTC	SVOC	VERIFY S=I AND S-IV DESTRUCT DOES NOT OCCUR AT THE VEHICLE.	
	S	1-15	CLTC	ORNG	RELEASE DESTRUCT COMMAND.	
	S	1-16	CLTC	C4DP	S-IV CDR SWITCHES NO. 1 AND NO. 2 "OFF" (VERIFY CDR'S REMAIN ON).	
	S	1-17	CLTC	ORNG	"SAFE" COMMAND AND RELEASE UPON REQUEST	-
	S	1-18	CLTC	C4DP	VERIFY "SAFE" COMMAND.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
780†						
	s	1-19	CLTC	C4DP B4TM	VERIFY Salv CDR #1 AND #2 AND EBW #1 AND #2 REMAIN "ON".	
	S	1-20	CLTC	C1DP C4DP	EBW #1 AND #2 TO "EXTERNAL" (VERIFY "OFF" AND "NOT READY"),	
	S	1-21	CLTC	C4DP	S-IV CDR'S #1 AND #2 TO "EXTERNAL" AND VERIFY HOFFH,	
	s	1-22	CLTC	CiDP	S-I CDR NO. 1 AND NO. 2 "OFF",	
	s	1-23	CLTC	CLVN	DESTRUCT ENABLE "OFF",	
	s	1-24	CLTC	RFSK	SEQUENCE RECORDERS TO 2 MM/SEC.	
	S	1~25	CLTC	R4VC	VISICORDER MOFFM,	
		1-26	CLTC	врор	ODOP GROUND TRANSMITTER #2 "OFF" AND #1 "ON"	
	S	1-27	CLTC	RSEQ	RECORDERS TO HOUR SPEED, TIME PULSE	
	S	1-28	CLTC	RBHM	SANBORN RECORDER TO 2 MM/SEC.	
	SRO	1-29	CNTS	ORNG	COMMAND CARRIER "OFF",	
	SRO	1-30	CNTS	ORNG	REPORT COMMAND CARRIER DEVIATION, ANTENNA, ANTENNA ORIENTATION, AND POWER OUTPUT.	
	RD3	2	CLTC	APCD	VERIFY COMPLETION OF PNEUMATIC PREPS FOR PRELAUNCH PROCEDURE LVO-E-9015.	
	sc	3	CNTS	CSTC	PERFORM SPACECRAFT PREPARATIONS.	
7761						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	CIPC	MANUAL CALIBRATION TO 0 VOLTS.	
	s	1-2	CLTC	BITM	TELEMETER RECORDINGS "ON",	
	s	1-3	CLTC	C1MP CUMP	TELEMETER CALIBRATION TO "AC COMMAND".	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
775! 15"						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS "ON" AND "OFF".	
7751						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1MP CUMP	TELEMETER CALIBRATION TO PREFLIGHT.	
	S	1-2	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMAND.	
	S	1-3	CLTC	C1MP CUMP	PREFLIGHT CALIBRATION TO 0%.	
	S	1-4	CLTC	CIPC	MANUAL CALIBRATION TO +2.5 VOLTS.	
	S	1-5	CLTC	C1CM	TELEVISION TO "B+".	
	RD1	2		C1CT C1FP S1VM	VENT LOX TANK STANDBY PRESSURE.	
	RD1	3		C1TC		
				SUMP	PERFORM ELECTRO-MECHANICAL TEST PER PROCEDURE LVO-MV-P-1006.	
7741						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	CIPC	MANUAL CALIBRATION TO +5 VOLTS.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-773' 30"						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1MP CUMP	PREFLIGHT CALIBRATIONS TO 100%.	
	S	1-2	CLTC	CIPC	MANUAL CALIBRATION TO 0 VOLTS.	
7721						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1MP CUMP	RECORDER TRANSFERS "ON".	
	S	1-2	CLTC	B1TM OTMD	8 KC OSCILLATORS "ON".	
	S	1-3	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS.	
	S	1-4	CLTC	CIPC	MANUAL CALIBRATION TO LIGHTS OFF STEP.	
	S	1-5	CLTC	C1MP CUMP CIPC	TAPE RECORDERS TO RECORD.	
	S	1-6	CLTC	C1MP CUMP	PREFLIGHT CALIBRATIONS TO 0%, 25%, 50%. 75%, AND 100% IN 2 SECOND INCREMENTS.	
	S	1-7	CLTC	CIPC	STEP THROUGH MANUAL CALIBRATION IN 2 SECOND INCREMENTS,	
	S	1-8	CLTC	C1MP CUMP CIPC	TAPE RECORDERS #OFF#.	
	S	1-9	CLTC	C1MP CUMP	TELEMETER CALIBRATION TO INFLIGHT.	
	S	1-10	CLTC	CIPC	CALIBRATION TO AUTOMATIC.	
	S	1-11	CLTC	CIPC	GIVE CAL START.	
	S	1-12	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMAND "ON" FOR 5 SECONDS.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-7721 30"						
	S	1-13	CLTC	CIPC	BRIDGE CALIBRATE HIGH FOR 10 SECONDS.	
	S	1-14	CLTC	CIPC	BRIDGE CALIBRATE LOW FOR 10 SECONDS.	
	S	1-15	CLTC	CUMP	P1 TELEMETER CALIBRATION.	
	S	1-16	CLTC	RBHM	S=I AND IU HIGH CALIBRATE COMMAND (15 SEC.).	
	S	1-17	CLTC	RBHM	S=I AND IU LOW CALIBRATE COMMAND (15 SEC.).	
	S	1-18	CLTC	RBHM	S=I AND IU RUN COMMAND.	
	S	1-19	CLTC	B1TM OTMD	8 KC OSCILLATOR "OFF".	
770 °						
	RD2	1	C4TC	CPTE	VERIFY COMPLETION OF PROPELLANT VALVES AND IGNITERS FUNCTIONAL CHECK PER DAC-P-7005, PART VI.	
	RD2	2	CATC		PERFORM S-IV ENGINE AND HELIUM HEATER PURGE PER PROCEDURE DAC-P-7005, PART VII;	
					NOTE	
					THERE WILL BE NO ACTIVITY IN THE S-IV AFT INTERSTAGE UNTIL THE COMPLETION OF PART VII OF PROCEDURE DAC-P-7005.	
	RD3	3	CPTE	RECS	PROVIDE ECS AIR TO S-IV AFT SECTION DURING HELIUM PURGE OPERATIONS AS REQUESTED.	
7691						
		1			RF AND TM CHECKS	
	S	1-1	CLTC	C1MP CUMP	RECORDER TRANSFERS "OFF".	
	S	1-2	CLTC	CIPC	GIVE CAL START.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
7691						
	S	1-3	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS "ON" FOR 5 SECONDS.	
	S	2	CLTC	CUMP	P1 CALIBRATION COMMAND "ON" AND "OFF".	
	S	2-1	CLTC	C1MP CUMP CIPC	TAPE RECORDERS TO "PLAYBACK".	
	S	2-2	CLTC	CUMP	C-BAND BEACON (RADAR) "OFF" WHEN READOUT COMPLETE.	
	S	2-3	CLTC	CUMP	AZUSA "OFF" (WHEN READOUT IS COMPLETE).	
7661 30"						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1MP CUMP CIPC	CONFIRM TAPE RECORDER READY INDICATIONS (BEFORE PROCEEDING).	
	S	1=2	CLTC	C1MP CUMP CIPC	TAPE RECORDERS MOFFM.	
	S	1-3	CLTC	CIPC	VERIFY VCO CALIBRATION "OFF" AND "AUTO"	
	S	1-4	CLTC	C1MP	TELEMETERS F1, F2, F3, P2, S1 AND S2 "OFF",	
	S	1-5	CLTC	CUMP	TELEMETERS F5, F6, S3 AND P1 "OFF".	
	S	1-6	CLTC	C1MP CUMP	SWEEP FREQUENCY CALIBRATOR "OFF".	
	S	1-7	CLTC	C1MP CUMP	AUXILIARY EQUIPMENTS "OFF",	
	S	1~8	CLTC	CIPC	TELEMETERS D1: D2 AND D3 "OFF",	
	S	1-9	CLTC	B1TM B4TM OTMD	TELEMETER RECORDINGS "OFF".	

PAGE

152 TEST NO. 7-LSVI-300 VEHICLE SA-7 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-766¹ 30"						
	S	1-10	CLTC	CUMP	ALTIMETER "OFF".	
	S	1-11	CLTC	C1CM	TELEVISION MOFFM.	
		2			RF AND TM CHECKS.	
	S	2-1	CLTC	CUMP	ODOP "OFF",	
	S	2-2	CLTC	CUMP	MINITRACK "OFF".	
	S	2-3	CLTC	CUMP	MISTRAM "OFF".	
	S	2-4	CLTC	CUMP	GUIDANCE COMMAND RECEIVER "OFF",	
	S	2=5	CLTC	BDOP	ODOP GROUND TRANSMITTER NO. 1 "OFF",	
	S	2-6	CLTC	B4TM	NORMALIZE ALL SWITCHES AND SECURE STATION.	
	RD1	3	C1TC	SVMP	BEGIN FUEL JACKET FILL PREPARATIONS.	
⇒755¹						
	BR1	1	C1TC	RFDM	VERIFY FIRE DETECTION PROBE COVERS ARE REMOVED.	
	BR1	2	C1TC	RÇSM	VERIFY CSM ACCELEROMETERS INSTALLED.	
	RD1	3	C1TC	SVMP	VERIFY COMPLETION OF FUEL JACKET PREPS.	
-745 ₹						
	RD1	1	CITC	Cict	VERIFY COMPLETION OF THE ELECTRO- MECHANICAL TEST PER LVO-MV-P-1006.	
		2			VERIFY THAT GEARCASE PRESSURE IS INDICATING BETWEEN 2 AND 10 PSIG AT EACH ENGINE POSITION,	
-740 *						
		1		SVMP	REMOVE PROTECTIVE COVERS FROM HEAT SHIELD PANEL MEASUREMENTS.	
		2			REMOVE TURBINE TACHOMETER PROTECTIVE COVER, 8 PLACES,	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-730°						
	sc	1	CNTS	CSTC	VERIFY PREPARATIONS COMPLETE.	
	SC	2		CSTC	STATUS CHECK.	
-725 <sup>1</sup>						
	RD2	1	C4TC	CPTE	VERIFY COMPLETION OF S-IV ENGINE AND HELIUM HEATER PURGES PER DAC-P-7005, PART VII.	
	RD2	2	C4TC	CPTE	PERFORM S-IV PROPULSION SECURING PER PROCEDURE DAC-P-7005, PART VIII,	
		3	S4PE	SVMP	INSTALL H2 DUCT DESSICATORS	
-720°						
	PA	1		CNTS	RF SILENCE ON AND CONTROLLED SWITCHING.	
	sc	2		CSTC	VERIFY SPACE VEHICLE RF SILENCE AND COMPLEX 37 RF SILENCE AND ORDNANCE CREW ON STATION.	
	sc	3		CSTC	BEGIN LES TOWER ORDNANCE CONNECTION PER C-10001.	
-695						
	SIV	1	CATC	CPTE	VERIFY COMPLETION OF S-IV PROPULSION SECURING PER DAC-P-7005, PART VIII,	
	RD1	2	C1TC	S1VM	BEGIN INSTALLATION OF FLIGHT CLOSURES ON THE FORWARD END OF FUEL TANKS 3 AND 4.	
		3		RECS	DISCONTINUE ECS AIR TO THE SHIV AFT SECTION AFTER OK FROM CATC.	
	RD3	4	CLTC	RUAC	PERFORM UMBILICAL SWING ARM PREPARA- TIONS FOR LAUNCH PER PROCEDURE LVO-E-7016.	
	RD3	5	C1TC	SGMP	VERIFY LIFTOFF AND SWING ARM CONTROL SWITCHES ARE CALIBRATED FOR LAUNCH AND PURGE HOSES ARE CONNECTED.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-6951						
	RD3	6	C1TC	SGMP	PREPARE HOLDDOWN ARMS FOR LAUNCH PER PROCEDURE LVO-E-7017.	
	BL1	7	CNTS	ANAG	TURN BREAKER 66A5A10-1 "OFF" (Q-BALL POWER BREAKER)	
	sc	8		CNTS	VERIFY TO SPACECRAFT TO THE Q-BALL POWER BREAKER IS "OFF".	
	RD2	9	C4TC	CVTE	VERIFY COMPLETION OF S-IV ORDNANCE INSTALLATION PRELIMINARY CHECKS PER DAC-0-7011, PART I.	
	RD2	10	C4TC	CVTE	PERFORM ULLAGE ROCKET JETTISON DETONATOR INSTALLATION AND CONNECTION PER PROCEDURE DAC-0-7011, PART II.	
	RD2	11	C4TC	CVTE	PERFORM ULLAGE ROCKET INITIATORS INSTALLATION AND CONNECTION PER DAC-0-7011, PART VI.	
-685↑						
	RD1	1		S1VM	VERIFY S.IV ENGINE CHAMBER SIDE AND EXIT COVERS INSTALLED AND REMOVE PROTECTIVE COVER FROM LOX/SOX PLENUM.	
+680°						
	RD1	1	C1TC	S1VM	INSTALL Sal DESTRUCT SEA UNIT.	
	BL3	2	C1TC	SINF	PERFORM S&A UNIT CHECK PER PROCEDURE	
	GY2	3	BRFS	SVRF	SECURE SI AND IU RF SYSTEMS FOR B.I.H. I-LSII-710. PART I.	
	SC	4		CSTC	VERIFY COMPLETION OF LES TOWER ORDNANCE CONNECTION BEGIN LES IGNITER INSTALLATION,	
-660 P						
	BR1	1	CLTC	SSMS	VERIFY MEASURING READY TO SECURE IU.	
	BL1	2	CLTC	SUVN	VERIFY NETWORKS READY TO SECURE IU.	

PAGE TEST NO. VEHICLE

155 7~LSVI=300 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
560'						
	RD1	3	CLTC	SUVM	INSTALL IU 70 INCH TUBF DOOR.	
	RD1	4	C1TC	S1VM	INSTALL DOORS ON INSTRUMENT CONTAINERS SECURE DOOR 1 FOR FLIGHT. SECURE DOOR 2 FOR BUILT IN HOLD.	
650'						
	RD1	1	CLTC	SUVM	BEGIN INSTALLATION OF IU DOORS AT TUBES 1 AND 2 FOR LAUNCH, TUBE 3 AND 4 FOR B.I.H., AND STOW WORK STANDS.	
	RD1	2	C1TC	SIVM	VERIFY ALL BLAST PLATES ON S-I FORWARD ARE SECURED FOR LAUNCH.	
6451						
	GY1	1	BITM	S1TM SUTM	SECURE S1 AND IU TELEMETER SYSTEMS FOR B, I, H,	
	BR1	2	C1TC	SVME	VERIFY PRESSURE MEASUREMENTS FOR AIR BEARINGS OPEN AND SAFETY WIRED.	
	RD1	3	C1TC	SiVM	VERIFY FUEL TANK 3 AND 4 FORWARD CLOSURES ARE INSTALLED FOR FLIGHT.	
		4		S1VM	SET UP GN2 PURGE LINES AND HAND VALVES FOR RETROPOCKET INITIATOR CONNECTION.	
	RD3	5	C1TC	SGMP	PERFORM SHORT CABLE MAST PREPARATIONS FOR LAUNCH PER PROCEDURE LVO-E-7013.	
	RD1	6	C1TC	S <sub>1</sub> VM	VERIFY COMPLETION OF S-I DESTRUCT SEA INSTALLATION.	
	RD2	7	C4TC	CVTE	VERIFY COMPLETION OF ULLAGE ROCKET JETTISON DETONATOR INSTALLATION AND CONNECTION PER DAC+0-7011, PART II.	
	RD2	8	C4TC	CVTE	PERFORM STAGE SEPERATION DETONATOR INSTALLATION AND CONNECTION PER PROCEDURE DAC=0-7011, PART III.	
		9		C1CT C1FP SVMP	APPLY LOX TANK STAND-BY PRESSURE.	

## SATURN/APOLLO

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
645						
	sc	10		CSTC	VERIFY COMPLETION OF LES IGNITER POST INSTALLATION CHECKS.	
	RD1	11	C1TC	CVMP	VERIFY COMPLETION OF THRUST CHAMBER FUEL JACKET FILL PER LVO-MV-P-1002.	
6351						
	BR1	1		SVME	VERIFY ALL COVERS INSTALLED ON UMA	
	BR1	2		SVME	VERIFY CALIBRATE PLUGS ARE REMOVED FROM LIQUID LEVEL RACKS.	
	RD2	3	C4TC	CVTE	PERFORM S-IV ORDNANCE INSTALLATION PRELIMINARY CHECKS PER PROCEDURE DAC-0-7011, PART I.	
		4		RGCC	FLIGHT COMPUTER SYSTEM "OFF" AFTER VERIFICATION PROCEDURE 5-LIUI-749A IS COMPLETE.	
		5		CNTS	RELEASE SPACECRAFT PERSONNEL FROM STATIONS.	
6201						
	RD3	1	C1TC	SGMP	VERIFY THAT THE FUEL MAST AND LOX MAST ARE IN A STATE OF OPERATIONAL READINESS FOR LAUNCH AS FOLLOWS:	
					A. VERIFY THAT WEATHER PROTECTIVE TAPES AND COVERS HAVE BEEN REMOVED FROM THE MASTS.	
					B. AFTER VERIFYING WITH C1CT THAT THE RELEASE VALVE SWITCHES ARE IN THE "OFF" POSITION, CONNECT THE RETRACT CYLINDERS TO THE MASTS AND VERIFY THAT THE CONNECTING PINS ARE SECURED WITH BALLLOC PINS.	
					C. VERIFY THAT THE AUTOMATIC RETRACT LINES ARE CONNECTED AND THE FITTINGS ARE TIGHT.	
					D. VISUALLY INSPECT THE LOX AND FUEL MAST ASSEMBLIES FOR PROPER	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-620 *						
					MECHANICAL INSTALLATION AND NON-INTERFERENCE WITH ASSOCIATED HARDWARE.	
					E, VERIFY FUEL MAST PURGE LINE CONNECTED .	
					F. VERIFY VACUUM LINE SCREEN INSTALLED AT LAUNCHER LEG.	
	RD2	2	C4TC	CVTE	VERIFY COMPLETION OF STAGE SEPARATION DETONATOR INSTALLATION AND CONNECTION PER DAC-0-7011, PART III.	
	Rúz	3	C4TC	CVTE	PERFORM VENT PANELS RELEASE DETONATORS INSTALLATION AND CONNECTION PER DAC-0-7011, PART IV.	
-5951						
	BR1	1	CITC	SVME	VERIFY ALL INSTALLED RADIATION SHIELD PNEUMATIC TUBING AND TEMPERATURE MEASUREMENTS ARE CONNECTED.	
		2		SVME	REMOVE COVERS FROM CALORIMETER FACES.	
	RD2	3	CATC	CVTE	VERIFY COMPLETION OF VENT PANELS RELEASE DETONATORS INSTALLATION AND CONNECTION PER DAC=0-7011, PART IV.	
	RD2	4	C4TC		PERFORM DESTRUCT S&A INSTALLATION AND CONNECTION PER DAC-0-7011, PART V.	
5701						
	RD1	1	C1TC		PERFORM THRUST CHAMBER FUEL JACKET FILL PER PROCEDURE LVO-MV-P-1002.	
545 9					PEN PRODEBONE EVOLUTIONS	
	RD2	1	CATC		VERIFY COMPLETION OF S-IV DESTRUCT S&A FUNCTIONAL TEST PER DAC-0-7011, PART VII, INCLUDING ORDNANCE SECURING OPERATION.	
		2	CNTS		VERIFY THAT S-IV FLIGHT S&A DEVICE IS ELECTRICALLY DISCONNECTED AND THAT DUMMY DEVICE IS CONNECTED.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
545 1						
		3			VERIFY THAT S-IV DUMMY S&A DEVICE IS CONNECTED, AND THAT FLIGHT S&A IS DISCONNECTED FROM S-IV ELECTRICAL	
		4		CINP	REMOVE POWER FROM S-I.	
		5		CUNP	REMOVE POWER FROM IU.	
		6		RBHM	SANBORN RECORDER "OFF"	
		7		S1VM SUVM	SECURE VEHICLE FOR STANDBY.	
		1	CNTS		"BUILT IN HOLD".	
	S	1-1	CNTS	CSTC	PROCEED TO ARM AND SAFE THE LES.	
		1-2	C4TC	S4ME	PRESSURIZE REMAINING S-IV FORWARD ELECTRONIC BOXES.	
	SRO	1-3	CNTS		LOAD COMPLEX CAMERAS AND CHECKOUT TO BE COMPLETED PRIOR TO RESUMPTION OF COUNT.	

PAGE

159 TEST NO. 7=LSVI=300 VEHICLE SA=7 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
					PART V	
					LAUNCH COUNTDOWN	
					PRIOR TO PICKUP	
		2		SIVM	OPEN S-I FOR BATTERY INSTALLATION.	
		3		SUVM	OPEN IU FOR BATTERY INSTALLATION,	
		4		SIVM	INSTALL SHI FLIGHT BATTERIES.	
		5		S1NF	CONNECT S=I FLIGHT BATTERIES.	
		6		SUVM	INSTALL IU FLIGHT BATTERIES.	
		7		SUVN	CONNECT IU FLIGHT BATTERIES.	
		8		RVFC	PERFORM FLIGHT CONTROL PREPARATIONS PER PROCEDURE 5-LLVI-319.	
		9			MAKE ALL PREPARATIONS NECESSARY FOR S-I AND IU POWER APPLICATION PRIOR TO T-545 MIN PICKUP TIME.	
		10		RGCC	BEGIN FLIGHT COMPUTER SYSTEM PREPARATIONS PER PROCEDURE 5-LIUI-749.	
		11		SUVM	REMOVE REMAINDER OF S-IV/IU ACCESS KIT.	
	GR1	12	BITM	S1TM SUTM	PREPARE S1 & IU TELEMETER SYSTEMS PER LV0-R=3001.	
	GR2	13	BRFS	SVRF	PREPARE S1 & IU RF SYSTEMS PER LVO-R-3002.	
5451						
	SRO	1	CNTS	ORNG	VERIFY COMPLEX IS ON CRITICAL POWER.	
	SIV	5	C4TC		VERIFY UMBILICAL TOWER WATER "ON".	
	YW3	3	CLTQ	RSOC	VERIFY COMPLETION OF 5-LLVI-722.	
	YW3	4	CLTC	BSOC	START DISCRETE MONITORING.	
	SIV	5	CATC	CSPC	APPLY S-IV STAGE POWER PER PROCEDURE DAC-N-7001.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
5451						
	BL2	6		R4FC	SET UP S-IV HYDRAULICS FOR OPERATION.	
	RD2	7	C4TC	CPTE	PERFORM S=IV PROPULSION PRELIMINARY PREPARATIONS PER DAC-P-7009, PART I.	
	BL1	8	CSRP	SINF	VERIFY S=1 SAFETY SWITCHES ARE INSTALLED,	
	BL1	9	CSRP	SUVN	VERIFY IU SAFETY SWITCHES ARE INSTALLED	
	BL1	10	CSRP	SVOC S1NF	VERIFY DESTRUCT MONITORING SYSTEM READY	
	BL1	11	CSRP	CSCG	APPLY S-I GROUND POWER.	
	BL1	12	CSRP	CSCG	APPLY IU GROUND POWER.	
		13		CINP	APPLY S-I STAGE POWER.	
		14		CUNP	APPLY IU STAGE POWER.	
		15		C1MP	GROUND MEASURING VOLTAGE "ON".	
		16		CIMP	VEHICLE MEASURING VOLTAGE "ON".	
	BL1	17	CSRP	C1MP	MEASURING RACKS "ON",	
	BL1	18	CSRP	CUMP	VEHICLE MEASURING VOLTAGE "ON".	
	BL1	19	CSRP	CUMP	MEASURING RACKS "ON",	
	BL1	20	CSRP	CUMP	GROUND MEASURING VOLTAGE "ON",	
	BL1	21	CSRP	CINV	S-I VEHICLE INVERTER #1 "ON",	
	BL1	22	CSRP	CINA	IU VEHICLE INVERTER #1 "ON".	
	BL1	23	CSRP	CINV	IU VEHICLE INVERTER NO, 2 "ON".	
	BL1	24	CSRP	CPSP	SERVICE STRUCTURE OAT POWER "ON",	
	BL1	25	CSRP	RBHM	SANBORN RECORDER TO 2 MM/SEC.	
	BL1	26	CSRP	SVOC	PULSE SENSOR POWER "ON",	
	BL1	27	CSRP	svoc	PULSE SENSOR TEST VOLTAGE "ON" AND	
	BL1	28	CSRP	svoc	VERIFY S-I AND S-IV DESTRUCT INDICATIONS "ON".	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
5451						
	BL1	29	CSRP	SVOC	RESET PULSE SENSORS.	
	811	30	CSRP	svoc	VERIFY S-1 AND S-IV DESTRUCT INDICATIONS "OFF".	
	BLi	31	CSRP	CINV	GROUND INVERTER NO. 1 "ON".	
	BL1	32	CSRP	CINV	GROUND INVERTER NO. 2 "ON".	
	BL1	33	CSRP	C10P	MODULE POWER SUPPLY "ON".	
	RD2	34	C4TC	CPTE	PERFORM PROPELLANT LINES VACUUM CHECKS PER DAC-P-7009, PART II.	
	RD2	35	CATC	CPTE	PERFORM COMMON BULKHEAD MONITOR PREPARATIONS PER DAC-P-7009, PART III.	
	RD1	36	S4PE	SVMP	REMOVE CHILLDOWN DUCT DESSICATORS AND STUB FIN MAST ADAPTERS.	
		37		SVMP	PRECHARGE HYDRAULIC ACCUMULATORS TO 1600 PSIG AND LEAK CHECK SHRADER FITTING: POSITION 1 POSITION 2 POSITION 3 POSITION 4	
	YW4	38	CLTC	SSSC	OPEN SILO C=3 AND B-1.	
	sc	39	CSTC	SSCO	INSTALL AND CONNECT SPACECRAFT INSTRUMENTATION BATTERIES PER PROCEDURE C=0007.	
5401						
	Y W 3	1		RGCC	FLIGHT COMPUTER POWER "ON".	
	SC	S	CSTC		APPLY SPACECRAFT GSE POWER PER PROCEDURE C-0007.	
		3		100	BEGIN REMOVING S-I ENVIRONMENTAL PROTECTION,	
		4		SQFC	REMOVE PROTECTIVE COVER FROM Q-BALL.	
	RD3	5			PREPARE THE PNEUMATIC DISTRIBUTION SYSTEM FOR PRELAUNCH OPERATIONS PER PROCEDURE LVO-E-7015.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-5401						
	BL1	6		ANAG	TURN BREAKER 66A5A10-1 "ON" Q-BALL POWER BREAKER).	
	BL1	7		SUVN	VERIFY IU FLIGHT BATTERIES ARE INSTALLED AND ELECTRICALLY CONNECTED,	
	BL1	8		CUNP	VERIFY +8D10 AND +8D20 BATTERY INDICATIONS,	
	BL1	9		SINF	VERIFY S≈I FLIGHT BATTERIES ARE INSTALLED AND ELECTRICALLY CONNECTED.	
	BL1	10		CINP	VERIFY *1D10 AND *1D20 BATTERY INDICATIONS.	
535 1	BL1	11	RS4C	ANAG	ST-124 HEATER SWITCH TO NORMAL POSITION	
	RD2	1		SMTE	VERIFY QUICK DISCONNECTS AND UMBILICAL CARRIERS PROPERLY INSTALLED (SWING ARM 2 AND GH2 VENT COUPLING.	
					NOTE	
					STEEL SHEAR PINS, SAFETY BLOCKS AND SAFETY WIRE INSTALLED, LANYARDS NOT CONNECTED, AND EJECT ACTUATOR PNEUMATIC LINES NOT CONNECTED.	
		2		C4TC	VERIFY SHIV POWER "ON".	
	SIV	3		C4TC	ALL S-IV STATUS SELECT SWITCHES TO "ON SCHEDULE".	
	BR1	4		SVME RBHM	VERIFY CONNECTION OF SPGG TEMPERATURE MEASURING AND PROPER INDICATION.	
	SC	5			VERIFY COMPLETION OF SPACECRAFT POWER APPLICATION.	
	SRO	6	CNTS		VERIFY FREQUENCY CLEARANCE FOR ALL SPACE VEHICLE RF SYSTEMS.	
	SRO	7	CNTS		VERIFY PATRICK 0-18 RADAR READY TO SUPPORT SPACECRAFT BEACONS AND THAT RADAR IS AWAY FROM THE PAD.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-535 <sup>†</sup>						
	SRO	8	CNTS	ORNG	VERIFY 1,16 RADAR READY TO SUPPORT LAUNCH VEHICLE (IU) BEACON AND THAT RADAR IS AWAY FROM THE PAD AND WILL NOT INTERROGATE UNTIL REQUESTED,	
	SRO	9	CNTS	ORNG	STANDBY FOR READOUT OF SPACECRAFT TM.	
-5301						
	S	1	CLTC	RVFC	CONTROL COMPUTER "ON",	
	S	2	CLTC	RVFC	CHECK CONTROL COMPUTER NULLS.	
	S	3	CLTC	RVFC	CONTROL VOLTAGE "ON".	
	s	4	CLTC	RVFC	Q-BALL "ON",	
	S	5	CLTC	RVFC	CONTROL ACCELEROMETER CHECK.	
	S	6	CLTC	RVFC	IU RATE GYRO CHECK,	
	S	7	CLTC	RVFC	S-I RATE GYRO CHECK.	
	S	8	CLTC	RVFC	ACCELEROMETER SWITCH CHECKS.	
	RD3	9		RECS	OPERATE THE ENVIRONMENTAL CONTROL SYSTEM PER PROCEDURE LVO-E-7018.	
		10		RECS	START ECS AIR TO THE IU.	
		11		SVMP	BEGIN REMOVAL OF STRIPPABLE COATING FROM FIN BASE (8 PLACES).	
	RD1	12		C1FP	VERIFY THAT POWER IS APPLIED TO S-I, THEN OPERATE PRE-VALVES 1 THRU 8 SWITCHES TO "OPEN",	
		13		SVMP	REMOVE CAPS FROM PRE-VALVE CONTROL VALVE VENT PORTS (8 PLACES),	
		14		100 100 mm 100 m	DISCONNECT HEISE PANEL CONNECTED TO FUEL VENT CONTROL LINE. RESTORE CONTROL LINE TO NORMAL AND VERIFY THAT FUEL VENTS 1 AND 2 "CLOSE".	
		15			RF AND TM CHECKS	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
5301						
	PA	15-1		CNTS	RF SILENCE "OFF".	
	S	15-2	CLTC	CUMP	IU C-BAND BEACON (RADAR) "ON",	
	SC	15-3	CSTC	CICM	SPACECRAFT TELEMETER A (LINK 1) "ON".	
	SC	15-4	CSTC	SICM	SPACECRAFT TELEMETER B (LINK 6) "ON".	
	SC	15-5	CSTC	SICM	SPACECRAFT TELEMETER C (LINK 14) "ON",	
	SC	15-6	CSTC	SICM	SPACECRAFT RADAR BEACON NO. 1 "ON".	
	SRO	15-7	CNTS	ORNG	PROCEED WITH TM READOUT AND REPORT WHEN AVAILABLE.	
	S	15-8	CLTC	C1MP	TELEMETERS F1, F2, F3, P2, S1, AND S2.	
	S	15-9	CLTC	CUMP	TELEMETERS F5, F6, S3, P1 AND MINITRACK "ON".	
	S	15-10	CLTC	C1MP CUMP	AUXILIARY EQUIPMENTS "ON".	
	S	15-11	CLTC	C1MP CUMP	MEAS. ON DDAS "ON".	
	S	15-12	CLTC	C1MP CUMP	SWEEP FREQUENCY CALIBRATOR "ON".	
	S	15-13	CLTC	CUMP	VERIFY TAPE RECORDER READY INDICATIONS.	
	S	15-14	CLTC	C1MP	TELEMETER CALIBRATIONS TO "PREFLIGHT".	
	S	15-15	CLTC	C1MP CUMP	PREFLIGHT CALIBRATIONS TO 50%.	
		15-16		CIPC	TELEMETERS D1, D2, AND D3 "ON".	
	S	15-17	CLTC		CALIBRATION SWITCH TO "MANUAL" (VERIFY VCO CALIBRATION LIGHTS "OFF").	
					The sacronal low Clonics of Fig.	

PAGE VEHICLE

165 TEST NO. 7-LSVI-300 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
5301						
	S	15-18	CLTC	B4TM	PERFORM TELEMETER CHANNEL READOUTS.	
					NOTE	
					VERIFY CHANNELS 21-24 OF TELEMETER D2, HIGH LEVEL PDM READING 100%.	
	S	15-19	CLTC	CUMP	RADAR ALTIMETER ™ON™.	
	S	16	CLTC	CUMP	GUIDANCE COMMAND RECEIVER "ON".	
525 9						
		1			RF AND TM CHECKS.	l)
	S	1-1	CLTC	ORNG	BEGIN INTERROGATION OF SPACECRAFT BEACON WITH 0.18 RADAR AND PROCEED WITH READOUT, REPORT WHEN COMPLETE.	
	S	1-2	CLTC	ORNG	INTERROGATE IU C-BAND BEACON WITH 1,16 RADAR AND REPORT READOUT WHEN COMPLETE.	
	S	1-3	CLTC	B1TM OTMD B4TM	TELEMETER RECORDINGS "ON".	
	s	1-4	CLTC	C1MP CUMP	TELEMETER CALIBRATIONS TO "AC COMMAND".	
	S	1-5	CLTC	C1MP CUMP	TM CALIBRATION COMMANDS "ON" AND "OFF", (DELAY 10 SEC BEFORE NEXT ITEM).	
	S	1-6	CLTC	C1MP CUMP	TELEMETER CALIBRATIONS TO "PREFLIGHT",	
	S	1-7	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS.	
	S	1-8	CLTC	CIPC	STEP THRU MANUAL CALIBRATION IN 2 SEC INCREMENTS.	
	S	1~9	CLTC	CIPC	BRIDGE CAL "HIGH" FOR 10 SECONDS.	
		1-10	CLTC	CIPC	BRIDGE CAL "LOW" FOR 10 SECONDS.	

PAGE

166 TEST NO. 7-LSVI-300
VEHICLE SA-7 SA = 7

TIME	COMM.	SEQUENCE	COMMAND STA,	RESPONSE STA.	DESCRIPTION	REMARKS
525 1						
	s	2	CLTC	RPUP	PERFORM P,U, SLEW CHECKS.	
	S	3	CLTC	RBHM	SANBORN RECORDER TO 5 MM/SEC.	
	S	4	CLTC	RS4C	BRING ST=124 UP TO OPERATING CONDITIONS PER PROCEDURE LVO=G-7040.	
	S	5	CLTC	RATC	BRING AUTO THEODOLITE UP TO OPERATING CONDITIONS PER PROCEDURE LVO-G-7044.	
	BR1	6		SVME	VERIFY ALL SOUND INTENSITY MICROPHONE COVERS ARE REMOVED.	
	BRU	7		SVME	VERIFY ALL AMBIENT PRESSURE MEASUREMENTS ARE "OPEN" TO ATMOSPHERE.	6
	BL1	8		SGNP	CHECK ESE LIMIT SWITCHES PER PROCEDURE 1-LLVI-716.	
	RD3	9		RWCP	PREPARE THE PAD WATER SYSTEM FOR LAUNCH PER PROCEDURE LVO-E-7014. PART II.	
		10		ORNG	BRING WATER PRESSURE TO 175 PSI ON 36-INCH LINE WITH 1 PUMP.	
	RD1	11		S1VM C1FP	VENT LOX TANK STANDBY PRESSURE.	
		12		S1VM	AFTER LOX TANKS ARE VENTED, INSTALL CAP ON LOX TANK TEST SENSING LINE AT LOX TANK 4.	
		13		SQFC UQBL	START INSTALLATION OF Q-BALL RETRACTABLE	
		14		PLHF	SET UP LH2 SYSTEM PER LVO-L-1032.	
	sc	15		CSTC	VERIFY COMPLETION OF FLIGHT BATTERY INSTALLATION AND CONNECTION.	
	SC	16		CSTC	BEGIN SPACECRAFT POWER TRANSFER TEST PER PROCEDURE C-0007.	
		17		SYME	BEGIN MANUAL PREFLIGHT PHOTOGRAPHIC INSTRUMENTATION SYSTEM PURGE.	
	YW3	18	CLTC	FIRST SERVICE CONTRACTOR	REQUEST APS-102 (SHORT COMMAND SYSTEM TEST).	

## SATURN/APOLLO

PAGE 167
TEST NO. 7-LSVI-300
VEHICLE SA-7 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA,	RESPONSE STA.	DESCRIPTION	REMARKS
525 1						
522	SRO	19	CNTS	ORNG	BEGIN GUIDANCE COMMAND SHORT TEST). TRANSMISSION.	
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1MP	PREFLIGHT CALIBRATIONS TO 0%.	
	S	1-2	CLTC	C1CM	TV TO "FILAMENT".	
	S	1-3	CLTC	CUMP	AZUSA "ON",	1
	S	1-4	CLTC	CUMP	MISTRAM "ON".	
	S	1-5	CLTC	CUMP	ODOP "ON",	
	S	1-6	CLTC	врор	ODOP GROUND TRANSMITTER NO. 1 "ON".	
520° 30"						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1MP CUMP	PREFLIGHT CALIBRATION TO 100%.	
520 7						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1CM	TV TO "B*",	
	S	1-2	CLTC	C1MP	VERIFY EBW MEAS, "OFF".	
	S	1-3	CLTC	CLVN	DESTRUCT ENABLE "QN".	
	S	1-4	CLTC	C4DP	VERIFY SAFETY BUS INDICATION "ON".	
1	S	1-5	CLTC	R4VC	VISICORDER "ON".	
	S	1-6	CLTC	C10P	PERFORM MODULE CHECK.	

# SATURN/APOLLO

PAGE 168
TEST NO. 7-LSV I-300
VEHICLE SA-7 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-520'						
	S	1-7	CLTC	CiDP	S-I CDR #1 AND #2 "ON".	
	S	1-8	CLTC	C10P	CHECK S-I EBW #1 AND #2 VOLTAGES.	
	S	1-9	CLTC	CiDP	S=I CDR #1 AND #2 TO "INTERNAL".	
	SRO	1-10	CNTS	ORNG	INTERROGATE MISTRAM AND REPORT READ OUT	
	SRO	1-11	CNTS	ORNG	INTERROGATE AZUSA AND REPORT READ OUT.	
	RD1	S	RVFC	SVMP	VERIFY HYDRAULIC ACCUMULATOR PRECHARGE COMPLETE.	
	BL1	3	RVFC	ANAG	CLOSE S-I HYDRAULIC PUMP CIRCUIT BREAKERS.	
	RD1	4		C1FP	AFTER VERIFICATION THAT PRE-VALVE CONTROL VALVES HAVE BEEN RESTORED TO "NORMAL", CYCLE EACH PRE-VALVE FROM "OPEN" TO "CLOSE" TO "OPEN".	
	RD1	5		C1FP	AFTER VERIFICATION THAT FUEL VENT CONTROL LINE IS RETURNED TO "NORMAL", OPERATE FUEL VENT SWITCH TO "OPEN",	
		6		S1VM	LEAK CHECK FUEL VENT CONTROL LINE,	
		7		S1VM	INSTALL CAP ON FUEL TANK TEST SENSING LINE AT LOX TANK 4 AFTER FUEL VENTS ARE "OPEN".	
516' 30"						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	B1TM OTMD	8 KC OSCILLATORS "ON".	
	S	1-2	CLTC	CIPC	CALIBRATION SWITCH TO "AUTOMATIC" AND GIVE CAL, START.	
	S	1-3	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS.	
	S	1~4	CLTC	C1MP CUMP CIPC	TAPE RECORDERS TO "RECORD",	

PAGE TEST NO. VEHICLE

TIME	COMM.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
516' 30"						
	S	1=5	CLTC	C1MP CUMP	PREFLIGHT CALIBRATIONS TO 0%, 25%, 50%, 75%, AND 100% IN 2 SEC INCREMENTS.	
	S	1~6	CLTC	C1MP CUMP	RECORDER TRANSFERS "ON".	
	S	1-7	CLTC	C1MP CUMP	TELEMETER CALIBRATIONS TO "INFLIGHT".	
	S	1=8	CLTC	CUMP	P1 TELEMETER CALIBRATION COMMAND.	
	S	1-9	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS "ON" FOR 5 SECONDS.	
	S	1-10	CLTC	CIPC	GIVE TELEMETER CALIBRATION COMMAND.	
	S	1-11	CLTC	RBHM	S=I AND IU HIGH CALIBRATE COMMAND (15 SECONDS).	
	S	1-12	CLTC	RBHM	S-I AND IU LOW CALIBRATE COMMAND (15 SECONDS).	
	S	1-13	CLTC	RBHM	S=I AND IU RUN COMMAND,	
	S	1-14	CLTC	C1MP CUMP	RECORDER TRANSFERS "OFF".	
5151						
		1	CLTC	RGCC	INITIATE APS 103 (CHI-ALIGN),	
		2			RF AND TM CHECKS.	
	S	2-1	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS "ON" FOR 5 SEC.	
	S	2-2	CLTC	CUMP	P1 TELEMETER CALIBRATION COMMAND.	
	S	2=3	CLTC	C1MP CUMP CIPC	TAPE RECORDERS "OFF", (DELAY 10 SEC BEFORE NEXT STEP)	
	S	2=4	CLTC	C1MP CUMP		

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
·515 ¹						
				CIPC	TAPE RECORDERS TO "PLAYBACK".	
	S	2-5	CLTC	C1MP CUMP CIPC	CONFIRM TAPE RECORDERS READY INDICATIONS (BEFORE PROCEEDING).	
	S	2~6	CLTC	B1TM OTMD	8 KC OSCILLATORS "OFF".	
	S	2-7	CLTC	B1TM OTMD B4TM	TELEMETER RECORDINGS "OFF",	
	RD2	3	C4TC	CPTE	VERIFY COMPLETION OF S-IV PROPULSION PRELIMINARY PREPS PER DAC-P-7009, PART I.	
	RD2	4	C4TC	CPTE	VERIFY COMPLETION OF BULKHEAD VACUUM MONITOR PREPS PER DAC-P-7009, PART III.	
	RD2	5	C4TC	CPTE	VERIFY COMPLETION OF PROPELLANT LINES VACUUM CHECK PER DAC-P-1002, PART II.	
	RD2	6	C4TC	CPTE	PERFORM LO2 AND LH2 SYSTEM PURGES AND COMPONENTS CHECK PER DAC-P=7009, PART IN	,
		7		S1VM	VERIFY LOX VENT AND RELIEF VALVES "OPEN" AND THEN INSTALL LOX VENT LINES.	
	SRO	8	CNTS	ORNG	COMMAND CARRIER "ON".	
	S	9	CLTC	CUNP	HORIZON SENSOR POWER "ON".	
	S	10	CLTC	CUNP	HORIZON SENSOR "ON".	
	S	11	CLTC	RFSR	SEQUENCE RECORDERS TO 10 MM/SEC.	
	S	12	CLTC	RSEQ	SEQUENCE AND E+1 RECORDERS TO MINUTE SPEED AND TIME PULSE ON.	
	S	13	CLTC	RSEQ	VERIFY VOLTAGE RECORDERS TO FINE RANGE.	
	S	14	CLTC	SVOR	SEQUENCE RECORDERS TO MINUTE SPEED AND TIME PULSE ON.	
	S	15	CLTC	CDCG	CHECK VEHICLE GROUND POWER GENERATOR SETTINGS FOR POWER TRANSFER.	
	S	16	CLTC	B4TM	TELEMETER RECORDING ON.	

PAGE TEST NO. VEHICLE

TIME	COMM.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
5151						
	S	17	CLTC	C4DP	TEST S=IV EBW#S FOR READY	
	S	18	CLTC	C4DP	VERIFY S&A "SAFE",	
	S	19	CLTC	CSRP	VERIFY S-1 AND IU FLIGHT SEQUENCER ZERO INDICATIONS.	
	S	20	CLTC	C4DP	S=IV CDR NO. 1 AND NO. 2 "ON",	
	S	21	CLTC	C4DP	VERIFY S-IV EBW NO. 1 AND NO. 2 "OFF"	
	s	22	CLTC	CSPC	S=IV EBW EXTERNAL "ON".	
	S	23	CLTC	C4DP	S-IV CDR NO. 1 AND NO. 2 TO "INTERNAL".	
	S	24	CLTC	C40P	VERIFY ALL S-IV NOT READY INDICATIONS	
	S	25	CLTC	CUNC	COOLING TO PREFLIGHT.	
	S	26	CLTC	SUVN	VERIFY ST-124 BLOWER "ON",	
	S	27	CLTC	CUNC	COOLING TO "INFLIGHT".	
	S	28	CLTC	SUVN	VERIFY PRIMARY BLOWER ON (AFTER 40 SECONDS).	
	S	29	CLTC	RBHM	SANBORN RECORDER TO 25 MM/SEC.	
	SRO	30	CNTS	ORNG	CUTOFF COMMAND AND RELEASE AT TEST SUPERVISOR REQUEST.	
	s	31	CLTC	C1DP C4DP	VERIFY S=I AND S-IV EBW #1 AND #2 ON "INTERNAL" AND CHARGED.	
	S	32	CLTC	C4DP	S=IV EBW #1 AND #2 TO EXTERNAL AND VERIFY "OFF";	
	S	33	CLTC		S=I EBW #1 AND #2 TO EXTERNAL AND VERIFY "OFF".	
510 1						
	SC	1			VERIFY COMPLETION OF POWER TRANSFER TEST.	

PAGE VEHICLE

172 TEST NO. 7 = LSVI - 300 SA-7

TIME	COMM.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
510 '						
	S	2	CNTS	ORNG	VERIFY 0-18 RADAR AWAY FROM THE PAD (AFTER BEACON NO. 1 READOUT).	
	S	3	CSTC	CICM	SPACECRAFT RADAR BEACON NO. 1 "OFF".	
	S	4	CSTC	CICM	SPACECRAFT RADAR BEACON NO. 2 "ON".	
	S	5	CNTS	ORNG	ONE (1) MINUTE AFTER NOTIFICATION OF BEACON ON, BEGIN INTERROGATION AND PROCEED WITH READOUT, REPORT READOUT WHEN COMPLETE.	
	RD4	6		PRPF	SET UP RP=1 SYSTEM PER LV0-L-1031.	
	SRO	7	CNTS	ORNG	MONITOR ALL RF SYSTEMS AND REPORT ANY CHANGE IN INTERNAL POWER.	
	S	8	CLTC	B1TM OTMD	TELEMETER RECORDINGS "ON".	
	S	9	CLTC	CUNP	VERIFY IU READY FOR POWER TRANSFER ON $_{\scriptscriptstyle Q}$	
		10	CLTC	RVFC	HYDRAULIC PUMPS ENABLE "ON" .	
	S	11	CLTC	RVFC	S=I HYDRAULIC PUMPS ON.	
	S	12	CLTC	RGCC	INITIATE S=1 STEERING COMMANDS (APS=104	
	S	13	CLTC		PERSONNEL CONNECTED WITH S-IV POWER TRANSFER SWITCH TO S-IV CIRCUIT.	
	SIV	14	CATC	CIPC	INSTRUMENTATION SYSTEM TO "INTERNAL".	
	SIV	14-1	C4TC	CSPC	STAGE POWER SYSTEMS TO "INTERNAL".	
		14-2	C4TC	CSPC	READOUT BATTERIES.	
	SIV	14~3	C4TC	CIPC	PERFORM AUTOMATIC VCO CALIBRATION.	
	SIV	14-4	C4TC	CSPC CIPC B4TM	VERIFY SATISFACTORY ON INTERNAL POWER.	
	SIV	14-5	C4TC	CSPC	STAGE POWER SYSTEMS "EXTERNAL",	

## SATURN/APOLLO

PAGE VEHICLE

173 TEST NO. 7 - LSVI - 300 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-510 9						
	SIV	14-6	C4TC	CIPC	INSTRUMENTATION SYSTEMS "EXTERNAL" .	
	SIV	14-7	C4TC	C4DP	S-IV CDR'S #1 AND #2 "EXTERNAL" AND	
	SIV	14-8	C4TC	C4DP	SHIV EBWS #1 AND #2 "EXTERNAL" AND VERIFY "OFF",	
	SIV	14-9		C4TC	S=IV POWER TRANSFER PERSONNEL SWITCH TO S=CIRCUIT.	
	S	14-10		CATC	VERIFY S⇔IV POWER TRANSFER SATISFACTORY	
	S	15	CLTC	C1CM	CAMERA LIGHTS "ON" (HOLD ON).	
	S	16	CLTC	CINP	S=I STAGE POWER TEST #1 "ON",	
	S	17	CLTC	CSRP	NOTE S-I POWER TRANSFER OK.	
	S	18	CLTC	C1CM	CAMERA LIGHTS "OFF".	
	s	19	CLTC	CUNP	IU STAGE POWER TEST #1 "ON".	
	S	20	CLTC	CSRP	NOTE IU POWER TRANSFER OK.	
	S	21		CSRP	VERIFY SATISFACTORY S-I AND IU POWER TRANSFER.	
	S	22		CINP	S=I STAGE POWER TEST #1 "OFF".	
	S	23		CUNP	IU STAGE POWER TEST #1 "OFF".	
	S	24	CLTC		S-I CDR #1 AND #2 TO "EXTERNAL" AND "OFF".	
	S	25	CLTC	CSPC	S=IV EBW EXTERNAL "OFF".	
	S	26	CLTC	RBHM	SANBORN RECORDER TO 2 MM/SEC.	
	S	27	CLTC	R4VC	VISICORDER TO "OFF".	
	S	28	CLTC	RVFC	S=I HYDRAULIC PUMPS "OFF".	
		29	CLTC	RVFC	HYDRAULIC PUMPS ENABLE "OFF",	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
5101						
	S	30	CLTC	CUNP	HORIZON SENSOR "OFF".	
	S	31	CLTC	CUMP	AZUSA "OFF".	
	S	32	CLTC	ORNG	COMMAND CARRIER "OFF".	
	S	33	CLTC	CUMP	C-BAND BEACON (RADAR) "OFF".	
	S	34	CLTC	CUMP	ALTIMETER "OFF".	
	S	35	CLTC	CUMP	MISTRAM "OFF",	
	S	36	CLTC	CUMP	GUIDANCE COMMAND RECEIVER "OFF".	
	S	37	CLTC	CUMP	ODOP "OFF",	
	S	38	CLTC	C1CM	TV CAMERA "OFF".	
		39		CLTC	PERSONNEL CONNECTED WITH S-IV CONTROL CHECKS SWITCH TO YW2 CIRCUIT.	
	YW2	39-1		CATC	CLEAR S-IV ENGINE AREA OF UNNECESSARY PERSONNEL AND EQUIPMENT.	
	YW2	39~2		RVFC	STAGE SELECTOR SWITCH TO "S=IV".	
	YW2	40		RVFC	HYDRAULIC PUMPS ENABLE "ON".	
	YW2	40-1	RVFC	R4FC	S=IV HYDRAULIC PUMPS "ON".	
	YW2	40-2	RVFC	RGCC	INITIATE S-IV STEERING COMMANDS (APS-105)	
	YW2	40~3		RVFC	IU RATE GYRO TORQUER COMMANDS.	
	Yw2	40-4	RVFC	R4FC	"CLOSE" S-IV HYDRAULIC ACCUMULATOR VALVES.	
	YW2	40-5	RVFC	R4FC	VERIFY S-IV HYDRAULIC PUMPS "OFF",	
	YW2	40-6		RVFC	FLIGHT CONTROL SWITCH TO "S-I".	
	YW2	40-7		RVFC	CONTROL VOLTAGE "OFF".	
	YW2	40-8		RVFC	CONTROL COMPUTER "OFF".	
	YW2	40-9		RVFC	IU RATE GYROS "OFF".	
	YWZ	40-10		RVFC	CONTROL ACCELEROMETERS "OFF".	

PAGE VEHICLE

175 TEST NO. 7-LSVI-300 SA-7

STA.	DESCRIPTION	REMARKS
RVFC	S-I RATE GYROS #OFF#,	
RVFC	Q-BALL SYSTEM "OFF".	
RVFC	Q-BALL HEATERS WOFF".	
RVFC	ACCELERATION SWITCH "OFF".	
R1TM DTMD	TELEMETER RECORDING "OFF".	
UNP	HORIZON SENSOR POWER "OFF".	
CIMP	TELEMETERS F1, F2, F3, F2, S1 AND S2	
CIMP CUMP	TELEMETER AUXILIARY EQUIPMENT "OFF".	
1MP UMP	SWEEP FREQUENCY CALIBRATOR "OFF".	
CUMP	TELEMETERS F5, F6, P1, S3 AND MINITRACK "OFF".	
LVN	DESTRUCT ENABLE "OFF",	
RSEQ	SEQUENCE AND E+1 RECORDERS TO "HOUR SPEED" AND TIME PULSE "OFF".	
SVOR	SEQUENCE RECORDERS TO "HOUR SPEED" AND TIME PULSE "OFF".	
1MP UMP	TELEMETER CALIBRATION TO PREFLIGHT.	
IPC	VERIFY VCO CALIBRATION "OFF" AND "AUTO".	
RFSR	SEQUENCE RECORDERS TO 2 MM/SEC.	
CUNC	COOLING TO PREFLIGHT.	
RNG	REPORT ANY CHANGE OR DEVIATIONS OF RF EQUIPMENTS DURING INTERNAL RUN.	
		RNG REPORT ANY CHANGE OR DEVIATIONS OF RE

## SATURN/APOLLO

PAGE

176 TEST NO. 7=LSVI=300 VEHICLE SA=7 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
5051						
	YW2	1		RVFC	S=IV CONTROL CHECKS PERSONNEL SWITCH TO S=CIRCUIT.	
	S	2	CLTC	RVFC	VERIFY COMPLETION OF S-IV CONTROL CHECKS.	
		3		CIPC	TELEMETER D1. D2 AND D3 OFF AFTER COMPLETION OF S-IV CONTROL CHECKS.	
		4		THE RESERVE TO STATE OF THE PARTY OF THE PAR	TELEMETER RECORDING OFF AFTER S-IV TELEMETERS **OFF**	
		5		C4TC	OPEN SOIV ENGINE AREA FOR NORMAL ACCESS	
	YW1	6		RS4C	SECURE ST=124 PER PROCEDURE LVO-G-7040 AFTER CLEARANCE FROM RGCC AND RVFC.	
	YW1	7		RATC	SECURE AUTO THEODOLITE AFTER CLEARANCE FROM RS4C PER PROCEDURE LV0-G-7044.	
		8		CUNC	COOLING **OFF**,	
		9		BDOP	ODOP GROUND TRANSMITTER NO. 1 "OFF".	
500						
		1		RGCC	FLIGHT COMPUTER SYSTEM POWER "OFF",	
		2		SVMP	INSTALL DRAIN SCREW ACCESS COVERS ON OUTBOARD ENGINES.	
		3		SVMP	RECORD GEAR CASE PRESSURE FOR EACH ENGINE;  POSITION 1  POSITION 2  POSITION 3  POSITION 4  POSITION 5  POSITION 6  POSITION 7  POSITION 8	
		4		SVMP	REMOVE GEAR CASE MONITOR GAGES AND INSTALL AN 929-4C CAP ON PORT OF CROSS.	
		5		S1TM SUTM S4ME	CONNECT TELEMETER SYSTEMS TO VEHICLE TELEMETER ANTENNAS.	

PAGE TEST NO. VEHICLE

COMM. SEQUENCE COMMAND RESPONS	DESCRIPTION	REMARKS
RD4 1 PLOF	SET UP LOX SYSTEM PER PROCEDURE LV0-L-1002,	
SRO 1 CNTS ORNG	VERIFY 0-18 RADAR AWAY FROM THE PAD (AFTER BEACON #2 READOUT),	
SC 2 CSTC CICM	SPACECRAFT RADAR BEACON #1 "ON".	
SRO 3 CNTS ORNG	ONE (1) MINUTE AFTER NOTIFICATION OF BEACON "ON", BEGIN INTERROGATION OF BOT BEACONS AND PROCEED WITH READOUT, REPORT READOUT WHEN COMPLETE,	Н
PA 4 CNTS	ANNOUNCE: COUNTDOWN READOUTS WILL BE INTERRUPTED.	
5 CGCM	PERFORM LIFTOFF CHECKS WITH BDOP, ANEMOMETER RECORDERS, HANGAR D AND COUNTDOWN CLOCK.	
BL1 6 CGCM	PERFORM RCA SEQUENCER CHECKS,	
RD3 1 SGMP	PREPARE THE HOLDDOWN ARMS FOR LAUNCH PER PROCEDURE LVO-E-7016.	
RD1 2 C1TC S1VM	SECURE INSTRUMENT CONTAINER #2.	
RD1 3 CLTC SUVM	SECURE IU TUBE 4 FOR FLIGHT,	
CLTC S1TM SUTM S4AE	VERIFY TELEMETER SYSTEMS CONNECTED TO VEHICLE TELEMETER ANTENNAS.	
1 ANAG	CLOSE ENGINE HEATER CIRCUIT BREAKERS.	
RECS	START ECS TO S-IV ENGINE COMPARTMENT.	
3 CLTC C2MP	TELEMETER S1 AND S2 ON.	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-480°						
	S	4	CLTC	CUMP	TELEMETER S3 ON.	
	S	5	CLTC	C1PC	TELEMETER D2 QN.	
476						
	s	1	CLTC	B1TM B4TM	VERIFY RECEPTION OF TELEMETER SIGNALS S1: S2: S3 AND D2:	
	S	2	CLTC	C1MP	TELEMETER S1 AND S2 OFF.	
	S	3	CLTC	CUMP	TELEMETER S3 OFF.	
	S	4	CLTC	SIPC	TELEMETER D2 OFF.	
					NOTE: IU TUBE DOOR #3 CAN BE SECURED FOR FLIGHT AFTER COMPLETION OF STEPS AT T=476?	
					NOTE: S-I CANISTER 13 CAN BE SECURED FOR FLIGHT AFTER COMPLETION OF STEPS AT T=476°	
475						
	RD1	1			REQUEST S=I COMPONENT HEATERS ON AND PERFORM HEATER CHECKS PER PROCEDURE LVO=MV=P=1003.	
		2		SVMP	CHECK SETTING OF CONTROL REG (750 PSIG) AND THEN SECURE FOR LAUNCH.	
		3			INSTALL AN 929-4C CAP ON 750 PRESSURE MEASUREMENT CALIBRATION TEE AND SAFETY WIRE.	
		4			OPEN 750 OK PRESSURE SWITCH CALIBRATION HAND VALVE AND SAFETY WIRE "OPEN".	
	BR1	5	C1TC		VERIFY CONTROL SUPPLY AND REGULATOR MEASUREMENTS HAND VALVE OPEN AND SAFETY WIRED (FUEL BAY 3).	
		6		C1CM	BEGIN LOX TANK CAMERAS LENS PURGE.	

PAGE TEST NO. VEHICLE

TIME	COMM.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-475 †						
	BR1	7	C1TC	SSMS	VERIFY MEASURING READY TO SECURE FUEL BAYS.	
	BL1	8	CITC	SINA	VERIFY NETWORKS READY TO SECURE FUEL BAYS.	
470 "						
		1	C1TC	SVMP	INSTALL FUEL BAY DOORS 1, 2 AND 4 (REMOVE INTERNAL ACCESS LADDERS).	
	SRO	2	CNTS	ORNG	VERIFY RADAR AWAY FROM PAD.	
	sc	3	CSTC	CICM	SPACECRAFT RADAR BEACON NO. 1 "OFF".	
	SC	4	CSTC	CICM	SPACECRAFT RADAR BEACON NO. 2 "OFF".	
	SC	5	CSTC	CICM	SPACECRAFT TELEMETER A (LINK 1) "OFF".	
	sc	6	CSTC	CICM	SPACECRAFT TELEMETER B (LINK 6) "OFF".	
	SC	7	CSTC	CICM	SPACECRAFT TELEMETER C (LINK 14) "OFF".	
	PA	8		CNTS	RF SILENCE "ON".	
4651						
	BR1	1	C1TC	RFDM	VERIFY FIRE DETECTION CALIBRATION PER PROCEDURE 2-LSII-707	
	BR1	2	C1TC	RCSM	VERIFY CSM CALIBRATION PER PROCEDURE 2-LSII-706.	
		3	CITC		VERIFY REMOVAL OF STRIPPABLE COATING FROM FIN BASE.	
	sc	4	CSTC	CPWR	VERIFY POWER IS REMOVED FROM THE SPACECRAFT.	
	BL1	5	C1TC	SINF	BEGIN S-I RETROROCKET INITIATOR CONNECTION PER PROCEDURE 1-LSII-707.	
	BL1	6	CLTC	SUVN	BEGIN CONNECTION OF HORIZON SENSOR DOME THRUSTER SQUIB PER PROCEDURE 1-LIUI-302	1
		7		SGMP	SECURE DOOR ON ENGINE SERVICE PANEL AT NW PEDESTAL LEG OF LAUNCHER.	

PAGE TEST NO. VEHICLE

TIME	COMM.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
4651						
		8		CSTC	VERIFY SPACECRAFT BATTERIES DISCONNECTE	
	BL1	9	CNTS	ANAG	TURN BREAKER 66A5A10=1 "OFF" (U=BALL POWER BREAKER),	
4551						
	RD2	1	CATC	CPTE	VERIFY COMPLETION OF S-IV LOX AND LH2 SYSTEM PURGES AND COMPONENT CHECKS PER DAC-P-7009, PART IV.	
	RD2	2	C4TC	CPTE	PERFORM AUTOMATIC VALVE CONTROL FUNCTIONAL CHECK PER DAC-P=7009, PART V	
	S	3	CSTC	CNTS	VERIFY SPACE VEHICLE AND LAUNCH COMPLEX RF SILENCE AND Q=BALL HEATER BREAKER "OFF".	
	SC	4	CSTC	CPYE	CONNECT SPACECRAFT LES IGNITER PER PROCEDURE C=10001.	
450						
		1		SUVM	INSTALL HORIZON SENSOR COVER,	
435						
	BLi	1	C1TC	SINF	VERIFY COMPLETION OF RETROPOCKET INITIATOR CONNECTION.	
430						
	RD1	1	CITC	SVMP	VERIFY COMPLETION OF HYPERGOL INSTALLATION.	
	RD3	2	SVMP	RECS	BEGIN ECS AIR TO S-I ENGINE COMPARTMENT	
	BR1	3	C1TC	SVME	END MANUAL PREFLIGHT PHOTOGRAPHIC INSTRUMENTATION.	
	BR1	4	C1TC	SVME	REMOVE PREFLIGHT MOD A EJECT TUBE COVERS.	

## EMERGENCY PROCEDURE (T-410' TO T-0')

(AFTER S-I LOX TANKING)

### PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-425 9						
		1	PPSO	ORNG	BRING UP 4 PUMPS TOTAL ON 36-INCH WATER SYSTEM (175 PSI).	
		2		PPSO	ESTABLISH COMPLEX ROADBLOCKS ( ACCESS TO PAD AFTER THIS TIME BY ACCESS LIST).	
		3		C1FP	VERIFY FUEL LOADING PANEL FUEL BUBBLING ON THEN OPERATE FUEL BUBBLING SWITCH ON	
	RD2	4	C4TC	CPTE	VERIFY COMPLETION OF AUTOMATIC VALVE CONTROL FUNCTIONAL CHECK, DAC-P-7009, PART V, INCLUDING SECURING OPERATIONS.	
	sc	5	CSTC	CPYR	VERIFY COMPLETION OF SPACECRAFT LES IGNITER CONNECTION.	
	PA	6		CNTS	ALL PERSONNEL CLEAR PAD AREA EXCEPT THOSE PERSONNEL SPECIFICALLY LISTED ON ACCESS. PERSONNEL ON ACCESS LIST REPORT TO STANDBY POSITIONS.	
-420°						
		1		RWCP	ARM WATER SYSTEM.	
	BL1	2	CNTS	ANAG	TURN BREAKER 66A5A10~1 "ON" (Q-BALL POWER BREAKER),	
	SC	3	CNTS	CSTC	VERIFY SPACECRAFT PERSONNEL ARE CLEARING THE PAD.	K
	SC	4	CSTC	CPWR	APPLY SPACECRAFT POWER PER PROCEDURE C-0007.	
	YW3	5	CLTC	BSOC	VERIFY CONSTANTS AND SCAN RATE OF ONE MINUTE INSERTED INTO PROPELLANT SCAN PROGRAM.	
-410°					500-5 NA 1440	
	RD4	1		PPSO	VERIFY AREA CLEAR FOR S-I LOX LOADING.	
	YW3	5	CLTC		INITI/TE PROPELLANT LOADING SCAN PROGRAM.	
	RD4	3	CLTC		LOAD S-I LOX TO 18 PERCENT PER PROCEDURE LVO-L-1004.	

### EMERGENCY PROCEDURE (T-110' to T-0')

(AFTER S-I LOX TANKING)

### PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

TEST NO. 7=LSVI-300
VEHICLE SA-7 182 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
4101						
					NOTE	
					IN EVENT OF ACCIDENTAL LOX	
					PREVALVE CLOSURE, REFER TO EMERGENCY PROCEDURE.	
3951						
		1		7.00 <b>.0</b> 0.00.00	S=I LOX LOADING CREW RETURN TO SERVICE STRUCTURE AND PEDESTAL (AFTER TANK PRESSURE IS BELOW 2 PSI).	
	PA	2		CNTS	RF SILENCE "OFF"	
390'						
		1			BEGIN S-I LOX LEAK CHECKS PER PROCEDURE LVO-MV-P-1007.	
	PA	2			AFTER VERIFICATION OF NO GROSS LEAKS	
				M. F. S. C. S.	ON S-I, ANNOUNCE CLEARANCE FOR OPERATIONAL PERSONNEL TO RETURN TO THE	
					PAD AND VEHICLE (BY ACCESS LIST).	
3851						
	BR1	1	C1TC	SCMC	VERIFY MEASURING READY TO SECURE LOX	
	DNI	1	CIT	33113	BAYS.	
	BL1	2	C1TC	The second secon	VERIFY NETWORKS READY TO SECURE LOX	
2227					BAYS.	
3801						
	RD4	1	C1TC	RLOO	VERIFY S=! LOX LOADED TO 18 PERCENT.	
		2		SVMP	AFTER LOX LEAK CHECK IS COMPLETE REMOVE	
					INTERNAL ACCESS LADDERS, SECURE LOX BAYS: #1, 2, 3 AND 4 DOOR,	
		3		- Secretary	AFTER LEAK CHECKING (LEAK TEC) LOX	
					DIFFERENTIAL STEP PRESSURE SWITCH SENSING LINE, REMOVE INTERNAL ACCESS LADDER, SECURE FUEL BAY 3 DOOR.	
	RD3	4			BEGIN PREPARATIONS OF THE UMBILICAL SWING ARMS FOR LAUNCH PER PROCEDURE	

## EMERGENCY PROCEDURE (T-410' TO T-0')

(AFTER S-I LOX TANKING)

### PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
3801						
					LV0-E-7016.	
	RD3	5		UQBL	BEGIN PREPARATIONS OF Q-BALL COVER RETRACT SYSTEM PER PROCEDURE LVO-E=7019	
375 1					RETRACT STSTEM FER PROCEDURE LVO-E-7019	
		1		SVMP	INSTALL CENTER BARREL COVER.	
		2		ANAG	INSTALL JUMPER FOR CDR LIFTOFF (84A1J17/LITTLE J TO 84A1J20/BIG F),	
		3		SVOR RSEQ	SEQUENCE RECORDERS TO MINUTE SPEED AND TIME PULSE "ON".	
	PA	4		CNTS	ANNOUNCEMENT = ALL PERSONNEL CONNECTED WITH COMMAND CHECKS, MAN APPOINTED STATIONS (NORCOM S CIRCUIT),	
	s	5		C1MP	TELEMETER F2 "ON",	
	S	6		CIMP	AUXILIARY EQUIPMENT "ON"	
	S	7		C1MP	TM CALIBRATION TO INFLIGHT,	
370						
	BL2	1	CATC	SMTE	PERFORM Salv AFT ACCESS KIT REMOVAL PER DAC-L-7002,	
		2			COMMAND CHECKS.	
	S	2~1	CNTS	S40E	VERIFY MODEL 243 PULSE SENSORS	
		S.T	CNIS	3402	CONNECTED TO 410A14P2 AND 410A19P2,	
	S	2-2	CNTS	S1NF S40E	VERIFY INERT DESTRUCT SEA UNITS ARE ELECTRICALLY CONNECTED.	
	S	2-3		CNTS	VERIFY PSO ON HAND TO OBSERVE DESTRUCT CHECKS ON VEHICLE AND IN BLOCKHOUSE.	
	S	2-4	CNTS	CSPC	VERIFY 400 CYCLE POWER "ON".	
	S	2~5	CNTS	C4DP	VERIFY MODULE POWER SUPPLY "ON" AND CHECK FOR 5 PLUS OR MINUS 0,2 VOLTS.	

## EMERGENCY PROCEDURE (T-410' TO T-0')

(AFTER S-I LOX TANKING)

### PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

# SATURN/APOLLO

PAGE

184 TEST NO. 7→LSVI-300 VEHICLE SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
370 *						
	S	2-6	CNTS	C10P	VERIFY MODULE POWER SUPPLY "ON".	
	S	2-7	CNTS	C4DP	TEST S=IV EBWIS FOR READY.	
	S	2-8	CNTS	C4DP	VERIFY S-IV S&A SAFE.	
	s	2-9	CNTS	SVOC	VERIFY PULSE SENSOR POWER "ON",	
	S	2-10	CNTS	CSSP	VERIFY MSFC TALKBACK ENABLE "ON".	
	S	2-11	CNTS	CLVN	DESTRUCT ENABLE "ON".	
	S	2-12	CNTS	C4DP	VERIFY SAFETY BUS INDICATION "ON",	
	S	2-13	CNTS	CIPC	TELEMETER D2 "ON",	
	S	2-14	CNTS	В4ТМ	VERIFY RASTER.	
	S	2~15	CNTS	B4TM OTMD	TELEMETER RECORDERS "ON".	
	S	2-16	CNTS	R4VC	VERIFY VISICORDER "ON",	
	S	2-17	CNTS	RFSR	SEQUENCE RECORDERS TO 10 MM/SEC.	
	S	2-18	CNTS	C10P	FUNCTION SELECTOR TO FIRING UNIT VOLTAGE CHECK.	
	S	2~19	CNTS	ORNG	COMMAND TRANSMITTER NO. 1 "ON",	
	S	5-50	CNTS	C40P	VERIFY PANEL SELECT TO AUXILIARY RECORDER.	
	S	2-21	CNTS	C1DP	S=1 CDR:S #1 AND #2 HONH,	
	S	5-55	CNTS	RHBM	SANBORN RECORDER TO 5 MM/SEC.	
	S	2-23	CNTS	C1DP	S-I CDR:S #1 AND #2 TO "INTERNAL".	
	S	2-24	CNTS	C4DP	S-IV CDR S #1 AND #2 "ON"	
	S	2~25	CNTS	C4DP	VERIFY S-IV EBW S #1 AND #2 "OFF" AND "NOT READY".	
	S	2-26	CNTS	C4DP	S-IV CDR'S #1 AND #2 TO "INTERNAL",	
	S	2~27	CNTS	RВНМ	SANBORN RECORDER TO 25 MM/SEC.	
	S	2-28	CNTS	ORNG	CUTOFF COMMAND AND RELEASE UPON REQUES!	r

## EMERGENCY PROCEDURE (T-410' TO T-0')

(AFTER S-I LOX TANKING)

### PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

7=LSVI=300 SA=7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
370						
	S	2-29	CNTS	C1DP C4DP	VERIFY CDR'S #1 AND #2 CUTOFF.	
	s	2-30	CNTS	C1DP C4DP	VERIFY S-I AND S-IV EBW'S #1 AND #2 ON INTERNAL AND CHARGED.	
	S	2-31		B4TM RFSR	NOTE ALL ENGINE CUTOFF NOT RECEIVED.	
	S	2-32		RFSR	NOTE HELIUM HEATER CUTOFF NOT RECEIVED.	
	S	2-33	CNTS	CPSO	VERIFY SEI AND S-IV EBWSS #1 AND #2 VOLTAGE INDICATIONS.	
	S	2-34	CNTS	ORNG	DESTRUCT COMMAND.	
	S	2-35	CNTS	SVOC	VERIFY S-1 AND S-IV DESTRUCT DOES NOT OCCUR ON VEHICLE.	
	S	2-35	CNTS	ORNG	RELEASE DESTRUCT COMMAND.	
	S	2-37	CNTS	RBHM	SANBORN RECORDER TO 5 MM/SEC.	
	s	2~38	CNTS	C1DP C4DP	S=I AND S=IV EBW°S #1 AND #2 TO "EXTERNAL".	
	S	2-39	CNTS	CiDP	VERIFY SOI EBW9S #1 AND #2 VOLTAGES "OFF".	
	s	2-40	CNTS	C4DP	VERIFY SHIV EBWIS #1 AND #2 "OFF" AND NOT READY.	
	S	2-41	CNTS	CIDP	ARM S-I DESTRUCT S&A (INERT).	
	S	2~42	CNTS	C4DP	ARM S-IV DESTRUCT S&A (INERT).	
	S	2-43	CNTS	CLVN	LIFTOFF ENABLE "ON".	
	S	2-44	CNTS	C1DP C4DP	SIMULATED LIFTOFF "ON".	
	S	2-45	CNTS	RBHM	SANBORN RECORDER TO 25 MM/SEC.	
	S	2-46	CNTS	ORNG	CUTOFF COMMAND AND RELEASE UPON REQUEST	
	S	2-47	CNTS	C1DP C4DP	VERIFY S=I AND S-IV EBW/S #1 AND #2 ON "INTERNAL" AND "CHARGED".	

## EMERGENCY PROCEDURE (T-410' TO T-0')

(AFTER S-I LOX TANKING)

### PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
5701						
	s	2-46	CNTS	CPSU	VERIFY S=1 AND S-IV EBWIS #1 AND #2 VOLTAGE INDICATIONS.	
	S	2-49		CSRP	NOTE S-I ENGINES CUTOFF DOES NOT OCCUR.	
	S	2-50		RFSR B4TM	NOTE ALL ENGINE CUTOFF COMMAND. CUTOFF COMMAND.	
	S	2-51	CNTS	RFSH	VERIFY HELIUM HEATER THERMAL CUTOFF COMMAND INDICATION.	
	S	2-52	CNTS	ORNG	DESTRUCT COMMAND.	
	S	2=53	CNTS	SVOC	VERIFY S-I AND S-IV DESTRUCT ON VEHICLE	
	S	2-54	CNTS	ORNG	RELEASE DESTRUCT COMMAND.	
	S	2-55	CNTS	RBHM	SANBORN RECORDER TO 5 MM/SEC.	
	S	2-56	CNTS	ORNG	SWITCH TO COMMAND TRANSMITTER #2.	
	S	2-57	CNTS	SVOC	RESET PULSE SENSORS,	
	S	2=58	CNTS	C1DP C4DP	S=I AND S=IV EBW'S #1 AND #2 TO "EXTERNAL"	
	S	2-59	CNTS	RFSR B4TM	VERIFY ALL ENGINE CUTOFF COMMAND REMOVED.	
	S	2-60	CNTS	C1DP C4DP	VERIFY S=I AND S-IV EBW'S #1 AND #2 "OFF" AND "NOT READY".	
	S	2-61	CNTS	CSPC	SEQUENCER RELAY POWER "OFF", THEN "ON",	
	S	2-62	CNTS	RFSR	VERIFY LOSS OF HELIUM HEATER CUTOFF COMMAND INDICATION.	
	S	2-63	CNTS	RBHM	SANBORN RECORDER TO 25 MM/SEC.	
	S	2-64	CNTS	ORNG	CUTOFF COMMAND AND RELEASE UPON REQUEST	
	S	2-65	CNTS	C1DP C4DP	VERIFY S-I AND S-IV EBW'S #1 AND #2 ON "INTERNAL" AND "CHARGED".	

## EMERGENCY PROCEDURE (T-410' TO T-0')

(AFTER S-I LOX TANKING)

#### PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
370 1						
	S	2-66	CNTS	CPSO	VERIFY S-I AND S-IV EBW S #1 AND #2 VOLTAGE INDICATIONS.	
	S	2-67		CSRP	NOTE S-I ENGINES CUTOFF DOES NOT OCCUR.	
	S	2~68	CNTS	RFSR B4TM	VERIFY ALL ENGINE CUTOFF COMMAND.	
	S	2-69	CNTS	RFSR	VERIFY HELIUM HEATER CUTOFF COMMAND INDICATION.	
	S	2-70	CNTS	ORNG	DESTRUCT COMMAND.	
	S	2-71	CNTS	SVOC	VERIFY Sal AND S-IV DESTRUCT ON VEHICLE	
	S	2~72	CNTS	ORNG	RELEASE DESTRUCT COMMAND.	
	S	2-73	CNTS	RBHM	SANBORN RECORDER TO 5 MM/SEC.	
	S	2-74	CNTS	SVOC	RESET PULSE SENSORS.	
	S	2-75	CNTS	C1DP C4DP	SIMULATED LIFTOFFS "OFF",	
	S	2-76	CNTS	RFSR B4TM	VERIFY ALL ENGINE CUTOFF REMOVED.	
	S	2-77	CNTS	C1DP C4DP	S=I AND S=IV EBW S #1 AND #2 TO	
	S	2-78	CNTS	C1DP C4DP	VERIFY S-I AND S-IV EBW S #1 AND #2	
	S	2~79	CNTS	CSPC	SEQUENCER RELAY POWER "OFF", THEN "ON",	
	S	2-80	CNTS	RFSR	VERIFY LOSS OF HELIUM HEATER CUTOFF COMMAND INDICATION.	
	S	2-81	CNTS	RBHM	SANBORN RECORDER TO 25 MM/SEC.	
	S	2-82	CNTS	ORNG	CUTOFF COMMAND AND RELEASE UPON REQUEST	
	S	2-83	CNTS	C1DP C4DP	VERIFY S-I AND S-IV EBW S #1 AND #2 TO "INTERNAL" AND "CHARGED".	
	S	2-84		RFSR B4TM	NOTE ALL ENGINE CUTOFF NOT RECEIVED:	

## EMERGENCY PROCEDURE (T-410' TO T-0')

(AFTER S-I LOX TANKING)

#### PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-370 1						
	S	2-85	CNTS	RFSR	NOTE HELIUM HEATER CUTOFF COMMAND NOT RECEIVED.	
	S	2~86	CNTS	CPSU	VERIFY S=1 AND S-1V EBW S #1 AND #2 VOLTAGE INDICATIONS.	
	S	2~87	CNTS	ORNG	DESTRUCT COMMAND.	
	S	2-88	CNTS	SVOC	VERIFY S=I AND S-IV DESTRUCT DOES NOT OCCUR ON VEHICLE.	
	S	2-89	CNTS	ORNG	RELEASE DESTRUCT COMMAND.	
	S	2-90	CNTS	квни	SANBORN RECORDER TO 5 MM/SEC.	
	S	2-91	CNTS	C4DP	S=IV CDR SWITCHES #1 AND #2 "OFF".	
	S	2-92	CNTS	C4DP	VERIFY S-IV CDR 0S #1 AND #2 REMAIN "ON"	
	S	2~93	CNTS	ORNG	SAFE COMMAND AND RELEASE UPON REQUEST,	
	S	2=94	CNTS	C4DP	VERIFY SAFE COMMAND RECEIVED.	
	S	2~95	CNTS	C4DP B4TM	VERIFY SHIV CDRIS #1 AND #2 AND EBWIS #1 AND #2 REMAIN "ON".	
	S	2-96	CNTS	C1DP C4DP	S=I AND S=IV EBW S #1 AND #2 TO "EXTERNAL",	
	s	2~97	CNTS	C1DP C4DP	VERIFY S-I AND S-IV EBW'S #1 AND #2	
	S	2-98	CNTS	RBHM	SANBORN RECORDER TO 2 MM/SEC.	
	S	2-99	CNTS	C1PC	TELEMETER D2 "OFF".	
	s	2-100	CNTS	B4TM	TELEMETER RECORDING "OFF".	
	S	2-101	CNTS	CIMP	TELEMETER F2 "OFF",	
	S	2-102	CNTS	CIMP	AUXILIARY EQUIPMENT "OFF",	
	S	2-103	CNTS	ОТМИ	TELEMETER RECORDING "OFF",	
	S	2-104	CNTS	C1MP CUMP	TELEMETER CAL TO **PREFLIGHT**.	

## EMERGENCY PROCEDURE (T-410' TO T-0')

(AFTER S-I LOX TANKING)

#### PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
~370'						
	S	2-105	CNTS	C1DP	SAFE S-I DESTRUCT S&A (INERT).	
	S	2-106	CNIS	C4DP	SAFE S-IV DESTRUCT S&A (INERT).	
	s	2-107	CNTS	CLVN	LIFTOFF ENABLE "OFF".	
	S	2-108	CNTS	C4DP	S-IV CDR'S #1 AND #2 TO "EXTERNAL" AND VERIFY "OFF".	
	S	2-109	CNTS	C1DP	S=I CDR'S #1 AND #2 TO "EXTERNAL",	
	S	2-110	CNTS	C1DP	S-I CDR'S #1 AND #2 "OFF".	
	S	2-111	CNTS	ORNG	COMMAND CARRIER "OFF".	
	S	2-112	CNTS	CLVN	DESTRUCT ENABLE "OFF",	
	S	2-113	CNTS	C4DP	VERIFY LOSS OF SAFETY BUS ON INDICATION	
	S	2-114	CNTS	RFSR	SEQUENCE RECORDERS TO 2 MM/SEC.	
	S	2-115	CNTS	R4VC	VISICORDER "OFF".	
	S	2-116	CNTS	CLVN	REMOVE DESTRUCT AND LIFTOFF ENABLE SWITCH KEYS AND GIVE TO THE PAD SAFETY REPRESENTATIVE IN THE BLOCKHOUSE.	
	S	2~117	CNTS	ANAG	REMOVE JUMPER 84A1J17/LITTLE J TO 84A1J20/BIG F.	
	S	2-118	CNTS		SEQUENCE RECORDERS TO HOUR SPEED, TIME PULSE "OFF",	
	S	3		CNTS	VERIFY COMMAND CHECKS COMPLETE.	
		4			REMOVE PRELAUNCH COVERS FROM STROBE LIGHTS AND FIBEROPTIC LENSES AFTER S-IV AFT ACCESS KIT REMOVAL,	
3651						
		1			REMOVE LOX VENT LINES (DO NOT REMOVE VENT II-III UNTIL IMMEDIATELY AFTER S=I S&A CONNECTION).	
		2			INSTALL HEAT SHEILD PANELS 30M03497 (FOUR EACH).	

## EMERGENCY PROCEDURE (T-410' TO T-0')

(AFTER S-I LOX TANKING)

#### PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

190 7=LSVI=300 SA=7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
3651						
		3		SVMP	INSTALL ESCAPE CHUTE COVER AND CONNECT CALORIMETER PURGE LINE.	
		4		SiVM	STOW WORK PLATFORMS ON ADJUSTABLE ONE.	
		5	CITC	SVME	VERIFY ESCAPE CHUTE AND HEAT SHEILD PANEL MEASUREMENTS ARE CONNECTED AND PROTECTIVE COVERS REMOVED:	
3601						
		1	CLTC	RGCC	REQUEST APS 100 FLIGHT COMPUTER SYSTEM POWER "ON",	
		2		CINP	VERIFY LCC ESE PER PROCEDURE 1-LLVI-706	
		3		SVOR	SEQUENCE RECORDERS "OFF", TIME PULSE	
		4		CPSP	AFTER ROGER FROM SVOC, TURN SERVICE STRUCTURE DAT POWER "OFF",	
355 °						
		1		SVMP	VERIFY ALL BAY DOORS SECURED FOR LAUNCH	
		5		PLOF	TOP OFF LN2 150-GALLON TANK.	
		3		SGMP	VERIFY ALL LAUNCHER AUXILIARY PLATFORMS ARE REMOVED AND SECURED FOR LAUNCH,	
	SC	4		CSTC	VERIFY FLIGHT BATTERIES CONNECTIONS COMPLETE.	
350'						
	PA	1.		CNTS	RF SILENCE "ON" AND CONTROLLED SWITCHING.	
		2		SGNP	AFTER ROGER FROM SVOC, C4TC, BCDC AND SCTC, DISCONNECT SERVICE STRUCTURE TEST CABLES,	
		3		SGMP	SECURE TEST CABLE AND LOX REPLENISH ACCESS COVER ON SOUTH SIDE OF LAUNCHER.	

## EMERGENCY PROCEDURE (T-410' TO T-0')

(AFTER S-I LOX TANKING)

#### PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
3501						
		4		SUVN S1NF	DISCONNECT PULSE SENSORS AND ASSOCIATED CABLING FOR S=I AND S-IV DESTRUCT.	
		5		SVOC	SECURE 136 FOOT LEVEL DAT ROOM.	
	BL1	6	C1TC	SINF	CONNECT S=I DESTRUCT S&A UNIT PER PROCEDURE 1-LSII-710, PART II.	
	BL1	7	C1TC	S1NF	CONNECT S=1 DESTRUCT DETONATORS PER PROCEDURE 1-LSII-708.	
	RD2	8	C4TC	The model of the control of	CONNECT S-IV DESTRUCT S&A PER PROCEDURE DAC-0-7012.	
		9		SUMP	CHARGE HYDRAULIC ACCUMULATORS TO 1600 PSIG, LEAK CHECK SHRADER FITTING AND SECURE FOR LAUNCH: POSITION 1 POSITION 2 POSITION 3 POSITION 4	
		10		SVMP	INSTALL AN 929-12C CAPS ON HYDRAULIC SYSTEM LOW PRESSURE RELIEF VALVES:  POSITION 1  POSITION 2  POSITION 3  POSITION 4	
		11		SVMP	INSTALL FINAL HEAT SHIELD PANELS.	
	SIV	12	C4TC	S4ME	PREPARE S⇔IV AREAS FOR SERVICE STRUCTURE REMOVAL,	
		13			VERIFY ALL LOOSE ITEMS ON ADJUSTABLE LEVELS 1, 2 AND 3 REMOVED OR MOVED BACK AND TIED DOWN SECURELY,	
		14		W 25	VERIFY ALL ELECTRICAL, MECHANICAL AND PNEUMATIC CONNECTIONS BETWEEN FIXED AND MOVEABLE PLATFORMS ON ADJUSTABLE LEVELS 1, 2 AND 3 DISCONNECTED.	
		15	C4TC	PERSONAL PROPERTY.	VERIFY EJECT PNEUMATIC LINES CONNECTED TO EJECT MECHANISMS ON SWING ARM 2 AND GH2 VENT COUPLING.	
		16		0.9 523	REMOVE SAFETY WIRE FROM SWING ARM 2 AND GH2 VENT COUPLING EJECT	

## EMERGENCY PROCEDURE (T-410' TO T-0')

(AFTER S-I LOX TANKING)

#### PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-3501						
					MECHANISMS.	
		17		S4ME	REPLACE STEEL SHEAR PINS WITH ALUMINUM SHEAR PINS ON SWING ARM 2 EJECT MECHANISM.	
		18			REMOVE SAFETY BLOCK FROM GH2 VENT COUPLING EJECT MECHANISM.	
		19		S4ME	VERIFY SWING ARM 2 AND GH2 VENT COUPLING SOLENOIDS ELECTRICALLY CONNECTED.	
		20		S4ME	CONNECT BUNGEE LANYARD TO GH2 COUPLING.	
		21		S4ME	CONNECT RETRACT LANYARDS TO SWING ARM 2 AND GH2 VENT COUPLING CARRIERS.	
	BL1	55	CATC	CSPC	SERVICE TOWER POWER ENABLE "OFF".	
		23		RECS	START ECS AIR TO THE S-IV AFT SECTION,	
-3451						
	RD3	1		\$500 M. N. SHENN	PREPARE THE PNEUMATIC DISTRIBUTION SYSTEM FOR LAUNCH PER PROCEDURE LVO-E-7020.	
	RD3	2	C1TC		COMPLETE SWING ARM PREPARATIONS PER LVO-E-7016.	
	RD3	3	C1TC	100000	COMPLETE Q=BALL RETRACT SYSTEM PREPARATIONS PER LVO-E-7019.	
		4		SVMP	REMOVE BAY EXTERNAL ACCESS LADDERS.	
340'						
	YW4	1			OPEN SERVICE STRUCTURE SILO C-1 AND RETRACT PLATFORM #4.	
-335'						
	YW3	1	CLTC	RGCC	INITIATE APS 101, PSEUDO FLIGHT TEST.	
		2			REMOVE PROTECTIVE COVERS FROM TEMP. MEASUREMENTS ON BLAST PLATES AND LOX/SOX SYSTEM.	

## EMERGENCY PROCEDURE (T-410' TO T-0')

(AFTER S-I LOX TANKING)

#### PREMATURE CLOSURE OF S-I PREVALVES

- 1. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened, and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 2. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained, and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period, and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

LAUNCH COUNTDOWN
DATE SEPTEMBER 5, 1964
REVISION

# SATURN/APOLLO LAUNCH OPERATIONS

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCR	IPTION	REMARKS
3351							
		3		CSTC	VERIFY COMMAND MODU	ILE INSPECTION	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
330'						
	BL1	1	C1TC	S1NA	CONNECT S.P.G.G. INITIATORS PER PROCEDURE 1-LSII-709, PART II.	
					NOTE	
					IF CUTOFF IS REQUIRED BECAUSE OF AN ACCIDENTAL IGNITION, HOLD THE CUTOFF BUTTON UNTIL PREVALVES 1 THRU 8 ARE "OFF".	
		2		SVMP	INSTALL INITIATOR ACCESS HAND HOLE COVERS AS INITIATOR CONNECTIONS ARE COMPLETED.	
	8L1	3	C1TC	S1NF	VERIFY S=I DESTRUCT S&A UNIT AND DETONATORS CONNECTED.	
	RD2	4	C4TC	CVTE	VERIFY S-IV SEA CONNECTED FOR FLIGHT.	
		5	CVTE	S4ME	REMOVE THE REMAINING 2 LH2 TANK PROTECTIVE PADS FROM THE FORWARD INTERSTAGE.	
		6	CVTE	SUVM	REMOVE IU/S-IV FORWARD ACCESS KIT DOOR PAN.	
		7		CVTE	VERIFY S-IV FORWARD INTERSTAGE HAS BEEN INSPECTED FOR FLIGHT CONFIGURATION	
		8			INSTALL AND SECURE THE 5861 S-IV FORWARD INTERSTAGE DOOR.	
		9		S1VM	REMOVE LOX TANK CAMERA PURGE VENT LINE.	
		10		And the second second	REMOVE REMAINING LOX VENT LINE BETWEEN II=III AFTER S-I S&A CONNECTION IS COMPLETED.	
		11		1300,000 100	VERIFY ECS AIR ON IN THE S=IV AFT SECTION. OPEN ENGINE INJECTOR PURGE HAND VALVE AND LOX VENT PILOT HAND VALVE AND SAFETY WIRE BOTH IN OPEN POSITION.	
	BL2	12	C4TC	SMTE	VERIFY COMPLETION OF S-IV AFT ACCESS KIT REMOVAL PER PROCEDURE DAC-L-7002.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE

195 TEST NO. 7-LSVI-300 VEHICLE SA-7 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
3251						
		1		SVMP	SECURE GN2 PANELS ON ENGINE SERVICE PLATFORM.	
		2		SGMP	SECURE AUXILIARY GN2 CONNECTIONS INSIDE LAUNCHER RING FOR LAUNCH.	
315°						
	8L1	1	RS4C	ANAG	ST-124 HEATER SWITCH TO NORMAL POSITION.	
310 '						
	S	1	CLTC	SVME	VERIFY MEASURING READY TO OPEN SILO 8-3	
	RD2	2	C4TC	CVTE	VERIFY COMPLETION OF S-IV DESTRUCT BLOCK CONNECTION PER PROCEDURE DAC-0=7012.	
	SIV	3		S4ME	VERIFY S-IV AREAS READY FOR SERVICE STRUCTURE REMOVAL.	
3051						
	BL1	1		UGRN	SECURE UMBILICAL TOWER ESE PER PROCEDURE 1-LLVI-715.	
	YW4	2	CLTC	SSSC SMTE	OPEN SERVICE STRUCTURE SILO 8-3 AND RETRACT PLATFORMS #2 AND #3,	
	SC	3		CSTC	BEGIN SPACECRAFT ECS PURGE.	
295						
		1		SGMP	BEGIN PREPARATIONS TO SECURE ENGINE SERVICE PLATFORM BENEATH LAUNCHER PER PROCEDURE LVO-E=7020.	
290 0						
	BL1	1	C1TC	S1NA	VERIFY S.P.G.G. INITIATOR CONNECTION COMPLETE.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARK
2901						
		5		SVMP	FINAL S-I TAIL SECURING OPERATIONS.	
2851						
		1		C1TC RBHM	PERFORM LOX BUBBLING CHECK.	
		2		SVMP	COMPLETE INITIATOR ACCESS HAND HOLE COVER INSTALLATION,	
		3		C1FP SVMP	INITIATE CALORIMETER PURGE AND VERIFY FLOW AT ALL 10 LOCATIONS.	
280 1						
	S	1	CLTC	SGNP	VERIFY ALL NETWORK CABLES ARE CLEAR FOR STRUCTURE REMOVAL.	
2751						
	BL1	1		SGNP	SECURE LAUNCHER ESE PER PROCEDURE 1-LLVI-714.	
	S	2	CLTC	RWCP	VERIFY LAUNCH VALVE "OPEN".	
2701						
		1		SVMP	COMPLETE S=1 TAIL SECURING OPERATIONS.	
	RD3	2	CITC	SGMP	SECURE ENGINE SERVICE PLATFORM FOR LAUNCH PER PROCEDURE LVO-E-7021	
		3	CLTC	SGMP	OBTAIN PPSO VERIFICATION THAT INSIDE OF LAUNCHER IS CLEAR OF ALL PERSONNEL AND SECURE LAUNCHER BLAST DOOR AND WINDOW FOR LAUNCH.	
	SC	4		CSTC	VERIFY COMPLETION OF SPACECRAFT HATCH	
	sc	5		CSTC	VERIFY COMPLETION OF SPACECRAFT ECS PURGE.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

# SATURN/APOLLO

PAGE 197
TEST NO. 7-LSVI-300
VEHICLE SA-7 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
2651						
	sc	1		CSTC	VERIFY ALL SPACECRAFT PERSONNEL ARE CLEARING THE PAD AREA.	
	AM2	2	CLTC	SSSC	OPEN SERVICE STRUCTURE SILO C-2 AND RETRACT PLATFORMS NO. 5 AND 6. NOTE SPACECRAFT PAD LEADER WILL REMAIN UNTIL SILO CLEARS SPACECRAFT.	
	PA	3		CNTS	ANNOUNCE TO CLEAR PAD FOR S-IV LOX TANKING.	
2601						
	BL1	1		ANAG	IGNITION CIRCUIT BREAKER "ON",	
	BL1	2		ANAG	IGNITION CONTACTOR ON AND SAFETY WIRE,	
	8L1	3		ANAG	VERIFY IGNITION RACK SWITCH IN SAFE POSITION.	
	8L1	4		ANAG	VERIFY AGOS ESE SECURED PER PROCEDURE 1-LLV1-713.	
245						
	PA	1		CNTS	RF SILENCE "OFF",	
	SRO	2	CNTS	DRNG	FREQUENCY CLEARANCE FOR THE FOLLOWING TM LINKS = 244.8, 251.5, AND 258.5 MC.	
	S	3	CLTC	PPSO	VERIFY PAD AREA CLEAR FUR S-IV LOX LOADING.	
	RD4	4	CLTC	RLOO	LOAD S-IV LOX PER PROCEDURE LVO-L-1004.	
					NOTE TM SYSTEMS ON AND OFF AS REQUIRED.	
215'	RD1	5	C1TC	C1VP	PRESSURIZE CONTROL SPHERES TO 3000 PSIG	
	RD4	i		PLOO	VERIFY Salv LOX LOADING COMPLETE.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

LAUNCH COUNTDOWN
DATE SEPTEMBER 5, 1964
REVISION

#### SATURN/APOLLO LAUNCH OPERATIONS

PAGE TEST NO. VEHICLE

TIME	COMM.	SEQUENCE	COMMAND STA,	RESPONSE STA.	DESCRIPTION	REMARKS
-215'						
		2		CIPC	TELEMETERS D1. D2 AND D3 "OFF".	
	RD4	3			S=IV LEAK CHECK CREW RETURN TO SERVICE STRUCTURE AND UMBILICAL TOWER AND LEAK CHECK PER PROCEDURE DAC-P-7007,	
1951						
		1		CATC	VERIFY S=IV LEAK CHECKS COMPLETE PER PROCEDURE DAC=P=7007.	
		2		SASE	CLOSE S-IV AFT INTERSTAGE.	
1851						
		1		CATC	VERIFY AFT INTERSTAGE SECURE FOR FLIGHT PER PROCEDURE DAC-P-7007.	
	YW4	2	CLTC	SSSC SMTE	OPEN SERVICE STRUCTURE SILO B=2 AND RETRACT PLATFORM #1.	
180'						
		1		CUNC	COOLING TO PREFLIGHT.	
	YW1	2	CLTC	RS4C	BRING ST-124 TO OPERATING CONDITIONS PER PROCEDURE LVO-G-7041.	
	YW1	3	CLTC	RATC	BRING AUTO THEODOLITE UP TO OPERATING CONDITIONS PER PROCEDURE LVO-G-7044.	
170'						
	YW1	1	CLTC	SSSC	MOVE SERVICE STRUCTURE TO LAUNCH LOCATION PER PROCEDURE LSO-F-1001.	
1651						
	PA	1		CNTS	CLEAR BLOCKHOUSE AND PARKING LOT OF NON-ASSIGNED PERSONNEL TO THE ROAD BLOCKS.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

# SATURN/APOLLO

PAGE

199 TEST NO. 7 = LSVI = 300 VEHICLE SA = 7 SA-7

TIME	COMM.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
150'						
		1		CPSO	SET BLAST DANGER AREA ROAD BLOCKS. AREA TO BE CLEAR BY T-115.	
	SRO	2	CNTS	ORNG	DO NOT INTERROGATE C-BAND BEACON WITH 1,16 RADAR UNTIL REQUESTED.	
1451						
	S	1	CLTC	RWCP	VERIFY WATER PRESSURE TO 175 PSIG.	
		2			RF AND TM CHECKS.	
	S	2-1	CLTC	врор	ODOP GROUND TRANSMITTER NO. 2 "ON",	
	S	5-5	CLTC	C1CM	TV TO FILAMENT.	
	S	2-3	CLTC	CUMP	ALTIMETER "ON".	
	S	2-4	CLTC	CUMP	ODOP "ON",	
	S	2-5	CLTC	CUMP	MINITRACK "ON".	
	S	2-6	CLTC	CUMP	AZUSA "ON",	
	S	2-7	CLTC	CUMP	C-BAND BEACON (RADAR) "ON",	
	S	2-8	CLTC	CUMP	MISTRAM "ON",	
	S	2-9	CLTC	Cimp	TELEMETERS F1, F2, F3, P2, S1 AND S2 "ON",	
	S	2-10	CLTC	CIMP	VERIFY TAPE RECORDER READY INDICATION,	
	S	2-11	CLTC	CUMP	TELEMETERS F5. F6. S3. AND P1 "ON".	
	S	2-12	CLTC	CUMP	VERIFY TAPE RECORDER READY INDICATION.	
	S	2-13	CLTC	C1MP CUMP	AUXILIARY EQUIPMENTS "ON".	
	S	2-14	CLTC	C1MP CUMP	SWEEP FREQUENCY CALIBRATOR "ON",	
	S	2-15	CLTC	CIPC	TELEMETERS D1, D2 AND D3 "ON",	
	S	2-16	CLTC	C1MP CUMP	TELEMETER CALIBRATIONS TO "PREFLIGHT".	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

LAUNCH COUNTDOWN DATE SEPTEMBER 5, 1964 REVISION

#### SATURN/APOLLO LAUNCH OPERATIONS

VEHICLE

PAGE 200 TEST NO. 7=LSVI=300 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
145						
	S	2~17	CLTC	CIPC	CALIBRATION SWITCH TO MANUAL (VERIFY VCO CALIBRATION LIGHT OFF).	
	S	2~18	CLTC	C1MP CUMP	PREFLIGHT CALIBRATIONS TO "50%".	
	SRO	2-19	CNTS	ORNG	INTERROGATE MISTRAM AND AZUSA AND REPORT READOUT.	
140						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	CiCM	TV TO B+.	
	S	1-2	CLTC	B1TM B4TM OTMD	TELEMETER RECORDINGS "ON".	
	S	1-3	CLTC	C1MP CUMP	TELEMETER CALIBRATIONS TO "AC COMMAND".	
	SRO	1-4	CNTS	ORNG	INTERROGATE IU C-BAND BEACON WITH 1,16 RADAR AND REPORT READOUT.	
139						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS	
138† 45**						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1MP CUMP	TELEMETER CALIBRATIONS TO "PREFLIGHT",	
	S	1-2	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-138 <sup>1</sup> 45 <sup>11</sup>						
	S	1-3	CLTC	C1MP CUMP	PREFLIGHT CALIBRATION TO "0%".	
-137° 30"						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1MP CUMP	PREFLIGHT CALIBRATIONS TO "100%",	
136 <sup>†</sup> 30 "						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1MP CUMP	RECORDER TRANSFERS "ON".	
	S	1=2	CLTC	B1TM OTMD	8 KC OSCILLATORS "ON",	
	S	1-3	CLTC	C1MP CUMP	TELEMETER CALIBRATIONS COMMANDS.	
	S	1~4	CLTC	C1MP CUMP	PREFLIGHT CALIBRATIONS TO 0%, 25%, 50%, 75%, AND 100% (IN 2 SECOND INCREMENTS).	
	S	1=5	CLTC	CIPC	STEP THROUGH MANUAL CALIBRATION IN 2 SECOND INCREMENTS.	
	S	1-6	CLTC	C1MP CUMP	TELEMETER CALIBRATION TO "INFLIGHT",	
	S	1-7	CLTC	RBHM	S=I AND IU HI CAL, COMMAND FOR 15 SEC.	
	S	1-8	CLTC	RBHM	S-I AND IU LOW CAL COMMAND FOR 15 SEC.	
	S	1-9	CLTC	RBHM	S-I AND IU RUN COMMAND.	
	S	1-10	CLTC	CIPC	CALIBRATION TO AUTOMATIC.	
	S	1-11	CLTC	CIPC	GIVE CAL START.	
	S	1-12	CLTC	Control of the State of the Sta	TELEMETER CALIBRATION COMMANDS "ON" FOR 5 SECONDS.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

# SATURN/APOLLO

PAGE 202
TEST NO. 7-LSVI=300
VEHICLE SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
136' 30"						
	S	1-13	CLTC	CIPC	BRIDGE CALIBRATION HIGH FOR 10 SECONDS.	
	S	1-14	CLTC	CIPC	BRIDGE CALIBRATION LOW FOR 10 SECONDS.	
		1-15	CLTC	CUMP	P1 CALIBRATION COMMAND "ON" AND "OFF".	
	S	1-16	CLTC	B1TM, OTMD	8 KC OSCILLATOR OFF.	
135'						
	RD4	1	CITC	PPSU	VERIFY PAD AREA CLEAR TO BEGIN S-I LOX PRECOOL:	
	RD4	2	CITC		BEGIN S-I FINAL LOX TANKING PER PRO- CEDURE LVO-L-1004 (AFTER RECORDER TRANSFER OFF).	
134' 30"						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1MP CUMP	RECORDER TRANSFERS "OFF".	
	S	1-2	CLTC	CIPC	GIVE CAL START.	
	S	1=3	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMAND "ON" FOR 5 SECONDS.	
	S	1-4	CLTC	CUMP	P1 CALIBRATION COMMAND "ON" AND "OFF",	
	S	1-5	CLTC	C1MP	TELEMETER CALIBRATION TO "PREFLIGHT".	
134'						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	CUMP	ODOP "OFF",	
	S	1-2	CLTC	BDOP	ODOP GROUND TRANSMITTERS #2 "OFF",	
	S	1-3	CLTC	C1MP	TELEMETERS F1, F2, F3, P2, S1 AND S2 "OFF",	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

# SATURN/APOLLO

PAGE

203 TEST NO. 7-LSVI-300 VEHICLE SA-7 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA,	RESPONSE STA.	DESCRIPTION	REMARKS
134						
	S	1-4	CLTC	CUMP	TELEMETERS F5, S3 AND MINITRACK "OFF",	
	S	1-5	CLTC	CIMP	TELEMETER AUXILIARY EQUIPMENT "OFF",	
	S	1-6	CLTC	CIPC	TELEMETERS D1, D2, AND D3 "OFF".	
	S	1-7	CLTC	В4ТМ	S-IV TELEMETER RECORDING "OFF",	
130 1						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	C1CM	TV "OFF".	
	S	1-2	CLTC	CUMP	C-BAND BEACON (RADAR) "OFF" WHEN READOUT COMPLETE.	
	S	1-3	CLTC	CUMP	AZUSA "OFF" WHEN READOUT IS COMPLETE.	
	s	1-4	CLTC	CUMP	MISTRAM "OFF" WHEN READOUT IS COMPLETE.	
1201						
		1			RF AND TM CHECKS.	
	S	1-1	CLTC	CUMP	ALTIMETER "OFF".	
	S	1-2	CLTC	CUMP	TELEMETERS F6 AND P1 "OFF",	
	S	1-3	CLTC	CUMP	TELEMETER AUXILIARY EQUIPMENT "OFF",	
	S	1-4	CLTC	CUMP	TELEMETER CALIBRATION TO "PREFLIGHT",	
	S	1-5	CLTC	OTMD B1TM	TELEMETER RECORDINGS "OFF".	
	YW1	2		RATC	ST-124 TO FLIGHT AZIMUTH PER PROCEDURE LVO-G-7042.	
	S	3	CLTC	RVFC	CONTROL COMPUTER "ON" (VERIFY "CAT ZERO"),	
	S	4	CLTC	RVFC	CONTROL ACCELEROMETERS "ON",	
	S	5	CLTC	RVFC	IU RATE GYROS "ON".	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-120'						
	s	6	CLTC	RVFC	S=I RATE GYROS "ON",	
	S	7	CLTC	RVFC	Q-BALL SYSTEM "ON",	
	S	8	CLTC	RVFC	G-BALL HEATERS "ON".	
	S	9	CLTC	RVFC	ACCELERATION SWITCH "ON".	
115'						
	S	1	CLTC	RGCC	INITIATE APS 103, FLIGHT COMPUTER SYSTEM FINAL LAUNCH PREPARATION.	
1000						
	S	1		RECS	SWITCH ECS MODE FROM AIR TO GN2 PER PROCEDURE LV0-E-7018.	
	YW4	2			VERIFY SERVICE STRUCTURE SECURED AT LAUNCH LOCATION, PERSONNEL CLEAR AREA TO ROADBLOCK OR BLOCKHOUSE AS ASSIGNED.	
901						
		1		CPSU	VERIFY COMPLEX CLEAR OF ALL PERSONNEL AND SECURE BLOCKHOUSE DOOR.	
	S	2	CLTC	RVFC	CONTROL VOLTAGE "ON",	
85 !						
		1		C1FP	DELUGE PURGE TO AUTO.	
	S	2	CLTC	CPSP	GENERATOR COMMIT ON ALL GENERATORS.	
	S	3	CLTC	CPSU	RELEASE DESTRUCT ENABLE KEY TO CLVN.	
	S	4	CLTC	RBHM	SANBORN RECORDER TO 5 MM/SEC.	
	S	5	CLTC	C4DP	VERIFY S&A SAFE.	
	S	6	CLTC	CLVN	DESTRUCT ENABLE ON.	
	S	7	CLTC	C40P	PANEL SELECT TO AUXILIARY RECORDER	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

7=LSVI=300 SA=7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-851						
	s	8	CLTC	R4VC	VERIFY VISICORDER ON,	
	S	9	CLTC	C1DP C4DP	S=I AND S=IV EBW°S #1 AND #2 ON.	
	s	10	CLTC	C1DP C4DP	VERIFY S-I AND S-IV EBW #1 AND #2 VOLTAGES OK.	
	S	11	CLTC	C1DP C4DP	S=I AND S=IV EBW'S #1 AND #2 "OFF",	
	S	12	CLTC	RBHM	SANBORN RECORDERS TO 2 MM/SEC.	
	S	13	CLTC	CLVN	DESTRUCT ENABLE OFF AND DELIVER DESTRUCT ENABLE KEY TO CPSO.	
		14		CPSO	VERIFY BLAST DANGER AREA SECURE FOR LH2 TANKING.	
-801						
	RD4	1		RLOO	COMPLETE S-I LOX LOADING TO 95%,	
	RD4	2		RLOO	BEGIN S-I AND S-IV LOX REPLENISH.	
	RD4	3	RLHC		VERIFY VEHICLE HAS BEEN PURGED WITH GN2 FOR AT LEAST 15 MINUTES PRIOR TO LH2 LOADING.	
	RD4	4			BEGIN S-IV LH2 LOADING PER PROCEDURE LVO-L-1033 (NOTE: TM SYSTEMS "ON" AND "OFF" AS REQUIRED);	
		5		CPSO	SET LAUNCH DANGER AREA ROADBLOCKS (CAPE ROAD TO REMAIN OPEN UNTIL T-50).	
-72 i						
	sc	1		CSTC	SPACECRAFT EXTERNAL POWER ON.	
-701						
	RD1	1			PERFORM S-I LOX BUBBLING CHECK (10 MINUTES AFTER S-I LOX LOADING COMPLETE)	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
701						
	RD1	2	C1TC	C1VP	PRESSURIZE HELIUM SPHERE TO 3000 PSIG.	
	RD1	3	CITC	C1VP	PRESSURIZE FUEL SPHERES AND CAMERA PURGE SPHERES TO 3000 PSIG.	
	RD1	4	CITC	C1VP	PRESSURIZE CAMERA EJECT SPHERE TO 3000 PSIG.	
	RD1	5	CLTC	CUNC	OPERATE GAS BEARING SWITCH "ON",	
621						
	sc	1	CSTC	CICM	SPACECRAFT TELEMETER A (LINK 1) "ON",	
	sc	2	CSTC	CICM	SPACECRAFT TELEMETER B (LINK 6) "ON".	
	SC	3	CSTC	CICM	SPACECRAFT TELEMETER C (LINK 14) "ON".	
601						
	SRO	1	CNTS	ORNG	VERIFY PATRICK 0-18 RADAR AWAY FROM PAD	
	SC	2	CSTC	CICM	SPACECRAFT RADAR BEACON #1 "ON".	
	SC	3	CSTC	CICM	SPACECRAFT RADAR BEACON #2 "ON".	
	SRO	4	CNTS	ORNG	ONE (1) MINUTE AFTER NOTIFICATION OF BEACON ON, BEGIN INTERROGATION OF BOTH BEACONS AND PROCEED WITH READOUT, REPORT READOUT WHEN COMPLETE.	
551						
	SRO	1	CNTS	ORNG	BEGIN READOUT OF SPACECRAFT TELEMETERS.	
501						
		1			START INCREASING COLD HELIUM SPHERES PRESSURE TO 3000 PSIG WHEN LH2 LEVEL REACHES 70%.	
	S	5	CLTC	C1NP CUNP	VERIFY FLIGHT SIMULATION SWITCHES "OFF"	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-40 *						
	SRO	1	CNTS	ORNG	REPORT STATUS OF SPACECRAFT RADAR BEACONS (GO/NO GO).	
-351						
	RD4	1		RLHC	COMPLETE LH2 LOADING.	
		2		CIPC	TELEMETERS D1, D2 AND D3 OFF,	
	SIV	3		C4TC	PERFORM S=IV TERMINAL COUNTDOWN PER PROCEDURE DAC-N-1004.	
	RD4	4		The state of the s	PERFORM S-IV LOX BUBBLING CHECK PER PROCEDURE DAC-P-7010 (AFTER LH2 LOADING IS COMPLETE).	
-321						
	SRO	1	CNTS	ORNG	REPORT STATUS OF SPACECRAFT TELEMETERS (GO/NO GO).	
-301						
	S	1	CLTC		VERIFY FLIGHT COMPUTER CHECK-OUT COMPLETE AND READY FOR S-IV AND S-I STEERING.	
	s	2	CLTC	C1FP RBHM	PERFORM GG LOX INJECTOR PURGE TEST.	
	S	3	CLTC	CUNC	FILL INFLIGHT COOLER SYSTEM.	
	S	4	CLTC	CUMP	ALTIMETER "ON".	
	S	5	CLTC	CPSO	VERIFY HOLD FIRE CHECKS COMPLETE.	
	S	6	CLTC	CISP	GODDARD SYSTEMS STATUS CHECK.	
	GY1	7			MAKE COMMUNICATION CHECK WITH HANGAR D TM STATION IN GRAY 1 CIRCUIT.	
-251						
	S	1	CLTC	C1CM	TELEVISION TO "FILAMENT".	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
251						
	S	2	CLTC	CUMP	AZUSA "ON",	
	S	3	CLTC	C1FP	G.G. LOX INJECTOR PURGE TO "AUTOMATIC".	
	S	4	CLTC	C1FP	LOX DOME PURGE TO "AUTOMATIC".	
	S	5		C1FR	NOTE PURGES ARMED INDICATION.	
	S	6,	CLTC	CIPC	TELEMETERS D1, D2 AND D3 "ON".	
	S	7	CLTC	CIPC	VERIFY VCO CALIBRATION OFF AND AUTO.	
	S	8	CLTC	B4TM	S-IV TELEMETER RECORDER "ON".	
	S	9	CLTC	R4VC	VISICORDER "ON".	
	S	10	CLTC	B4TM	PERFORM TM CHANNEL READOUTS (NOTE: VERIFY CHANNELS 21-24 OF TELEMETER D2 HIGH LEVEL PDM READING 100%).	
	S	11		CLTC	PERSONNEL CONNECTED WITH S-IV STEERING COMMANDS SWITCH TO YW2 CIRCUIT.	
		12			S-IV STEERING COMMANDS.	
	YW2	12-1		RVFC	FLIGHT CONTROL SWITCH TO "S-IV".	
	YW2	12-2		RVFC	HYDRAULIC PUMP ENABLE "ON",	
	Yw2	12-3	RVFC	R4FC	S-IV HYDRAULIC PUMPS "ON" .	
	Y W 2	12-4	RVFC	RGCC	INITIATE S-IV STEERING COMMANDS (APS 105),	
	YW2	12-5		RVFC	IU RATE GYRO TORQUER COMMANDS.	
	YW2	12-6	RVFC	R4FC	CLOSE S-IV ACCUMULATOR VALVES.	
	Y W 2	12-7	RVFC	R4FC	VERIFY S-IV HYDRAULIC PUMPS "OFF"	
	YW2	12-8		RVFC	HYDRAULIC PUMPS ENABLE "OFF".	
	YW2	12-9	RVFC	R4FC	CHECK ACTUATOR ACCUMULATOR PRESSURES AT 2950 PSIA PLUS OR MINUS 150.	
	YW2	12-10		RVFC	STAGE SELECTOR SWITCH TO "S-I",	
	S	13	CLTC	RRPO	MAKE RP-1 LEVEL ADJUSTMENT PER PROCEDURE LVO-L-1030.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
251						
	S	14	CLTC	C4TC	REPORT COMPLETION OF S-IV LOX BUBBLING CHECK	
	S	15	CLTC	C4TC	REPORT COMPLETION OF S-IV TERMINAL COUNTDOWN PREPARATIONS.	
	SRO	16	CNTS	ORNG	VERIFY 0.18 AND 1.16 RADARS AWAY FROM PAD.	
	SRO	17	CNTS	ORNG	DO NOT INTERROGATE C-BAND BEACON WITH 1,16 RADAR UNTIL REQUESTED.	
24 7						
	s	1	CLTC	C1MP	TELEMETERS F1, F2, F3, P2, S1 AND S2 "ON".	
	S	2	CLTC	CUMP	C-BAND BEACON (RADAR) "ON".	
	S	3	CLTC	C1MP	TELEMETER AUXILIARY EQUIPMENT "ON",	
	S	4	CLTC	CUMP	TELEMETERS F5, F6, S3, P1 AND MINITRACK "ON".	
	S	5	CLTC	CUMP	TELEMETER AUXILIARY EQUIPMENT "ON",	
	S	6	CLTC	C1CM	CYCLE ALL S-I CAMERAS AS REQUIRED.	
	S	7	CLTC	C1MP CUMP	TELEMETER CALIBRATIONS TO "PREFLIGHT".	
	S	8	CLTC	C1MP CUMP	PREFLIGHT CALIBRATIONS TO 50%.	
	S	9	CLTC	OTMD	SPECIAL RECORDER "ON",	
	S	10	CLTC	C1MP CIPC CUMP	VERIFY TAPE RECORDER READY INDICATIONS.	
21 0						
	S	1	CLTC	BSOC	VERIFY SCAN RATE OF 10 SEC INSERTED INTO PROPELLANT SCAN PROGRAM.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE VEHICLE

210 TEST NO. 7-LSVI-300 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
201						
	s	1	CLTC	BDOP	ODOP GROUND TRANSMITTER "ON".	
	S	2	CLTC	C1MP CUMP	PREFLIGHT CALIBRATIONS TO 0%.	
	S	3	CLTC	CUMP	MISTRAM "ON",	
	S	4	CLTC	CUMP	ODOP "ON".	
	SRO	5	CNTS	ORNG	INTERROGATE AZUSA AND MISTRAM.	
		6	CNTS	CPSO	VERIFY LES ARMING KEY HAS BEEN RELEASED TO CSTC.	
	S	7	CLTC	BSOC	INITIATE PROPELLANT LOADING SCAN RATE OF 10 SEC.	
191						
	s	1	CLTC	RSEQ	SEQUENCE AND E&I RECORDERS TO "MINUTE SPEED" AND TIME PULSE "ON".	
	S	2	CLTC	RSEQ	VERIFY VOLTAGE RECORDERS TO "FINE RANGE",	
	SRO	3	CNTS	ORNG	INTERROGATE IU C-BAND BEACON WITH 1.16 RADAR.	
	SRO	4	CNTS	ORNG	INTERROGATE SC C-BAND BEACON WITH 0.18 RADAR.	
	S	5	CLTC	CPSO	RELEASE DESTRUCT ENABLE KEY TO CLVN.	
181						
	s	1	CLTC	C4DP	MODULE POWER SUPPLY ON AND CHECK FOR 5 PLUS OR MINUS 0.2 VOLTS.	
	S	2	CLTC	C4DP	TEST S-IV EBWIS FOR "READY",	
	S	3	CLTC	C4DP	VERIFY S-IV S&A "SAFE".	
	S	4	CLTC	CSRP	VERIFY S-I AND IU FLIGHT SEQUENCERS	
	S	5	CLTC	CSPC	S-IV EBW EXTERNAL "ON".	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
181						
	S	6	CLTC	C40P	VERIFY ALL READY INDICATIONS "OFF" AND NOT READY INDICATIONS "ON",	
	S	7	CLTC	CSSP	VERIFY MSFC TALKBACK ENABLE "ON".	
17 1						
	S	1	CLTC	CLVN	DESTRUCT ENABLE "ON".	
	S	2	CLTC	C4DP	VERIFY SAFETY BUSS "ON" INDICATION.	
	S	3	CLTC	C1MP CUMP	PREFLIGHT CALIBRATION TO 100%.	
	S	4	CLTC	ORNG	BRING UP WATER PUMPS 5 AND 6 TO "STANDBY",	
161						
	S	1	CLTC	CUMP	HORIZON SENSOR POWER "ON".	
	S	2	CLTC	CDCG	CHECK VEHICLE GROUND POWER GENERATOR SETTINGS FOR POWER TRANSFER.	
15 †						
	SRO	1	CNTS	ORNG	GUIDANCE COMMAND CARRIER "ON".	
	SRO	2	CNTS	ORNG	RANGE SAFETY COMMAND CARRIER "ON",	
	Y W Z	3		RVFC	S-IV STEERING COMMAND PERSONNEL SWITCH TO S-CIRCUIT.	
	S	4	CLTC	RVFC	VERIFY S-IV STEERING COMMANDS COMPLETE.	
	S	5	CLTC	C1CM	TELEVISION TO "8+".	
	S	6	CLTC	CUMP	GUIDANCE COMMAND RECEIVER "ON",	
	S	7	CLTC	RRPO	RP=1 LINE INERT PER PROCEDURE LVO-L-1030.	
	SC	8	CNTS	CSTC	PROCEED WITH PREPARATIONS FOR SPACECRAFT INTERNAL POWER.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

# SATURN/APOLLO

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-14130"						
	S	1	CLTC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS.	
	S	2	CLTC	RBHM	SANBORN RECORDER TO 5 MM/SEC.	
	S	3	CLTC	C10P	FUNCTION SELECTOR TO FIRING UNIT VOLTAGE CHECK.	
	S	4	CLTC	C1DP	S-I CDR #1 AND #2 TO "INTERNAL".	
	S	5	CLTC	C1MP CUMP	TELEMETER CALIBRATIONS TO "INFLIGHT".	
	S	6	CLTC	C4DP	S-IV CDR #1 AND #2 "ON".	
	S	7	CLTC	C4DP	VERIFY S-IV EBW #1 AND #2 "OFF" AND "NOT READY".	
	S	8	CLTC	C4DP	S-IV CDR #1 AND #2 TO "INTERNAL".	
141						
	S	1	CLTC	CIPC	GIVE CAL START.	
	S	2	CLTC	CUNP	VERIFY IU READY FOR POWER TRANSFER "ON"	
	SRO	3	CNTS	ORNG	CUTOFF COMMAND.	
	S	4	CLTC	C1DP C4DP	VERIFY EBW #1 AND #2 "INTERNAL" AND "CHARGED",	
	S	5		B4TM RFSR	NOTE ALL ENGINE CUTOFF NOT RECEIVED.	
		6		RFSR	NOTE HELIUM HEATER CUTOFF NOT RECEIVED.	
	SRO	7	CNTS	ORNG	RELEASE CUTOFF COMMAND,	
	S	8	CLTC	C1DP C4DP	S-I AND S-IV EBW #1 AND #2 TO "EXTERNAL".	
	S	9	CLTC	C1DP C4DP	VERIFY S-I AND S-IV EBW'S #1 AND #2	
	S	10	CLTC	CUNP	HORIZON SENSOR "ON".	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
141						
	s	11	CLTC	1,500,000,000	TELEMETER CALIBRATION COMMANDS "ON" FOR 5 SECONDS.	
	S	12	CLTC	CUMP	P1 CALIBRATE COMMAND "ON" AND "OFF".	
131						
	S	1	CLTC	ORNG	MONITOR ALL RF SYSTEMS AND REPORT ANY CHANGE ON INTERNAL POWER.	
	S	2	CLTC	RVFC	HYDRAULIC PUMP ENABLE "ON".	
	S	3	CLTC	RVFC	S-I HYDRAULIC PUMPS "ON".	
	S	4		CLTC	PERSONNEL CONNECTED WITH S-IV POWER TRANSFER SWITCH TO S-IV CIRCUIT.	
	SIV	5	C4TC	CIPC	INSTRUMENTATION SYSTEMS TO "INTERNAL",	
	SIV	5-1	C4TC	CSPC	STAGE POWER SYSTEM TO "INTERNAL",	
	SIV	5=2	C4TC	CSPC	READOUT BATTERIES.	
	SIV	5-3	C4TC	CIPC	PERFORM AUTOMATIC VCO CALIBRATION.	
	SIV	5-4	C4TC	CSPC CIPC B4TM	VERIFY SATISFACTORY ON INTERNAL POWER.	
	SIV	5=5	C4TC	CSPC	STAGE POWER SYSTEM TO "EXTERNAL".	
	SIV	5-6	C4TC	CIPC	INSTRUMENTATION SYSTEMS TO "EXTERNAL",	
	SIV	5-7	C4TC	C4DP	CDR'S #1 AND #2 TO "EXTERNAL".	
	S	6	CLTC	RGCC	INITIATE S-I STEERING COMMANDS (APS-104).	
	S	7	CLTC	C1CM	CAMERA LIGHTS "ON" (HOLD ON).	
	S	8	CLTC	CINP	ON MARK S-I STAGE POWER TEST #2 "ON".	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
131						
	S	9		CSRP	NOTE S-I POWER TRANSFER OK.	
	S	10	CLTC	C1CM	CAMERA LIGHTS "OFF".	
	S	11	CLTC	CUNP	ON MARK IU STAGE POWER TEST #2 "ON",	
	S	12		CSRP	NOTE IU POWER TRANSFER OK.	
	S	13	CLTC	CSRP	VERIFY SATISFACTORY S-I AND IU POWER TRANSFER.	
	S	14		C1NP	S-I STAGE POWER TEST #2 "OFF",	
	S	15		CUNP	IU STAGE POWER TEST #2 "OFF",	
	SIV	16		CATC	S-IV POWER TRANSFER PERSONNEL SWITCH TO S-CIRCUIT,	
	S	17	CLTC	C4TC	VERIFY SATISFACTORY S-IV POWER TRANSFER	
	S	18	CLTC	CUNP	HORIZON SENSOR "OFF".	
	S	19	CLTC	CPSC	VERIFY TEMPERATURES OK FOR LOX RUBBLING.	
12130"						
	SC	1	CNTS	CSTC	BEGIN TRANSFER OF SPACECRAFT TO "INTERNAL POWER",	
111						
	S	1	CLTC	RFDM	ARM FIRE DETECTION AND RECORDERS TO SLOW SPEED,	
	S	2	CLTC	RCSM	CSM POWER "ON".	
	S	3	CLTC	RÇSM	CSM ACTIVATE "ON",	
	S	4		C1FR	NOTE CSM READY "ON".	
	S	5		CLTC	PERSONNEL CONNECTED WITH TELEMETER PREFLIGHT CALIBRATION SWITCH TO GRAY 1 CIRCUIT.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

# SATURN/APOLLO

PAGE

215 TEST NO. 7-LSVI-300 VEHICLE SA-7 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
10:30"						
		1			TELEMETER PREFLIGHT CALIBRATION.	
	GY1	1-1	C1TC	B1TM OTMD	TELEMETER RECORDINGS "ON",	
	SRO	1-2	CNTS	ORNG	TELEMETER RECORDINGS "ON".	
	GY1	1-3	C1TC	C1MP CUMP	TELEMETER CALIBRATIONS TO AC COMMAND.	
	GY1	1-4	C1TC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS "ON" AND 10 SECONDS BEFORE NEXT STEP).	
	GY1	1-5	CITC	CIPC	VCO CALIBRATION TO "MANUAL".	
	GY1	1-6	C1TC	C1MP CUMP	TELEMETER CALIBRATIONS TO "PREFLIGHT".	
	GY1	1-7	C1TC	C1MP CUMP	TELEMETER CALIBRATION COMMANDS.	
	GY1	1-8	C1TC		PREFLIGHT CALIBRATIONS TO 0%, 25%, 50%, 75%, AND 100% IN 2 SECOND INCREMENTS,	
	GY1	1-9	C1TC	Section of the sectio	STEP THROUGH MANUAL CALIBRATION IN 2 SECOND INCREMENTS.	
	GY1	1-10	Citc	C1MP CUMP	TELEMETER CALIBRATIONS TO "INFLIGHT".	
	GY1	1-11	C1TC	C1MP CUMP	RECORDER TRANSFERS "ON".	
	GY1	1-12	CITC		TELEMETER CALIBRATION COMMANDS "ON" FOR 5 SECONDS,	
	GY1	1-13	C1TC		P-1 TELEMETER CALIBRATION COMMAND "ON" AND "OFF",	
	GY1	1-14	C1TC	CIPC	VCO CALIBRATION TO "AUTOMATIC".	
	GY1	1-15	CITC	CIPC	GIVE CAL START.	
	GY1	1-16	C1TC	RBHM	S-I AND IU HIGH CAL COMMAND FOR 15 SEC.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-10'30"						
	GY1	1-17	C1TC	RBHM	S=I AND IU LOW CAL COMMAND FOR 15 SEC.	
	GY1	1-18	C1TC	RBHM	S-I AND IU RUN COMMAND.	
	GY1	1-19	CITC	C1MP CUMP	RECORDER TRANSFER "OFF".	
	GY1	1-20	C1TC	B1TM OTMD B4TM	TELEMETER RECORDING "OFF" (AND RELOAD IF REQUIRED).	
	SRO	1-21	CNTS	ORNG	TELEMETER RECORDING "OFF",	
	GY1	1-22		C1TC	TELEMETER CALIBRATION PERSONNEL RETURN TO S CIRCUIT.	
	S	5		C1TC	TELEMETER PREFLIGHT CALIBRATION COMPLETE,	
	S	3	CLTC	CPPC	VERIFY ALL HELIUM SPHERE PRESSURES NORMAL AND SUFFICIENT COLD HELIUM MASS ABOARD.	
-10'10"						
	S	1	CLTC	CPSC	LOX BUBBLING TO "MANIFOLD COOLDOWN".	
-10'		8		l)		
	S	1	CLTC	RVFC	S-I HYDRAULIC PUMPS "OFF".	
	S	2	CLTC	RVFC	HYDRAULIC PUMPS ENABLE "OFF".	
	S	3	CLTC		VERIFY STEERING COMPLETE AND READY TO LAUNCH.	
8'10"						
	S	1	CLTC	CPSC	LOX BUBBLING TO "BUBBLE SUPPLY".	
8 *						
	S	1	CLTC	CISP	GODDARD SYSTEMS STATUS CHECK.	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-8 *						
	S	2	CLTC	RUAC	CONDUCT SWING ARM HYDRAULIC TEST PER LV0-E-7016.	7
	S	3	CLTC	RVFC	REMOVE Q-BALL COVER.	
-7130"						
	SRO	1	CNTS	ORNG	TELEMETER RECORDING "ON".	
	sc	2		CSTC	PROCEDE WITH SPACECRAFT R AND Z CALIBRATIONS.	
71						
	S	1	CLTC	CPSO	PERMISSION TO ARM IGNITION "ON".	
		2	CLTC	C1LS	LAUNCH SEQUENCER ARM "ON".	
		3	CLTC	C1LS	TIME PULSE "ON".	
		4		C1FR	NOTE LAUNCH SEQUENCER READY.	
		5	CLTC	C4DP	VERIFY S-IV EBW'S #1 AND #2 "OFF" AND "NOT READY".	
		6	CLTC	RRPO	VERIFY RP=1 LEVEL ADJUST & LINE INERT COMPLETE.	
	SC	7		CSTC	VERIFY SPACECRAFT SATISFACTORY POWER TRANSFER.	
	S	8	CNTS	CSTC	PROCEDE WITH LES ARM SEQUENCE.	
6 1 3 0 "						
	SRO	1	CNTS	ORNG	TELEMETER RECORDING "OFF".	
61						
	S	1	CLTC	CUNP	ARM IU SAFETY SWITCHES,	
	S	2	CLTC	CINP	ARM S-I SAFETY SWITCHES.	
	S	3	CLTC	CGCM	CAMERA START TO "ARM".	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-6 †						
	s	4	CLTC	C4DP	S-IV CDR #1 AND #2 TO "INTERNAL".	
	S	5	CLTC	C4DP	S-IV CDR #1 AND #2 SWITCHES "OFF" (VERIFY CDR'S REMAIN "ON"),	
-51						
	S	1	CLTC	CPER	VERIFY Q-BALL COVER REMOVED.	
	S	2	CLTC	C1FR	VERIFY RANGE SAFE,	
	S	3	CLTC	RGCC	VERIFY FLIGHT COMPUTER SYSTEM IN READY TO LAUNCH MODE.	
	S	4	CLTC	CINP	FUNCTION SELECTOR TO "LAUNCH".	
	S	5	CLTC	CINP	IGNITION ARMING "ON".	
	S	6		C1FR	NOTE IGNITION SEQUENCER READY.	
	s	7	CLTC	C4TC	VERIFY ALL STATUS SELECTOR SWITCHES TO "READY TO FIRE".	
	S	8	CLTC	RFSR	SEQUENCE RECORDERS TO 10 MM/SEC.	
-41301						
	sc	1		CSTC	SPACECRAFT TO "READY",	
	S	2	CLTC	RVFC	HYDRAULIC PUMPS ENABLE "ON".	
	S	3	CLTC	RVFC	S-I HYDRAULIC PUMPS "ON".	
- 4 1						
	s	1	CLTC	CUNC	SWITCH TO "INFLIGHT COOLING".	
	SRO	2	CNTS	ORNG	CLEARANCE FOR LAUNCH.	
	S	3	CLTC	C1CM	ZERO THE CAMERA TIMER.	
	S	4	CLTC	RVFC	RATE GYRO SIMULATED COMMANDS,	
	S	5	CLTC	RPSQ	SEQUENCE RECORDERS TO "MINUTE SPEED".	

- 1. C1NP: GIVE CUTOFF (CAUTION: HOLD ON)
- 2. C1FP: PREVALVES 1-8 OFF
- 3. C1NP: RELEASE CUTOFF
- 4. C1FP: FUEL BUBBLING SWITCH OFF
- 5. Note: The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks should then be vented and bubbling discontinued when the pump inlet temperatures reach -275° F.
- 6. Note: Upon exceeding 10 minutes the prevalves must remain closed, Lox drained and standby observed until all suction line frost is melted. The pump inlet pressures must be monitored during this period and if the pressure exceeds 200 PSIG (full chart scale) the prevalves must be opened.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-3 1 40 11						
	s	1	CLTC	в4ТМ	S-IV TELEMETER RECORDING "ON".	
	S S	2	CLTC	C1DP C4DP	ARM DESTRUCT S&A.	
	S	3		C1FR	NOTE S-IV SYSTEMS READY INDICATION "ON"	
	S	4		C1FR	NOTE SC SYSTEMS READY INDICATION "ON",	
		5	CLTC	BSOC	CDC INHIBIT "OFF".	
		6	CLTC	BSOC	DISCRETE INHIBIT "OFF",	
-3'10"						
	S	1	CLTC	CPSC	LOX BUBBLING "OFF".	
-31						
	S	1	CLTC (	RSDC	ENABLE GUIDANCE RELEASE.	
-2153"						
	S	1	CLTC	RGCC	VERIFY GUIDANCE RELEASE.	
	S	2		C1FR	NOTE "IU READY".	
	S	3	CLTC	C1FR	VERIFY "PREP COMPLETE".	
2145"						
	SRO	1	CNTS	ORNG	TELEMETER RECORDING "ON".	
	S	2	CLTC	OTMD B1TM	TELEMETER RECORDINGS "ON".	
-2138#						
	S	1	CLTC	RSEQ	SEQUENCE RECORDERS TO "FAST SPEED".	

Note: Spacecraft has indefinite recycle capability until T-18", Limited Thereafter

#### To INTERRUPT AUTOMATIC LAUNCH SEQUENCE

- 1. C1FR: GIVE CUTOFF
- 2. C1NP: IGNITION ARMING OFF
- 3. CPSO: IGNITION PERMISSION OFF
- 4. CPSO: HOLD FIRE ON
- 5. RSEQ: RECORDERS TO SLOW SPEED
- 6. RWCP: OPERATE WATER CONTROL SYSTEM AS REQ! D
- 7. RVFC HYDRAULIC PUMPS 1, 2, 3, AND 4 OFF
- 8. RBHM: VERIFY FUEL AND LOX TANKS VENTED
- 9. RPLC: VERIFY LH2 AND LOX TANKS VENTED
- 10. C1NP: FUNCTION SELECTOR TO PRELAUNCH
- 11. C1LS: LAUNCH SEQUENCER ARM OFF
- 12. C1DP. CLDP: SAFE DESTRUCT
- 13. C1NP, CUNP: SAFETY SWITCH ARMING OFF
- 1)L. C1NP: RESET CUTOFF
- 15. C1NP: RESET IGNITION SEQUENCER
- 16. CNTS: Secure ALL OTHER EQUIP. AS REQ'D AND MAKE ANNOUNCEMENT FOR ALL PERSONNEL TO STAND BY UNTIL DECISION IS MADE TO RECYCLE OR SCRUB

#### PREMATURE CLOSURE OF PREVALVES

- 1. NOTE: THE PREVALVES MAY REMAIN CLOSED FOR 10 MINUTES DURING WHICH THE LOX TANKS MUST BE PRESSURIZED. PREVALVES OPENED AND LOX BUBBLING INITIATED. THE LOX TANKS SHOULD THEN BE VENTED AND BUBBLING DIS-CONTINUED WHEN THE PUMP INLET TEMPERATURES REACH -275°F.
- 2. Note: Upon exceeding 10 minutes the pre-VALVES MUST REMAIN CLOSED, LOX DRAINED AND STANDBY OBSERVED UNTIL ALL SUCTION LINE FROST IS MELTED. THE PUMP INLET PRESSURES MUST BE MONITORED DURING THIS PERIOD AND IF THE PRESSURE EXCEEDS 200 PSIG (FULL CHART SCALE) THE PREVALVES MUST BE OPENED.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-2†33#						
	S	1	CLTC	C1FR	LAUNCH SEQUENCE START,	
	S	2		C1FR	MONITOR FUEL PRESSURIZING.	
	S	3		C1FR	MONITOR LOX BUBBLING.	
	S	4		CSSP	MONITOR S-IV STAGE POWER AND INSTRUMENTATION POWER SYSTEMS INTERNAL.	
2128"						
	S	1		C1FR	MONITOR S-I FUEL TANKS PRESSURIZED.	
	S	2		CSSP	MONITOR LO2 & LH2 SYSTEMS READY INDICATION.	
2'23"						
	S	1		RSEQ	SEQUENCE RECORDERS TO SLOW SPEED.	
2 † 3 **						
	S	1		CSSP	REPORT S-IV READY FOR LAUNCH WHEN INDICATION APPEARS AND REMAINS FOR AT LEAST 5 SECONDS.	
	S	2			S-IV UMBILICAL LINE VENT AND PURGE SWITCHES TO OPEN (AFTER S-IV READY FOR LAUNCH),	
1'53"						
	S	1		RPLC	MONITOR S=IV LOX SYSTEM CONTROL TO LOCAL (AFTER LOX SYSTEM READY).	
1 9 48 11						
CONSIDERATION.	S	1		RSEQ	SEQUENCE RECORDERS TO FAST SPEED.	
1 1 4 3 "						
	S	1		C1FR	MONITOR S-I LOX BUBBLING "OFF".	

Note: Spacecraft has indefinite recycle capability until T-18", Limited Thereafter

#### To INTERRUPT AUTOMATIC LAUNCH SEQUENCE

- 1. C1FR: GIVE CUTOFF
- 2. C1NP: IGNITION ARMING OFF
- 3. CPSO: IGNITION PERMISSION OFF
- 4. CPSO: HOLD FIRE ON
- 5. RSEQ: RECORDERS TO SLOW SPEED
- 6. RWCP: OPERATE WATER CONTROL SYSTEM AS REQ! D
- 7. RVFC HYDRAULIC PUMPS 1, 2, 3, AND 4 OFF
- 8. RBHM: VERIFY FUEL AND LOX TANKS VENTED
- 9. RPLC: VERIFY LH2 AND LOX TANKS VENTED
- 10. C1NP: FUNCTION SELECTOR TO PRELAUNCH
- 11. C1LS: LAUNCH SEQUENCER ARM OFF
- 12. C1DP. CLDP: SAFE DESTRUCT
- 13. C1NP, CUNP: SAFETY SWITCH ARMING OFF
- 1)L. C1NP: RESET CUTOFF
- 15. C1NP: RESET IGNITION SEQUENCER
- 16. CNTS: Secure ALL OTHER EQUIP. AS REQ'D AND MAKE ANNOUNCEMENT FOR ALL PERSONNEL TO STAND BY UNTIL DECISION IS MADE TO RECYCLE OR SCRUB

#### PREMATURE CLOSURE OF PREVALVES

- 1. NOTE: THE PREVALVES MAY REMAIN CLOSED FOR 10 MINUTES DURING WHICH THE LOX TANKS MUST BE PRESSURIZED. PREVALVES OPENED AND LOX BUBBLING INITIATED. THE LOX TANKS SHOULD THEN BE VENTED AND BUBBLING DIS-CONTINUED WHEN THE PUMP INLET TEMPERATURES REACH -275°F.
- 2. Note: Upon exceeding 10 minutes the pre-VALVES MUST REMAIN CLOSED, LOX DRAINED AND STANDBY OBSERVED UNTIL ALL SUCTION LINE FROST IS MELTED. THE PUMP INLET PRESSURES MUST BE MONITORED DURING THIS PERIOD AND IF THE PRESSURE EXCEEDS 200 PSIG (FULL CHART SCALE) THE PREVALVES MUST BE OPENED.

# SATURN/APOLLO

PAGE

221 TEST NO. 7-LSVI-300 VEHICLE SA-7 SA-7

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARK
1 43 "						
	s	2		C1FR	MONITOR S-I LOX VENT AND RELIEF VALVES	
	S	3		C1FR	MONITOR S-I LOX PRESSURIZING.	
	S	4		RBHM	REPORT S-I LOX TANK PRESSURES DURING LOX TANK PRESSURIZATION.	
1'0"						
	S	1		RCSM	RECORDERS "ON".	
	S	2		RFDM	RECORDERS TO "FAST SPEED",	
	S	3		RBHM	OSCILLOGRAPH "ON" :	
-38"						
	S	1		RWCP	MONITOR PAD FLUSH OPENING.	
	S	2		OTMD, B1TM	8 KC OSCILLATOR "ON".	
		3		ORNG	WATER PUMPS 5 AND 6 TO FULL SPEED.	
-28"						
	S	1		CINP	VERIFY S-I POWER TRANSFER.	
	S	2		CUNP	VERIFY IU POWER TRANSFER.	
	S	3		C1FR	MONITOR TO FUEL INJECTOR PURGE "ON".	
	S	4		C1FR	MONITOR GG LOX INJECTOR PURGE "ON",	
	S	5		C1FR	MONITOR S-I LOX DOME PURGE "ON",	
	S	6		C1FR	MONITOR S-I LOX TANKS PRESSURIZED.	
18"						
	S	1		CSRP	MONITOR READY FOR IGNITION "ON",	
	S	2		CSRP	MONITOR SC UMBILICAL ARM RELEASE.	

Note: Spacecraft has indefinite recycle capability until T-18", Limited Thereafter

#### To INTERRUPT AUTOMATIC LAUNCH SEQUENCE

- 1. C1FR: GIVE CUTOFF
- 2. C1NP: IGNITION ARMING OFF
- 3. CPSO: IGNITION PERMISSION OFF
- 4. CPSO: HOLD FIRE ON
- 5. RSEQ: RECORDERS TO SLOW SPEED
- 6. RWCP: OPERATE WATER CONTROL SYSTEM AS REQ! D
- 7. RVFC HYDRAULIC PUMPS 1, 2, 3, AND 4 OFF
- 8. RBHM: VERIFY FUEL AND LOX TANKS VENTED
- 9. RPLC: VERIFY LH2 AND LOX TANKS VENTED
- 10. C1NP: FUNCTION SELECTOR TO PRELAUNCH
- 11. C1LS: LAUNCH SEQUENCER ARM OFF
- 12. C1DP. CLDP: SAFE DESTRUCT
- 13. C1NP, CUNP: SAFETY SWITCH ARMING OFF
- 1)L. C1NP: RESET CUTOFF
- 15. C1NP: RESET IGNITION SEQUENCER
- 16. CNTS: Secure ALL OTHER EQUIP. AS REQ'D AND MAKE ANNOUNCEMENT FOR ALL PERSONNEL TO STAND BY UNTIL DECISION IS MADE TO RECYCLE OR SCRUB

#### PREMATURE CLOSURE OF PREVALVES

- 1. NOTE: THE PREVALVES MAY REMAIN CLOSED FOR 10 MINUTES DURING WHICH THE LOX TANKS MUST BE PRESSURIZED. PREVALVES OPENED AND LOX BUBBLING INITIATED. THE LOX TANKS SHOULD THEN BE VENTED AND BUBBLING DIS-CONTINUED WHEN THE PUMP INLET TEMPERATURES REACH -275°F.
- 2. Note: Upon exceeding 10 minutes the pre-VALVES MUST REMAIN CLOSED, LOX DRAINED AND STANDBY OBSERVED UNTIL ALL SUCTION LINE FROST IS MELTED. THE PUMP INLET PRESSURES MUST BE MONITORED DURING THIS PERIOD AND IF THE PRESSURE EXCEEDS 200 PSIG (FULL CHART SCALE) THE PREVALVES MUST BE OPENED.

# SATURN/APOLLO

PAGE TEST NO. VEHICLE

COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
S	1		RCSM	CSM RECORDERS TO "FAST SPEED".	
9			6000	MONITOR SC HMRTI TON ARM RETRACTED	
3	1.		CSKr	MONITOR SC UMBILICAL ARM RETRACTED.	
S	1		CSRP	MONITOR SHORT MAST 2 AND 4 PRESSURIZING VALVES #1 "OPEN".	
ŀ					
	S S	S 1	S 1	S 1 RCSM  S 1 CSRP	S 1 RCSM CSM RECORDERS TO "FAST SPEED",  S 1 CSRP MONITOR SC UMBILICAL ARM RETRACTED.

# EMERGENCY PROCEDURE (T-3" TO T-0")

## AUTOMATIC CUTOFF

- 1. C1FR: OBSERVE AUTOMATIC CUTOFF
- C1FP: Verify deluge purge ON if Not ON, operate deluge purge to MANUAL
- 3. RSEQ: SEQUENCE RECORDERS TO SLOW SPEED
- 4. RWCP: OPERATE WATER CONTROL SYSTEM AS REQUIRED
- 5. RVFC: HYDRAULIC PUMPS 1, 2, 3, AND 4 OFF
- 6. C1NP: FUNCTION SELECTOR TO PRELAUNCH
- 7. CGCM: CAMERA ARMING OFF
- 8. C1LS: LAUNCH SEQUENCER ARM OFF
- 9. C1DP, C4DP: SAFE DESTRUCT
- 10. RBHM: VERIFY FUEL AND LOX TANKS VENTED TO ZERO PSI
- 11. RPLC: VERIFY LH2 AND LOX TANKS VENTED TO ZERO PSI
- 12. C1NP, CUNP: SAFETY SWITCH ARMING OFF
- 13. The prevalves may remain closed for 10 minutes during which the Lox tanks must be pressurized, prevalves opened and Lox bubbling initiated. The Lox tanks

- SHOULD THEN BE VENTED AND BUBBLING DISCONTINUED WHEN THE PUMP INLET TEMPERATURES REACH -275° F.
- 14. Upon exceeding 10 minutes, the prevalves must remain closed, Lox
  drained, and standby observed until
  all suction line frost is melted.
  The pump inlet pressures must be
  monitored during this period, and
  if the pressure exceeds 200 PSIG
  (full chart scale) the prevalves
  must be opened.
- 15. CAUTION: PRIOR TO RESETTING CUTOFF, VERIFY THE FOLLOWING:
  - A. C1FP: ALL PREVALVE SWITCHES TO OFF POSITION
  - B. C1FP: Lox dome purge ON (REQUIRED FOR 15 SECONDS MINIMUM AFTER CUTOFF)
  - c. C1FP: GG Lox INJECTOR PURGE ON (REQUIRED FOR 10 MINUTE MINIMUM AFTER CUTOFF)
- 16. C1NP: IGNITION ARMING OFF
- 17. CNTS: SECURE ALL OTHER EQUIPMENT AS REQUIRED

PAGE TEST NO. VEHICLE

TIME	COMM.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
3"						
	S	1		C1FR	VERIFY IGNITION COMMAND.	
2,98"						
	S	1		CSRP	MONITOR IGNITION OF ENGINES #5 AND #7.	
2.88"						
	S	1		CSRP	MONITOR IGNITION OF ENGINES #6 AND #8.	
2.78"						
	S	1		CSRP	MONITOR IGNITION OF ENGINES #2 AND #4.	
2.68"						
	S	1		CSRP	MONITOR IGNITION OF ENGINES #1 AND #3.	
	S	2		CSRP	ALL ENGINES RUNNING "ON".	
	S	3		C1FR	TC FUEL INJECTOR PURGE "OFF",	

# EMERGENCY PROCEDURE (T-0" to T+0")

# AUTOMATIC CUTOFF

1.	C1FR:	OBSERVE AUTOMATIC CUTOFF	13.	CINP,	CUNP: SAFETY SWITCH ARMING OFF
2.	CIFP:	ALL PREVALVE SWITCHES TO OFF POSITION	14.	RBHM:	VERIFY S-I FUEL AND LOX TANKS VENTED TO ZERO PSI
3.	CIFP:	VERIFY DELUGE PURGE ON - IF NOT ON, OPERATE DELUGE PURGE TO MANUAL	15.	RPLC:	VERIFY S-IV LH2 AND LOX TANKS VENTED TO ZERO PSI
4.	RWCP:	OPERATE WATER CONTROL SYSTEM AS REQUIRED	16.	CILS:	LAUNCH SEQUENCER ARM OFF
5.	CIMP:	RECORDER TRANSFER SWITCH TO EMERGENCY POSITION			THE PREVALVES MAY REMAIN CLOSED
6.	RVFC:	Hydraulic pumps 1, 2, 3 and 4 OFF			FOR 10 MINUTES DURING WHICH THE LOX TANKS MUST BE PRESSURIZED,
7.	CINP:	FUNCTION SELECTOR TO PRELAUNCH			PREVALVES OPENED AND LOX BUBBLING INITIATED. THE LOX TANKS SHOULD
8.	CINP:	VERIFY PREVALVE SWITCHES OFF THEN RESET CUTOFF			THEN BE VENTED AND BUBBLING DIS- CONTINUED WHEN THE PUMP INLET TEMPERATURES REACH -275° F.
9.	CIFP:	GG LOX INJECTOR PURGE TO MANUAL	18.	NOTE:	
10.	CIFP:	LOX DOME PURGE TO MANUAL			VALVES MUST REMAIN CLOSED, LOX DRAINED AND STANDBY OBSERVED UNTIL
11.	RSEQ:	SEQUENCE RECORDERS TO SLOW SPEED			ALL SUCTION LINE FROST IS MELTED. THE PUMP INLET PRESSURES MUST BE
12.	CIDP,	C4DP: SAFE DESTRUCT			MONITORED DURING THIS PERIOD AND IF THE PRESSURE EXCEEDS 200 PSIG (FULL CHART SCALE) THE PREVALVES MUST BE

OPENED.

19. CNTS: SECURE ALL OTHER EQUIPMENT AS REQUIRED.

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
-010"						
	S	1		CSRP GSSP	MONITOR COMMIT.	
	S	2		CSRP	HOLDDOWNS RELEASE,	
	S	3		CSRP	S-I LOX AND FUEL MASTS EJECT VALVES	
+0'0"						
	S	1		CSRP	VERIFY LIFTOFF.	
	S	2		CSRP	SHORT MAST 2 AND 4 RELEASE.	
	S	3		CSRP	UMBILICAL ARMS 1, 2 AND 3 RELEASE AND RETRACT.	
	S	4		CVMP	"CLOSE" VACUUM SUPPLY VALVES.	
	S	5		CVMP	"OPEN" PURGE SUPPLY.	
	S	6		RECS	PERFORM ECS POST LAUNCH OPERATIONS PER PROCEDURE LVO-E-7018.	
	SRO	7		ORNG	START GUIDANCE COMMAND TRANSMISSION.	
+5"						
	S	1		RWCP	TORUS RING WATER VALVES "OPEN".	
	S	2		RWCP	PAD FLUSH VALVE FAST "CLOSES".	
+6"						
	S	1		CPSP	LAUNCHER POWER "OFF".	
+8"						
	S	1		RSEQ	SEQUENCE RECORDERS TO "SLOW SPEED".	
+10"						
	S	1		RCSM	RECORDERS TO "SLOW SPEED".	

LAUNCH COUNTDOWN
DATE SEPTEMBER 5, 1964
REVISION

#### SATURN/APOLLO LAUNCH OPERATIONS

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
+10"						
	S	2		RBHM	SANBORN RECORDER #OFF#.	
÷30#						
		1			511 CAMERA OPERATION,	
	S	2		OCCF	SHUT OFF ECS LOW PRESSURE GN2 SUPPLY.	
	s	3	8	BSOC	TEST END SENSE SWITCH "ON",	
+1 1						
	SIV	1		CSSP	RESET TIMER.	
	SIV	2		RFSR	SEQUENCE RECORDER TO 2 MM/SEC.	
	SIV	3		CPPC	CLOSE CONTROL HELIUM SUPPLY.	
	SIA	4		CPSC	DECREASE COLD HELIUM SUPPLY DOME PRESSURE TO 0 PSIG.	
	SIV	5		CPSC	SWITCH COLD HELIUM SUPPLY MOMENTARILY TO VENT, THEN SUPPLY.	
	SIV	6		RPLC	SWITCH S-IV LO2 CONTROL TO "MANUAL".	
	SIV	7		CPSC	CLOSE LO2 NOZZLE PURGE.	
	SIV	8		CPSC	CLOSE LH2 NOZZLE PURGE.	
	SIV	9		RPLC	CHECK OR POSITION THE FOLLOWING SWITCHES:	
					A, LO2 UMBILICAL LINE PURGE "OPEN"	
					B, LO2 UMBILICAL LINE VFNT "OPEN"	
					C, LH2 UMBILICAL LINE PURGE "OPEN"	
					D. LH2 UMBILICAL LINE VENT "OPEN"	
					E, LH2 TRANSFER (MOMENTARILY) "SECURE"	
					F. LH2 TRANSFER "MANUAL"	
					G. LH2 REPLENISH "OPEN"	
					H. LH2 MAIN FILL "OPEN"	

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARK
+1 1						
	SIV	10		RPLC	SWITCH COMPUTER READOUT LOCK "ON".	
	SIV	11		RPLC	SWITCH LOX AND LH2 COMPUTERS "OFF".	
	SIV	12		C4DP C4OP CSSP	NORMALIZE ALL SWITCHES.	
	SIV	13		C4TG	COMPLETE S-IV SECURING PER PROCEDURE DAC-P-7011.	
47.8"						
		1			SEPARATION CAMERA OPERATION.	
		2			LH2 PRESTART.	
14.9"						
		1			S-I STAGE LEVEL SENSORS ARMED.	
19.5"						
		1			FIRE VENT PORTS.	
2'21"						
		1			S-I INBOARD ENGINE CUTOFF (NOMINAL).	
2'26"						
26,9"						
		1			S-I OUTBOARD ENGINE CUTOFF (NOMINAL).	
27.6"						
		1			S-IV ULLAGE ROCKET IGNITION (NOMINAL).	
		2			SEPARATION (NOMINAL).	

TEST NO. VEHICLE

TIME	COMM.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
2 <sup>†</sup> 27,72"						
		1			S-I RETRORDCKET IGNITION (NOMINAL).	
29,4"						
		1			S-IV ENGINE START (NOMINAL).	
3 1 0 #						
	RD4	1		RLHC	INERT LH2 SYSTEM PER LVO-L-1006.	
6'0"						
	RD3	1		RWCP	SECURE WATER SYSTEM PER PROCEDURE LVO-E-7014.	
7 / 30 #						
		1		ORNG	AZUSA SWITCHOVER.	
43.5"						
		1			S-IV CUTOFF ARM (NOMINAL).	
10'19"						
		1			S-IV CUTOFF (NOMINAL).	
		1			ACCORDING TO PREARRANGED CRITERIA AND TIME SEND "SAFE" COMMAND.	
		2		ORNG	PERFORM ORBITAL LOAD ACCORDING TO INFORMATION IN OD 2460.	
					NOTE	
					UPON RETURN TO PAD.	

LAUNCH COUNTDOWN
DATE SEPTEMBER 5, 1964
REVISION

# SATURN/APOLLO LAUNCH OPERATIONS

PAGE TEST NO. VEHICLE

TIME	COMM. CH.	SEQUENCE	COMMAND STA.	RESPONSE STA.	DESCRIPTION	REMARKS
		1		1	KEEP LAUNCHER CLEAR OF ALL NON DESIGNATED PERSONNEL UNTIL SGMP VERIFIES ITEMS FOLLOWING ARE COMPLETE:	
		1-1		SGMP	INSTALL LAUNCHER SAFETY NET.	
		1-2			OPEN LAUNCHER ACCESS DOOR AND VERIFY LAUNCHER INTERIOR CLEAR OF GN2.	
		1-3			SAFETY WIRE HOLDDOWN ARM RELEASE PANEL SUPPLY VALVE (4444) "CLOSED" AND VENT VALVE (4442) "OPEN".	
		1-4			VERIFY APCD HAS SECURED ALL LAUNCHER PNEUMATICS.	
		1=5			PERFORM POST LAUNCH PNEUMATIC OPERATIONS PER PROCEDURE LVO-E-7022.	
		2		CONTRACTOR OF THE PARTY OF THE	PERFORM USA POST LAUNCH OPERATIONS PER PROCEDURE LVO-E-7023.	